

Lecture 26

Tue. 11.22.2016

Geometry and Symmetry of Coulomb Orbital Dynamics

(Ch. 2-4 of Unit 5 11.22.16)

Rutherford scattering and hyperbolic orbit geometry

Backward vs forward scattering angles and orbit construction example

Parabolic “kite” and orbital envelope geometry

Differential and total scattering cross-sections

Eccentricity vector $\boldsymbol{\epsilon}$ and (ϵ, λ) -geometry of orbital mechanics

Projection $\boldsymbol{\epsilon} \cdot \mathbf{r}$ geometry of $\boldsymbol{\epsilon}$ -vector and orbital radius \mathbf{r}

Review and connection to usual orbital algebra (previous lecture)

Projection $\boldsymbol{\epsilon} \cdot \mathbf{p}$ geometry of $\boldsymbol{\epsilon}$ -vector and momentum $\mathbf{p} = m\mathbf{v}$

General geometric orbit construction using $\boldsymbol{\epsilon}$ -vector and (γ, \mathbf{R}) -parameters

Derivation of $\boldsymbol{\epsilon}$ -construction by analytic geometry

Coulomb orbit algebra of $\boldsymbol{\epsilon}$ -vector and Kepler dynamics of momentum $\mathbf{p} = m\mathbf{v}$

Example of complete (\mathbf{r}, \mathbf{p}) -geometry of elliptical orbit

Connection formulas for (γ, \mathbf{R}) -parameters with (a, b) and (ϵ, λ)

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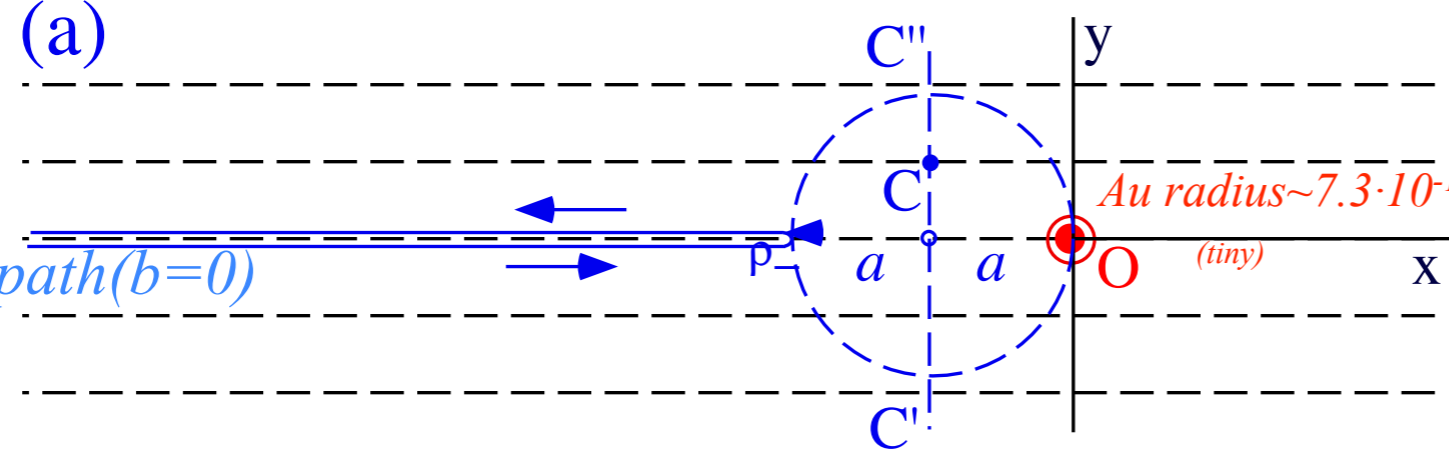
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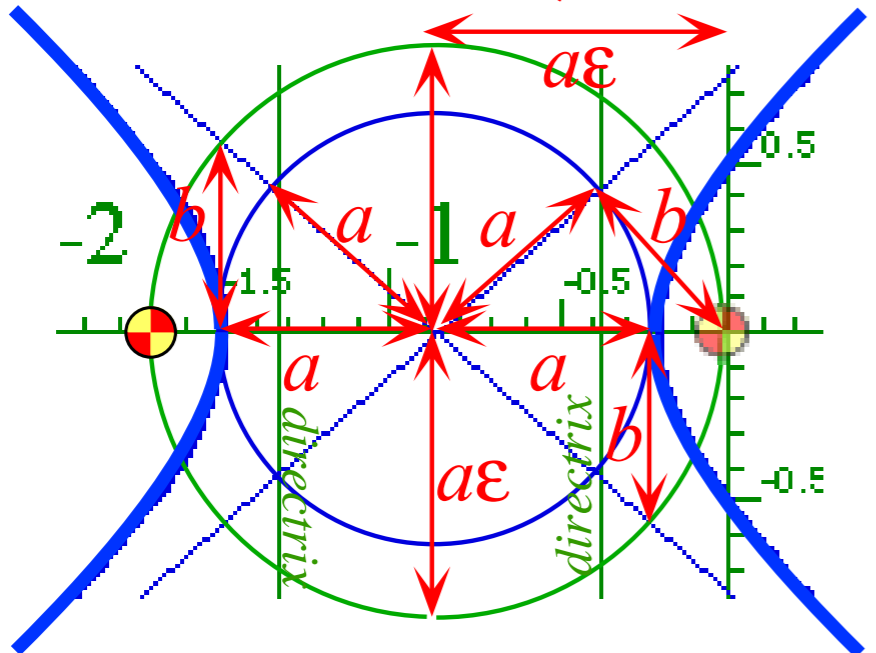
Connection formulas for (γ, R) -parameters with (a, b) and (ε, λ)

(a)

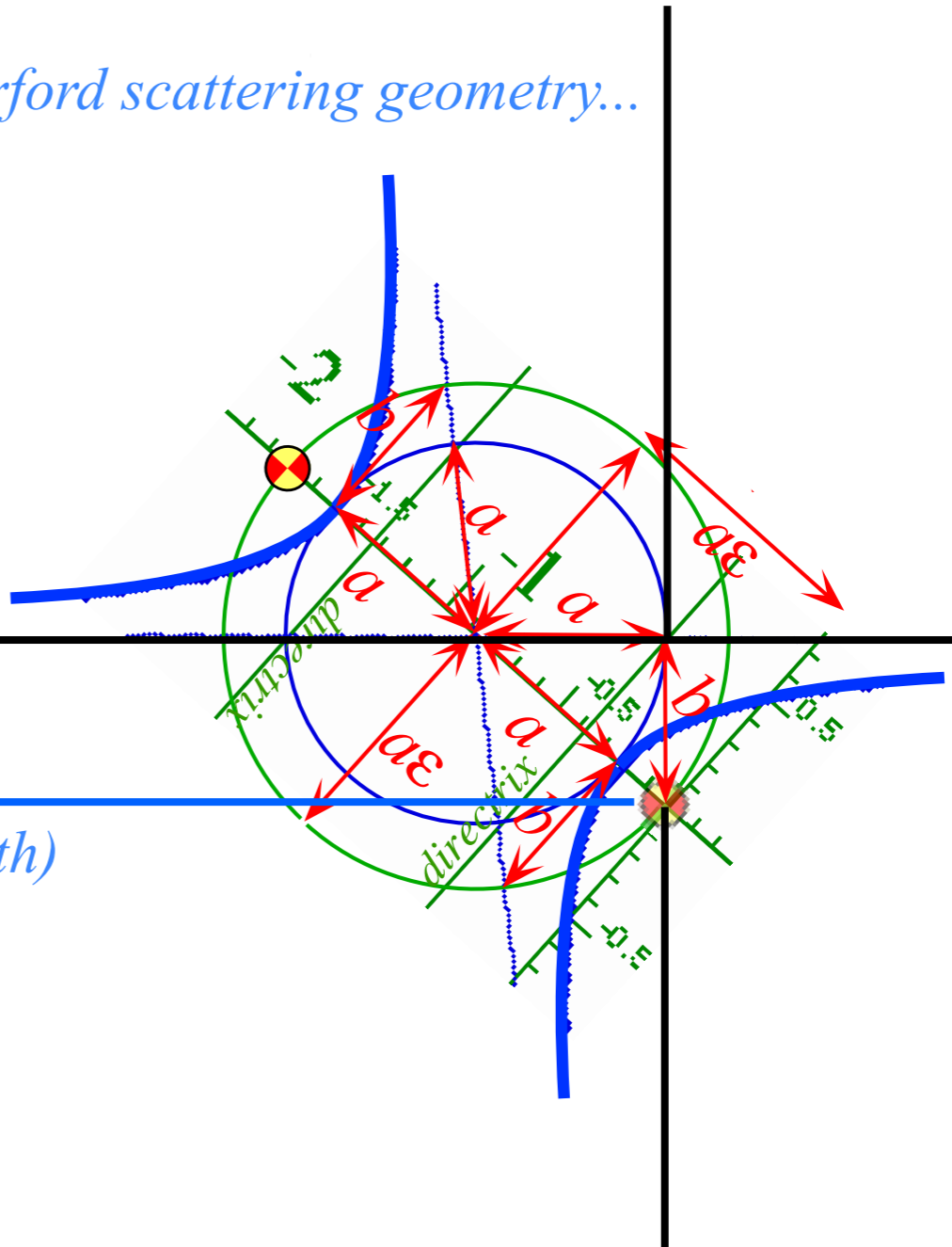
Dead-on-path ($b=0$)



Rutherford scattering of α^{+2} particles from Au^{+79} nucleus at O
 Assume "Dead-On" closest approach $2a$.
 $(E=k/2a)$ $a \sim 10^{-11}m \gg 7.3 \cdot 10^{-15}m$

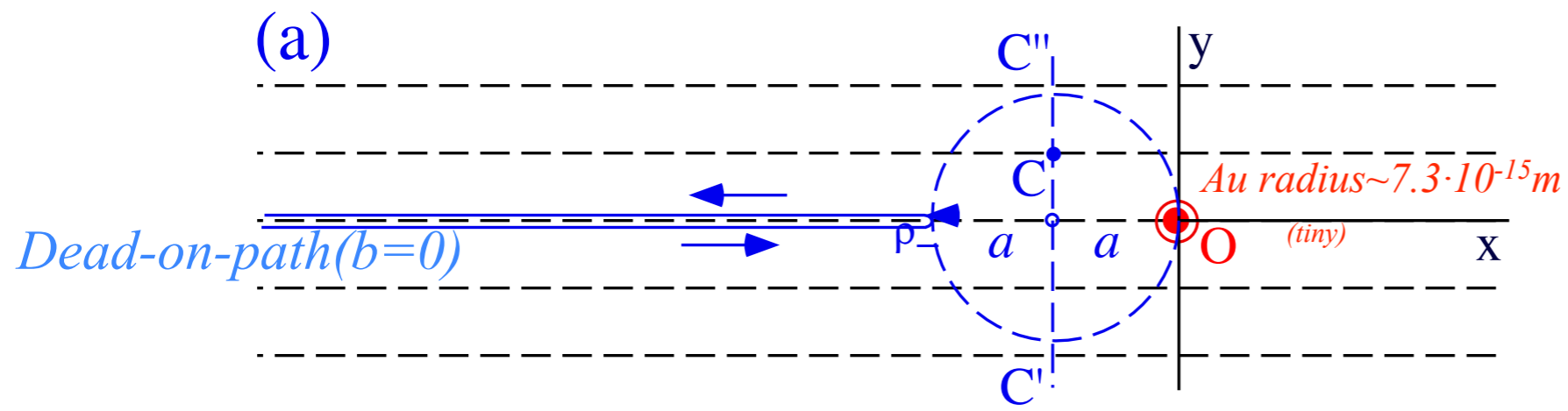


Rutherford scattering geometry...

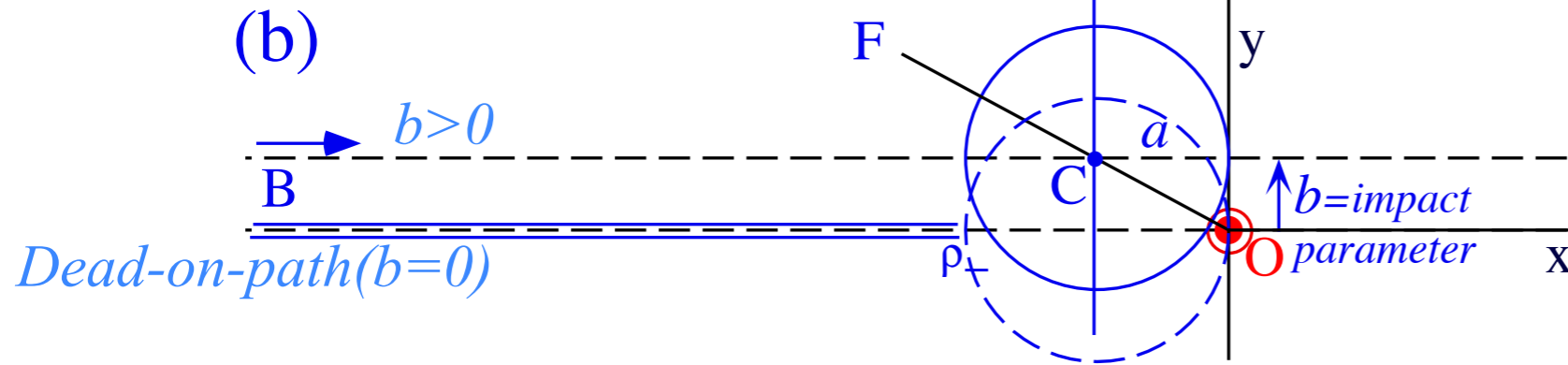


Alpha-particle beam direction →

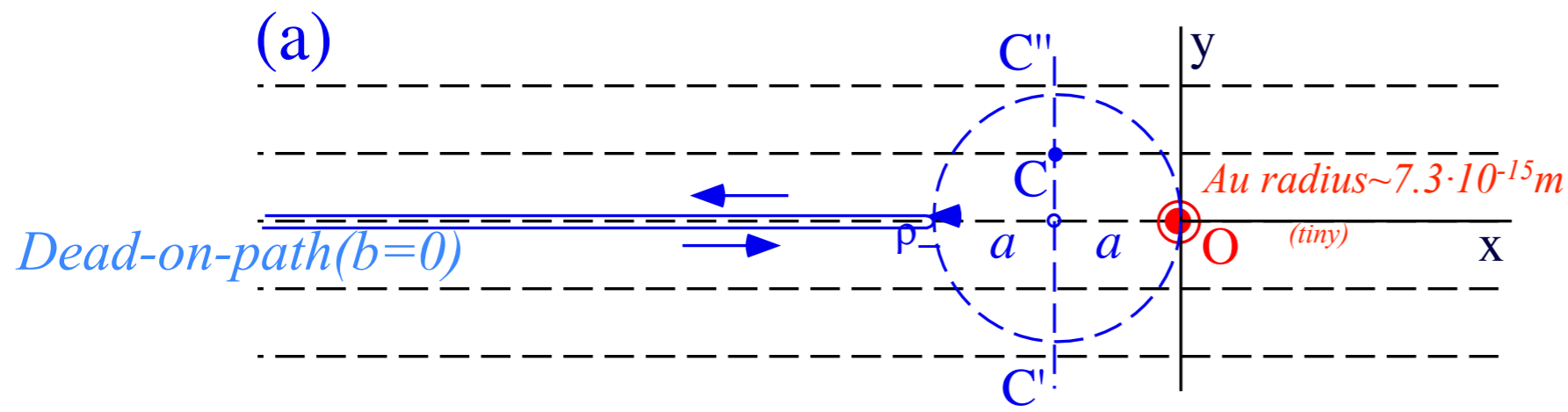
Gold nuclear target → (Dead-on-path)



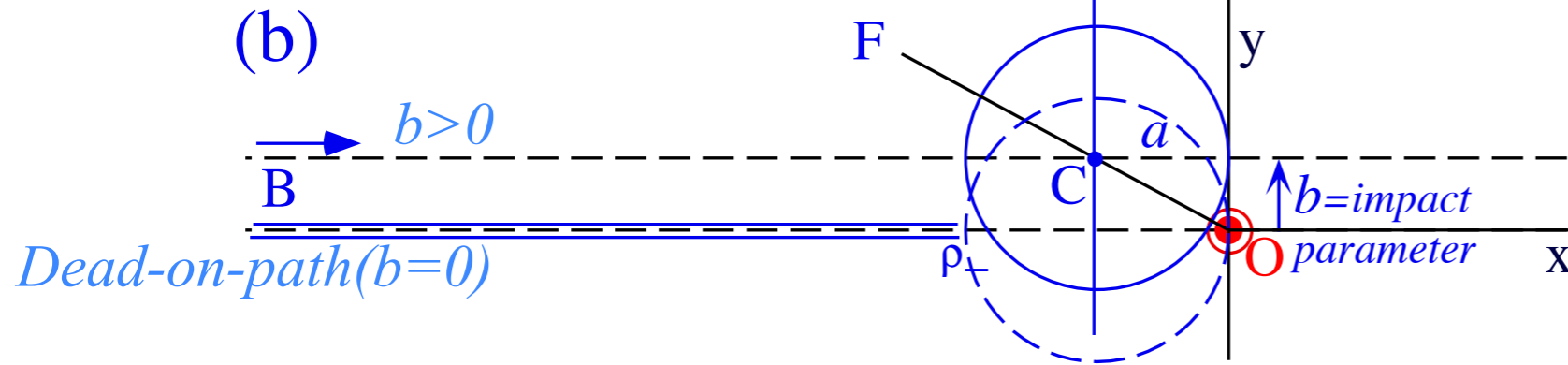
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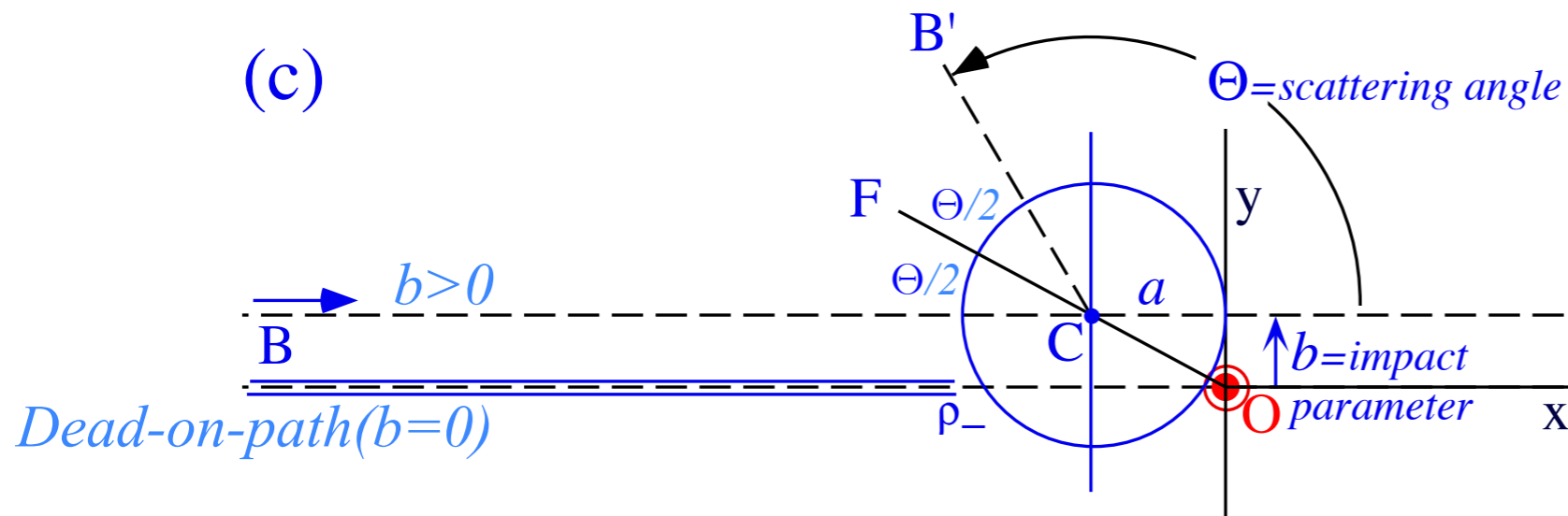
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 Draw circle of radius a around center point $C = (-a, b)$ tangent to y -axis.
 Draw "focus-locus" line OCF.



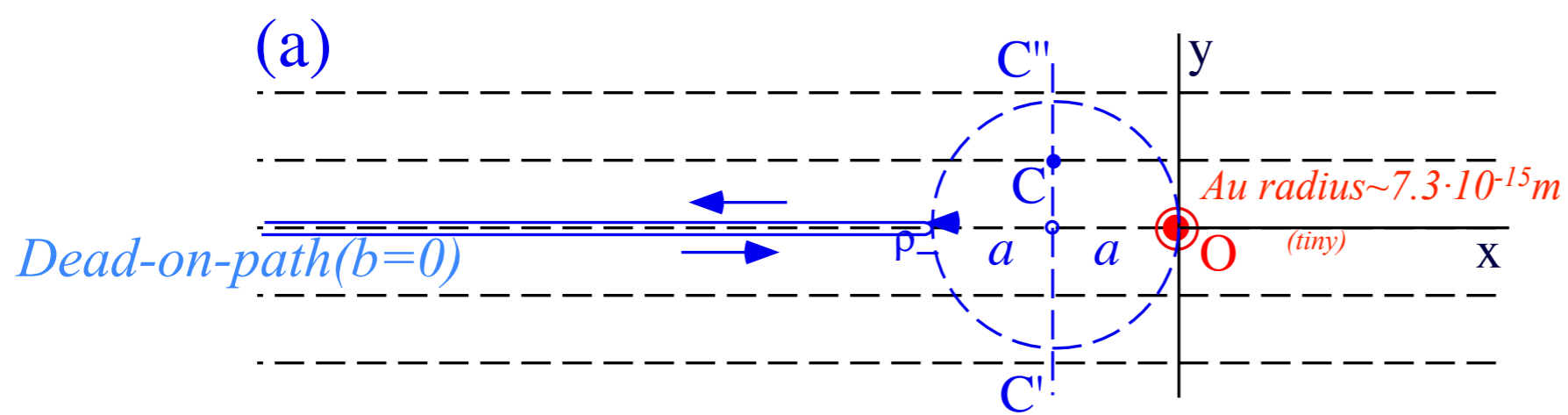
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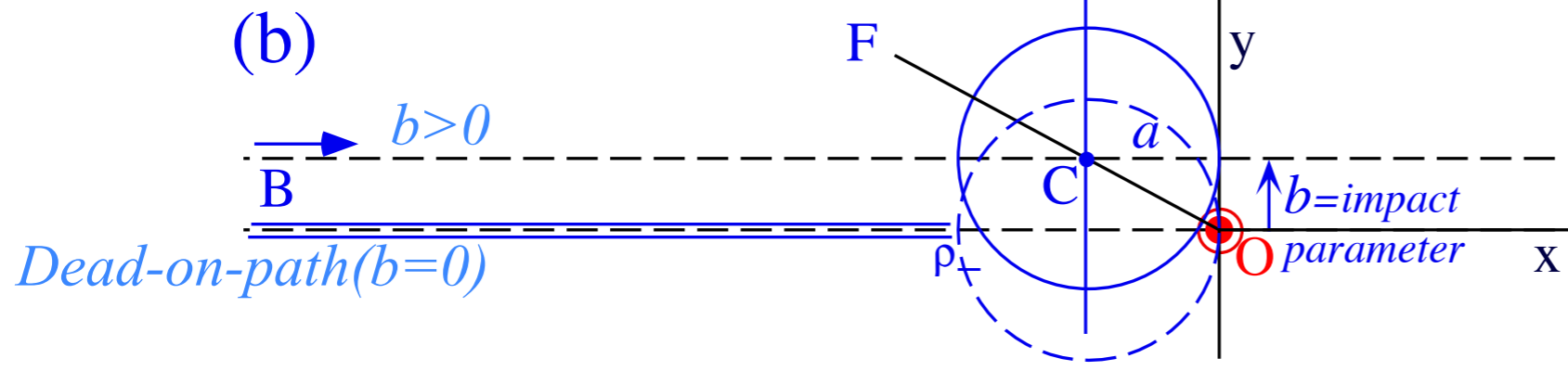
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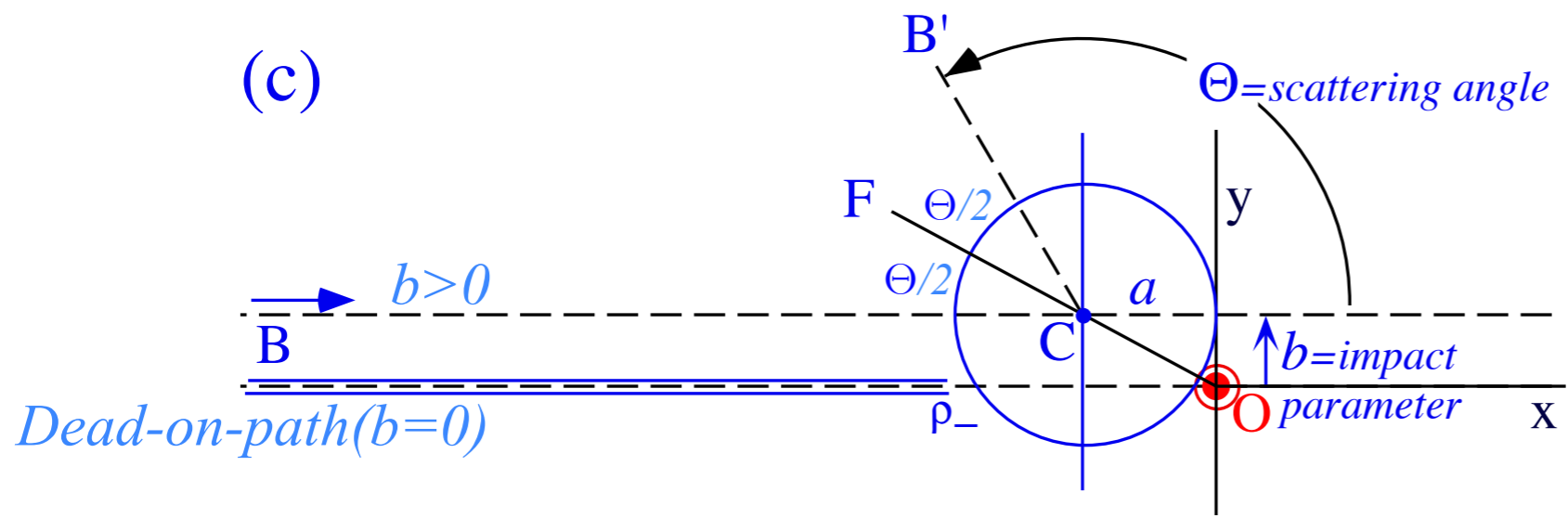
Copy angle $\angle BCF$ (equal to $\Theta/2$) to make angle $\angle FCB'$ (also equal to $\Theta/2$)
 Resulting line CB' is outgoing asymptote at scattering angle Θ .



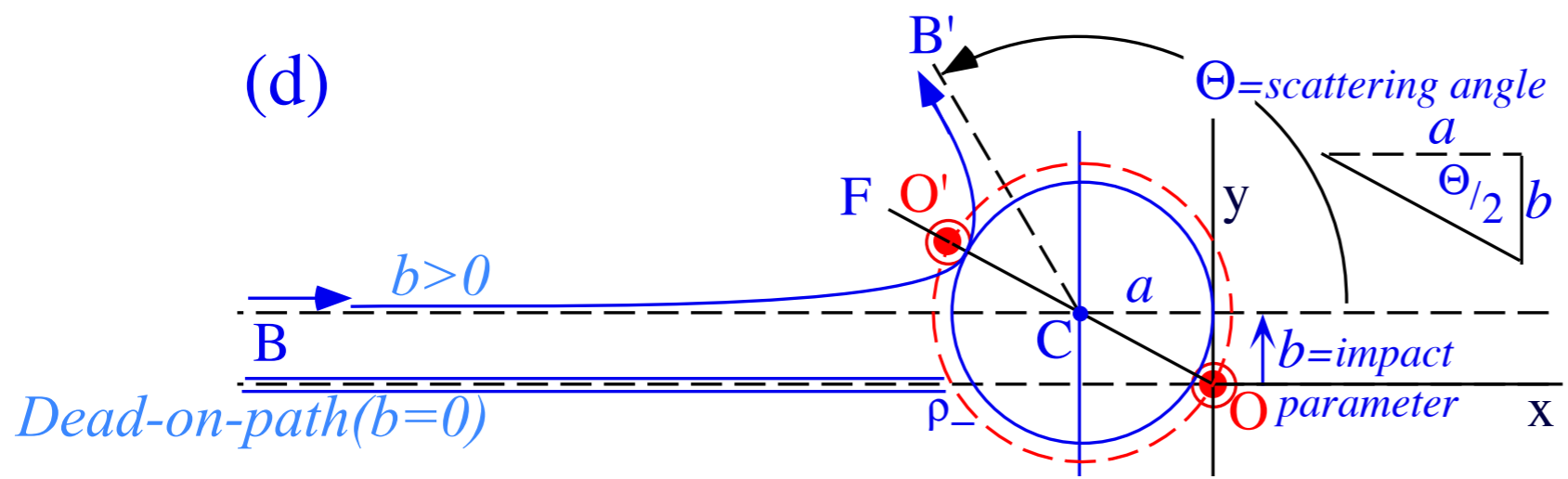
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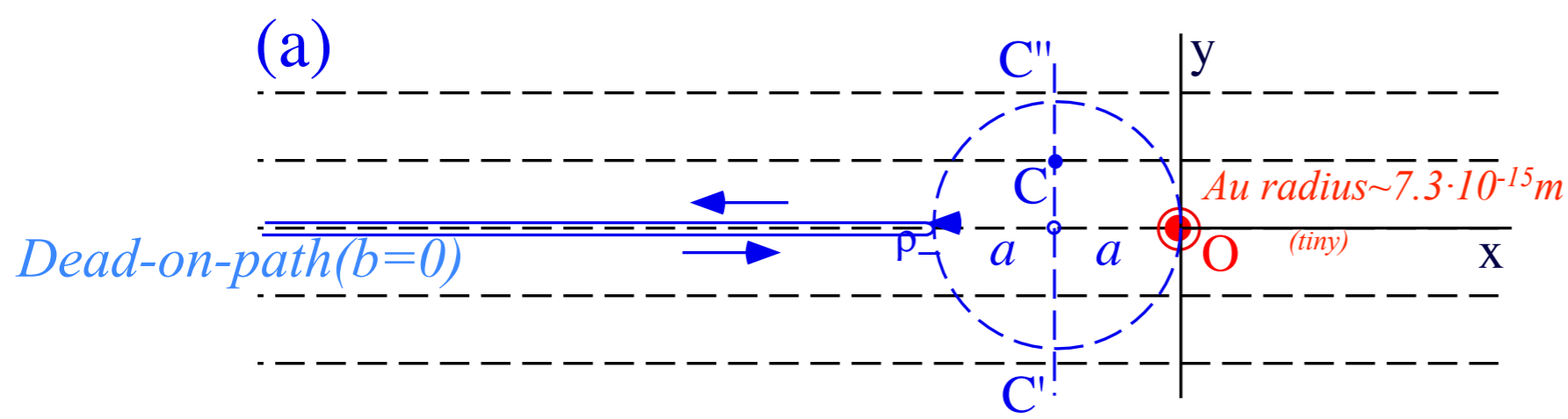
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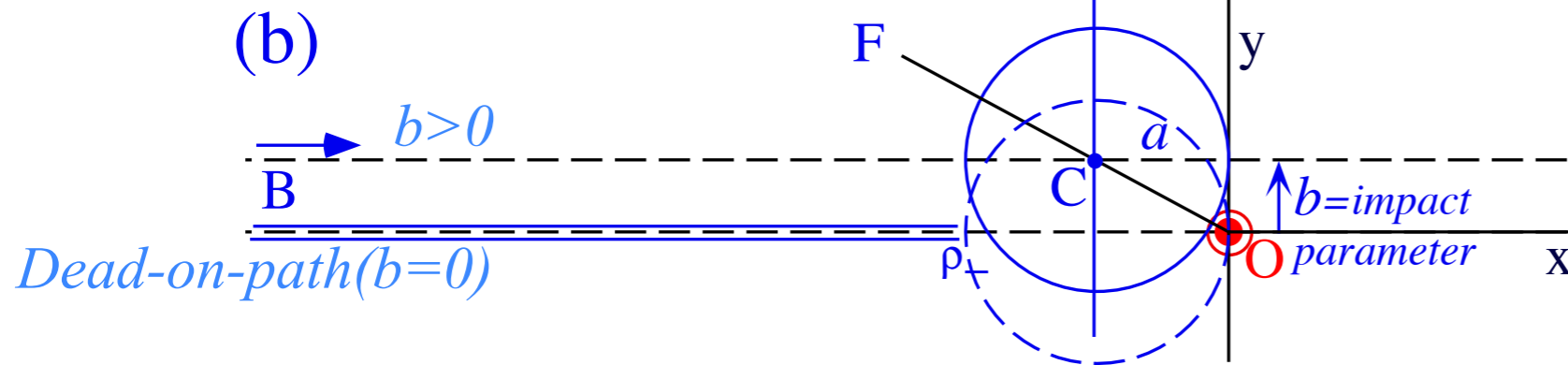
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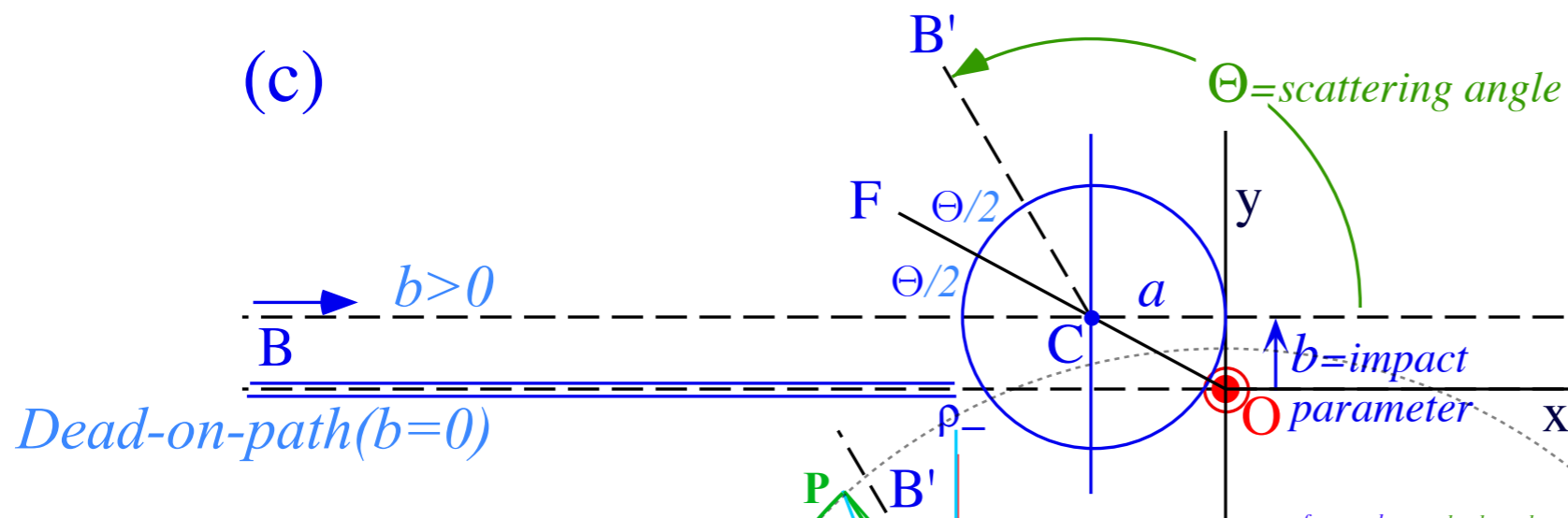
Locate secondary focus O' by drawing circle around point C of diameter CO thru point O. Diameter $O'CO$ is $2a\epsilon$. Hyperbolic orbit points P now found using constant $2a = \text{PO} - \text{PO}'$



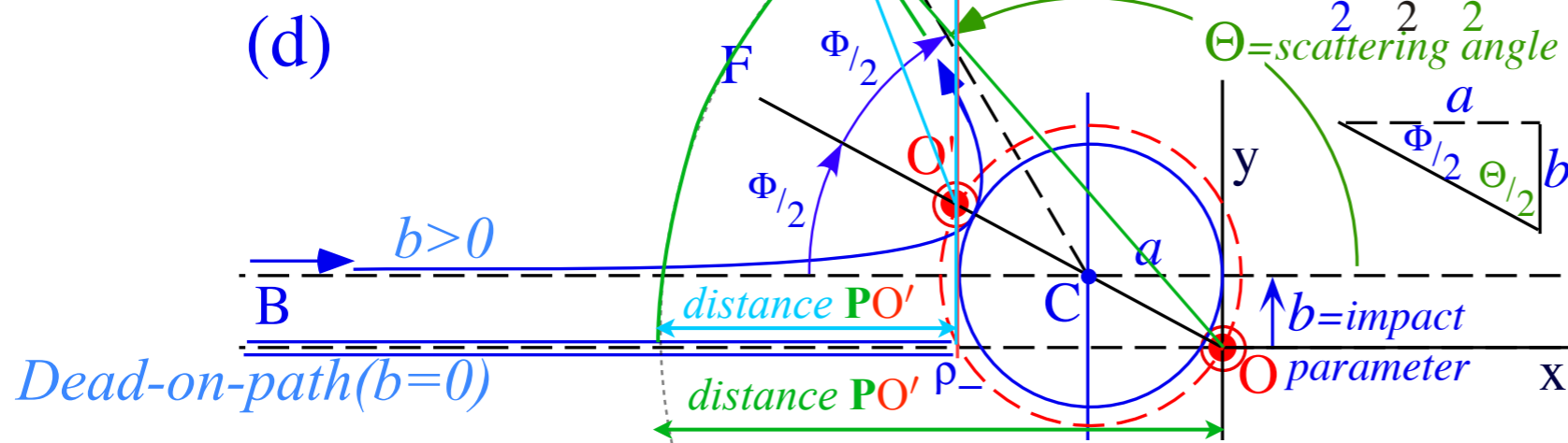
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Rutherford scattering and hyperbolic orbit geometry

- ➔ *Backward vs forward scattering angles and orbit construction example*
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- Differential and total scattering cross-sections*

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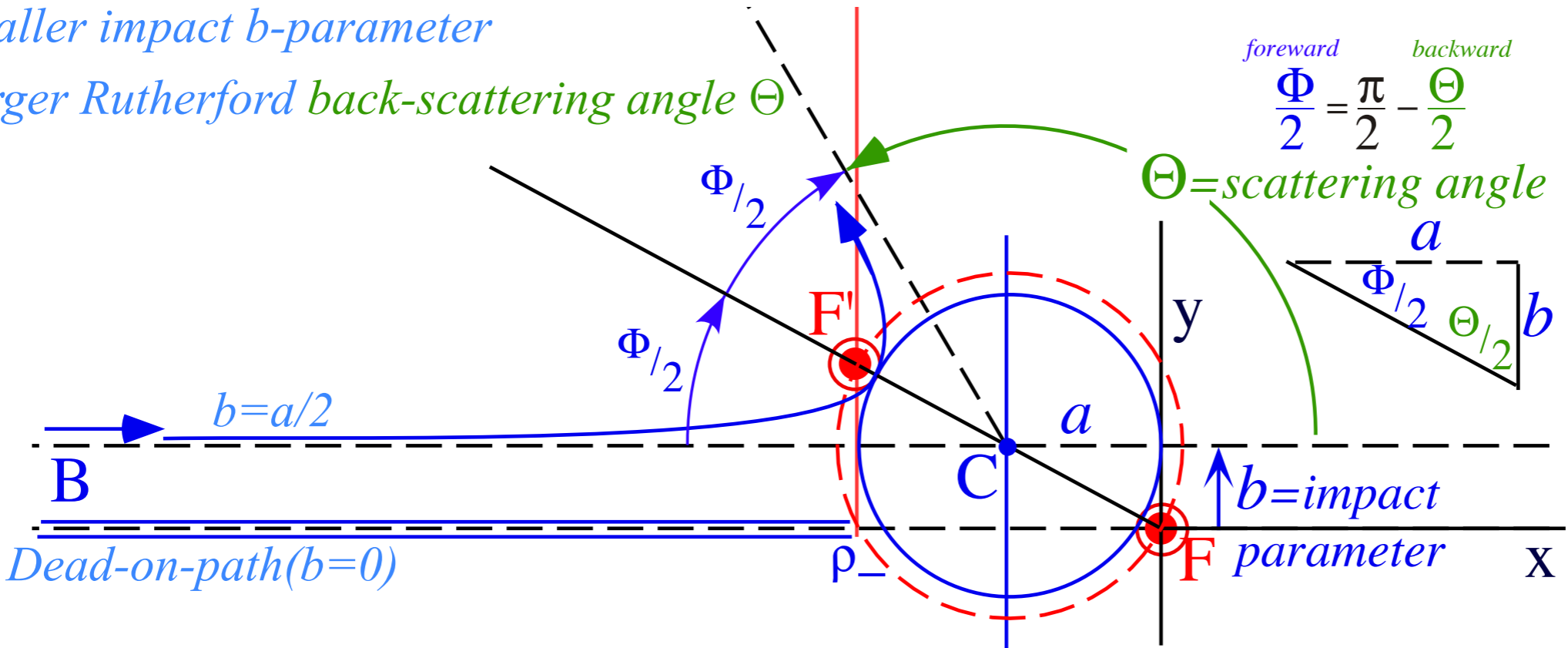
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Example of complete (\mathbf{r}, \mathbf{p}) -geometry of elliptical orbit

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Smaller impact b-parameter

Larger Rutherford back-scattering angle Θ



$$\frac{\Phi}{2} = \frac{\pi}{2} - \frac{\Theta}{2}$$

forward *backward*

$\Theta =$ scattering angle

$b = a/2$

B

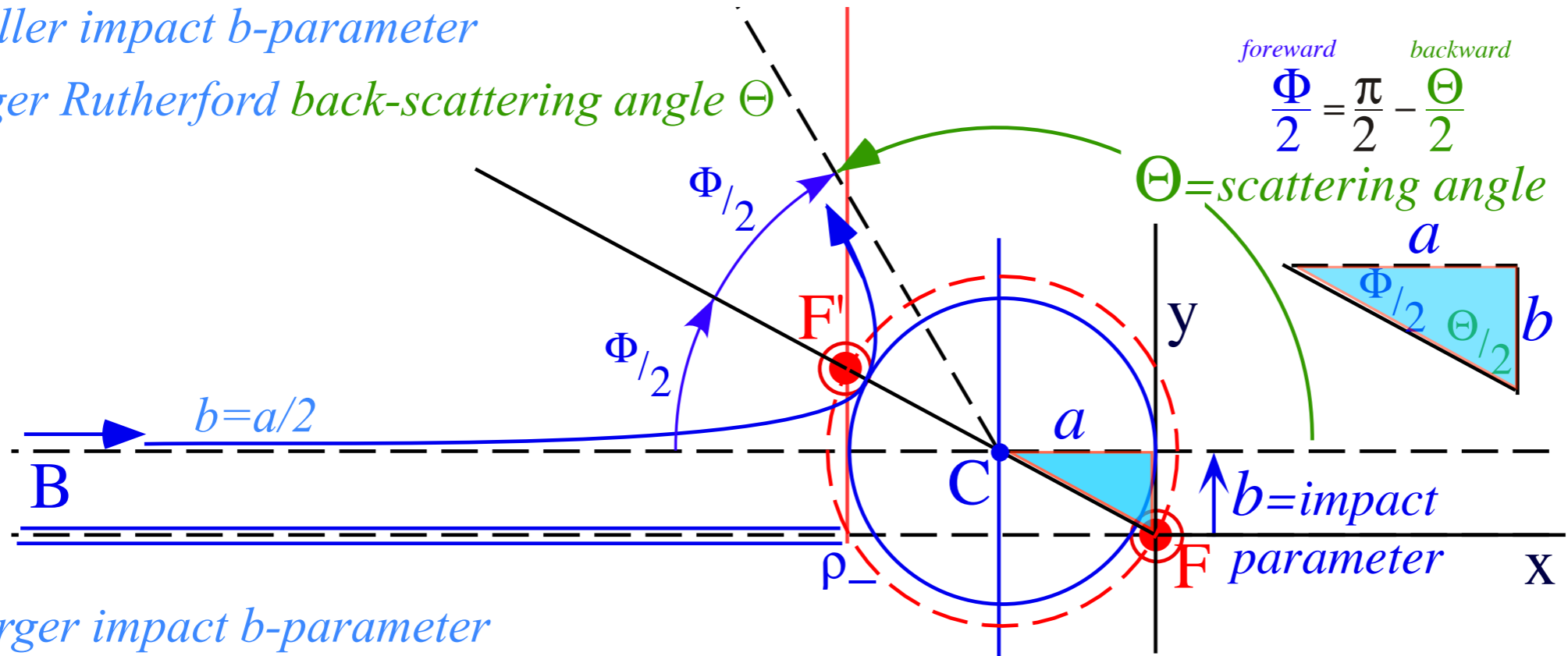
Dead-on-path ($b=0$)

$b =$ impact parameter

F

Smaller impact b -parameter

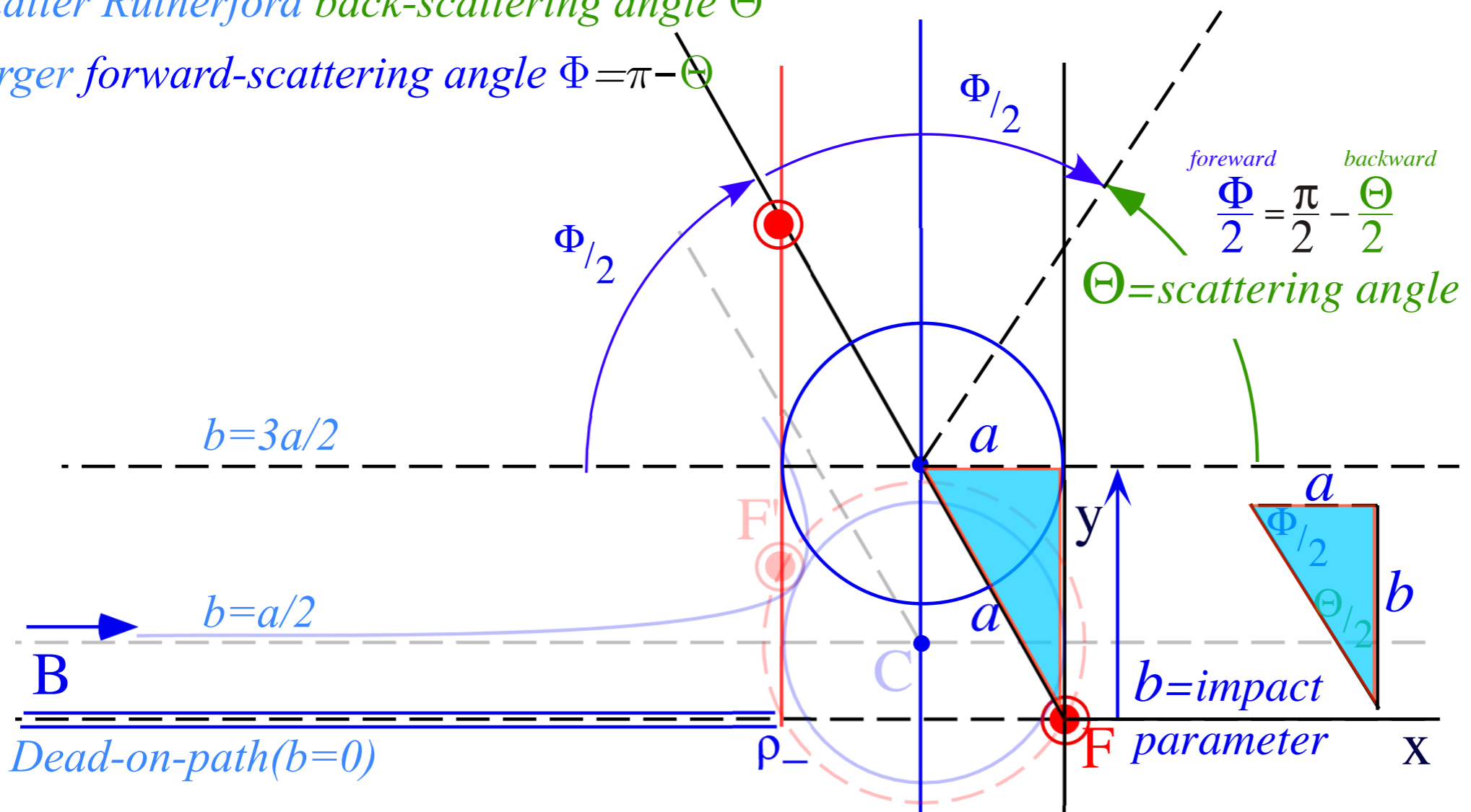
Larger Rutherford back-scattering angle Θ

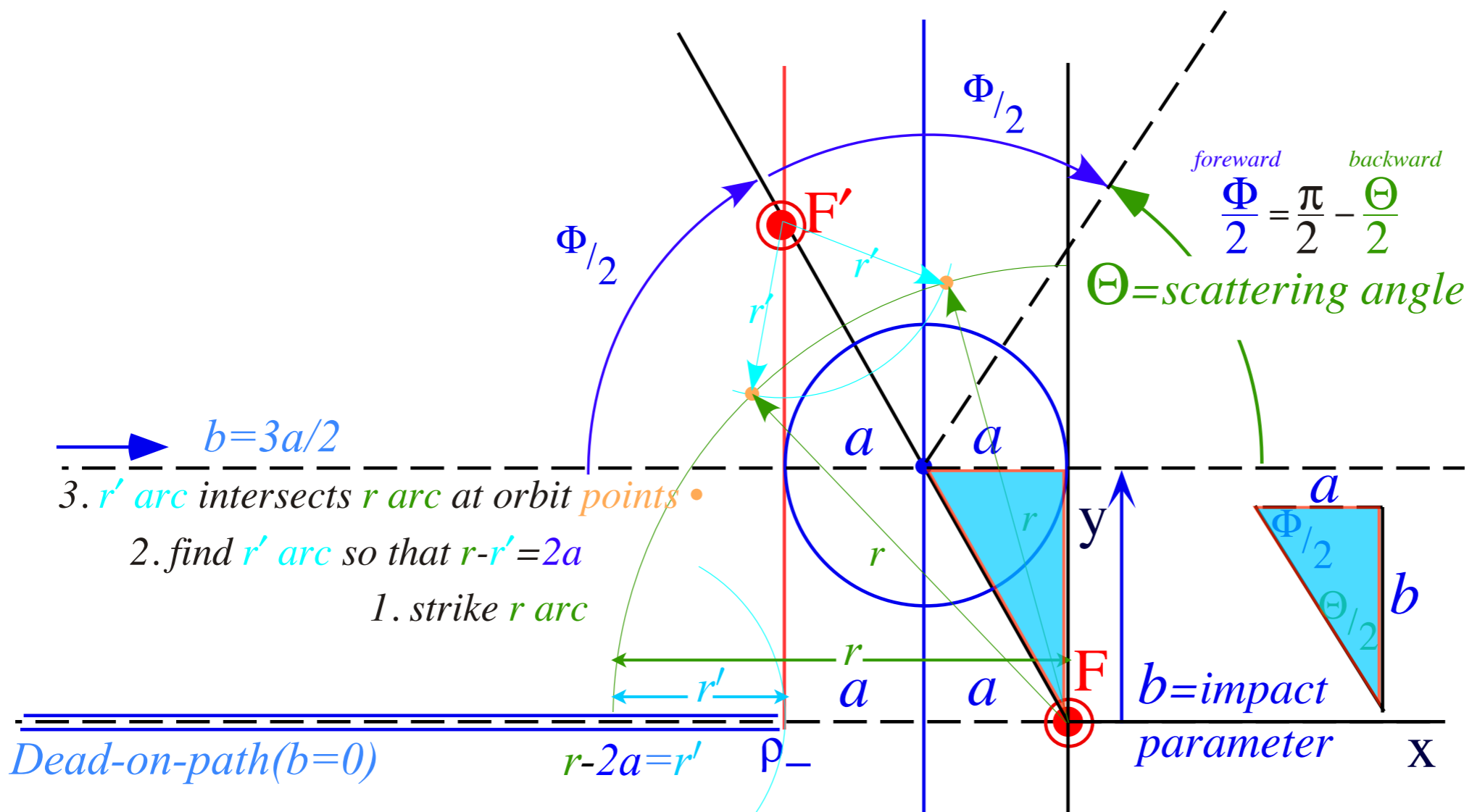
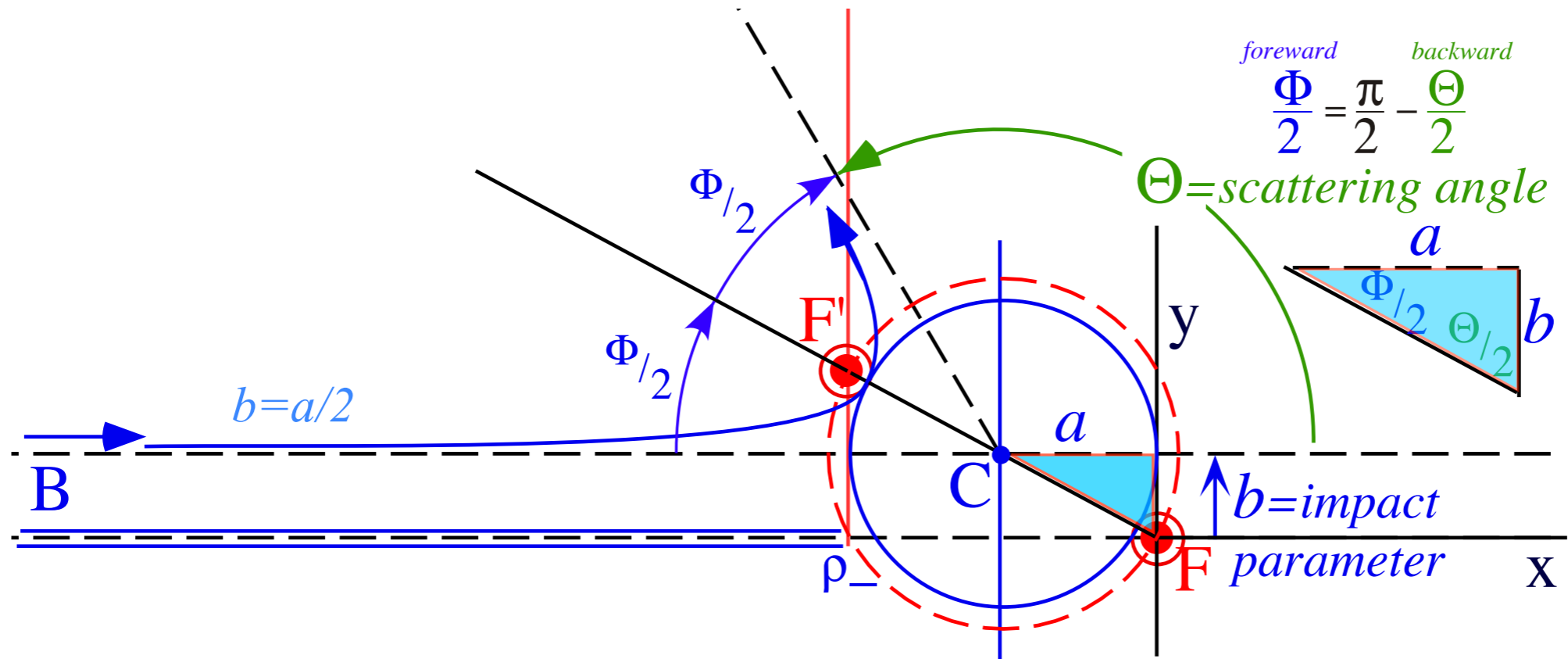


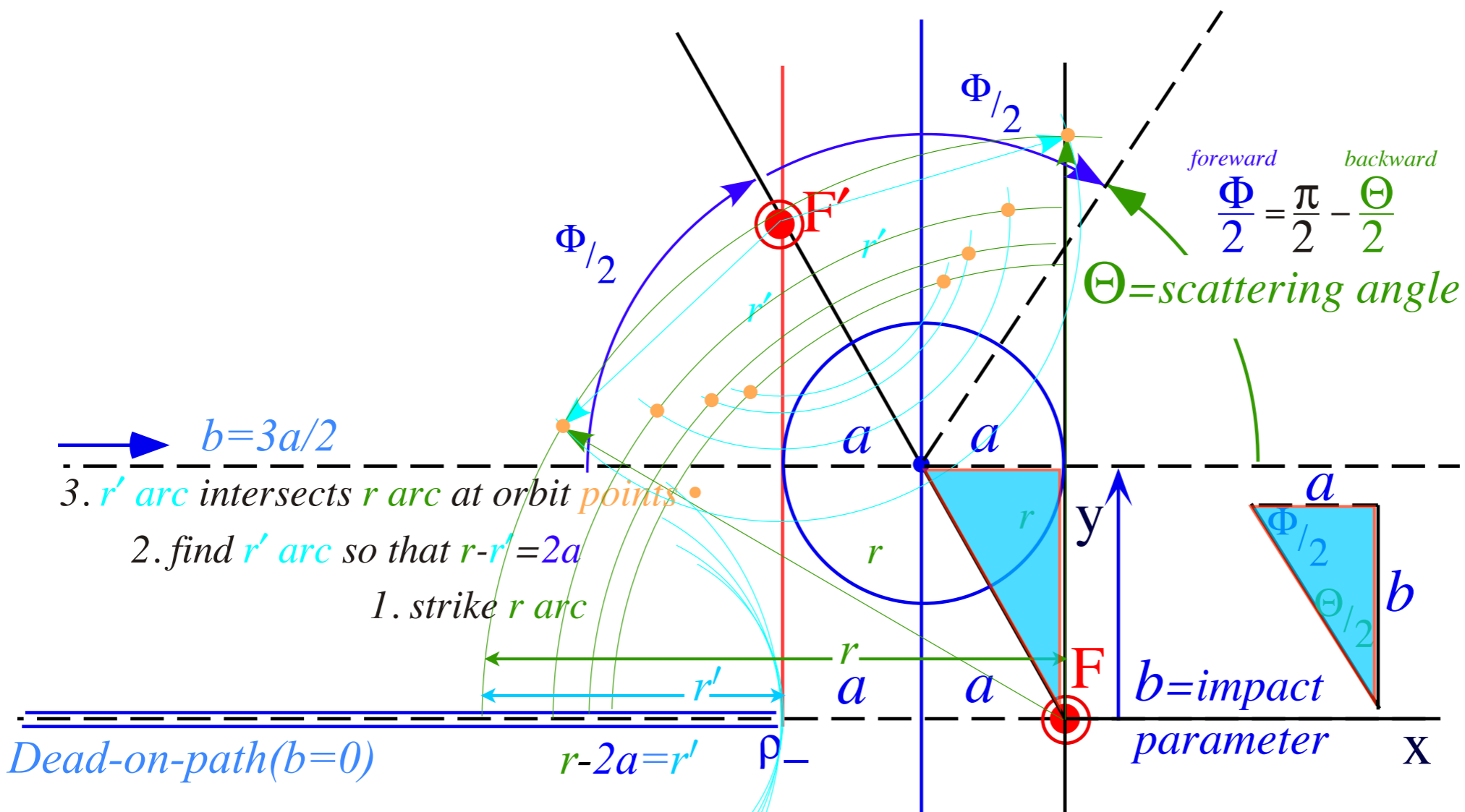
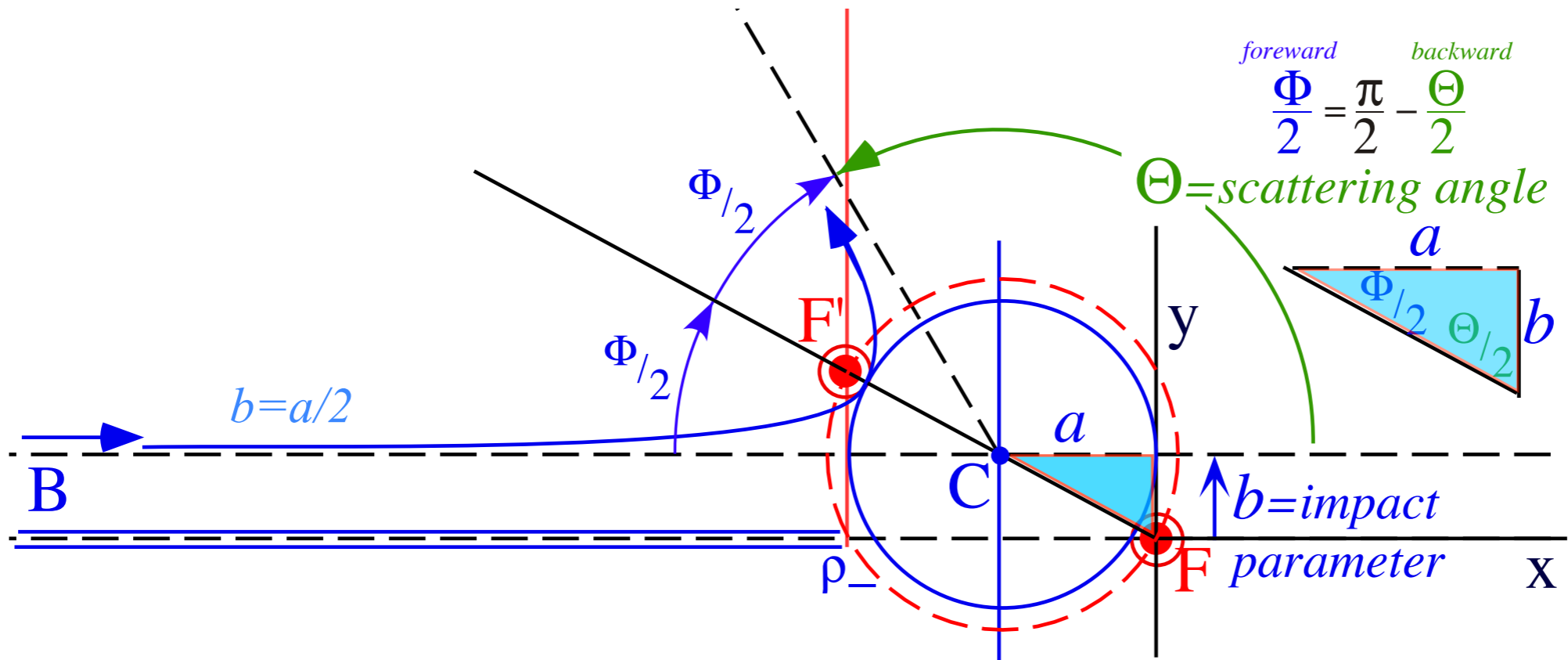
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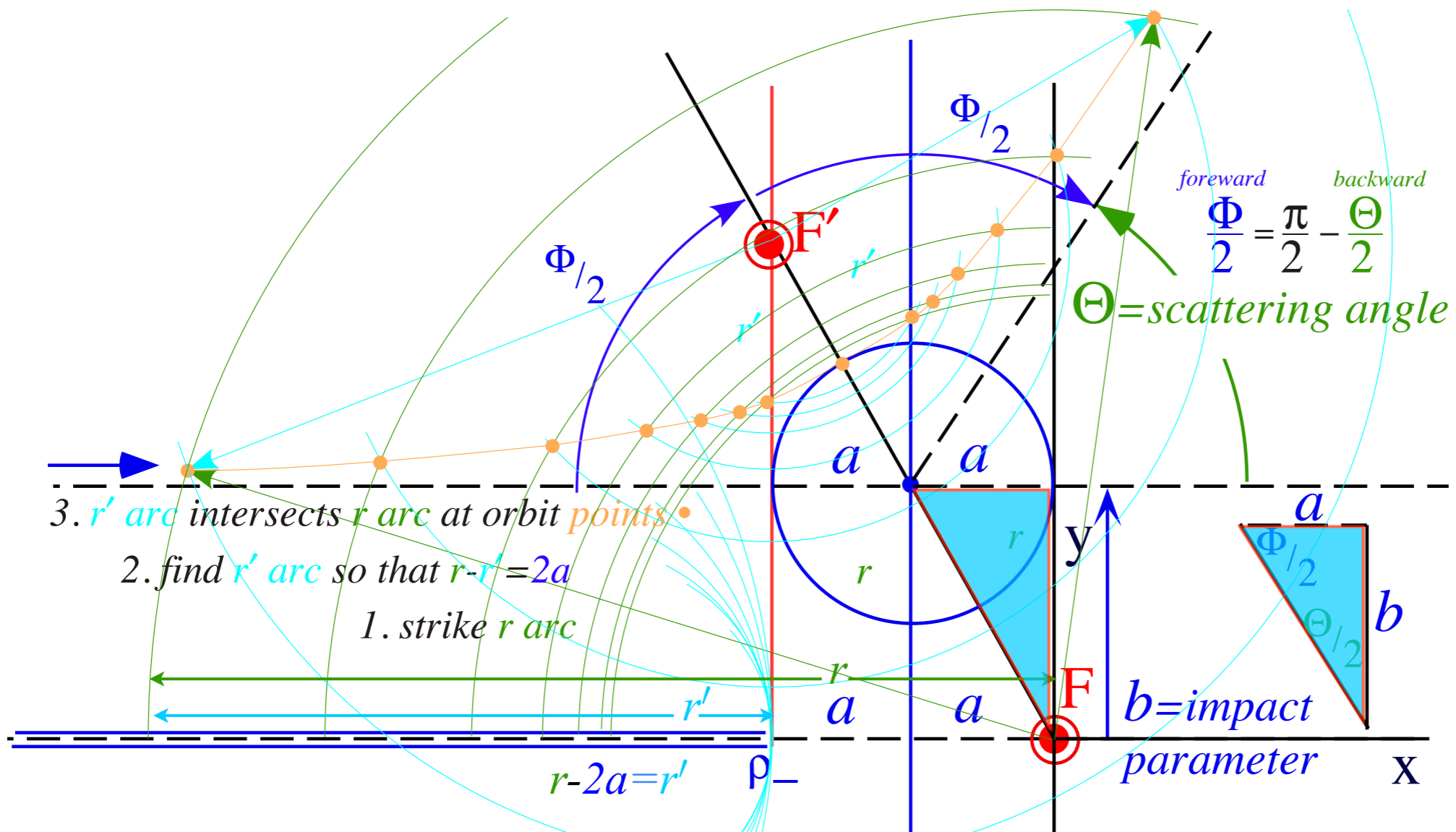
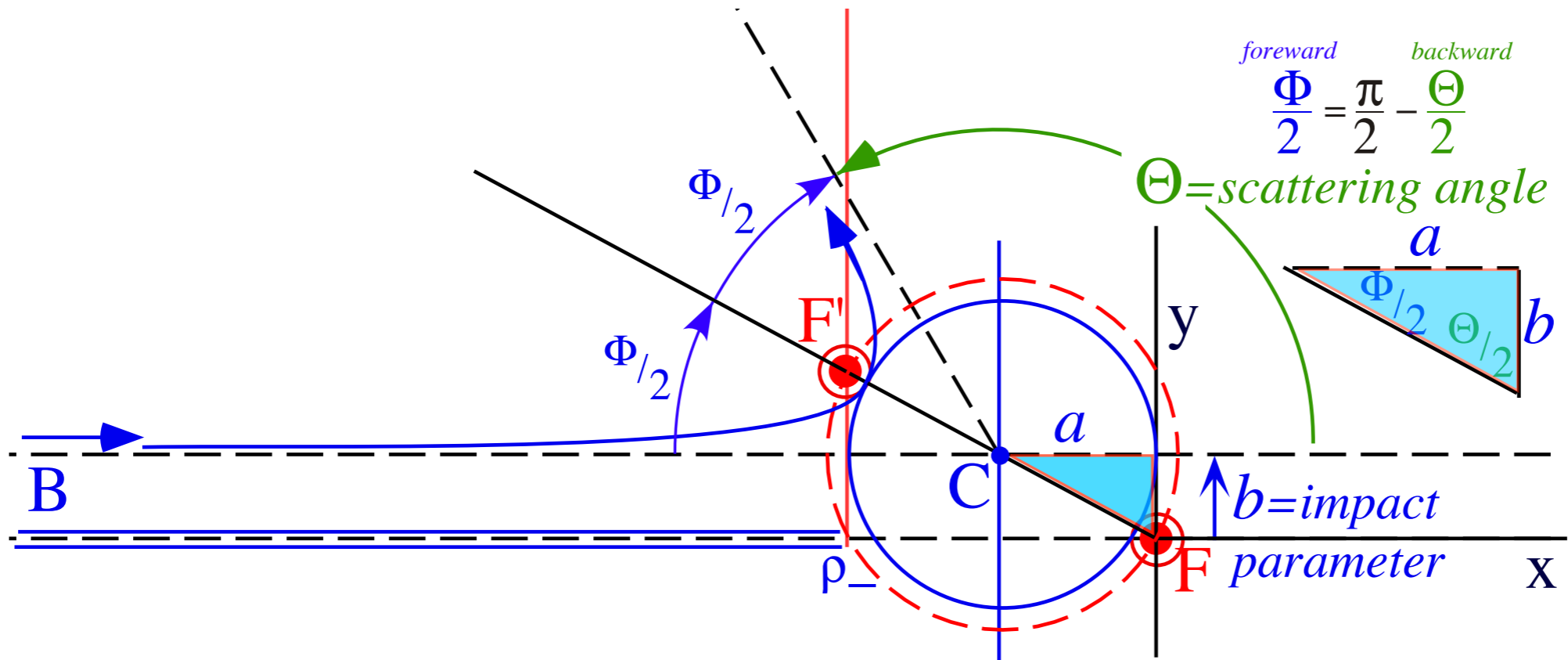
Smaller Rutherford back-scattering angle Θ

Larger forward-scattering angle $\Phi = \pi - \Theta$









Rutherford scattering and hyperbolic orbit geometry

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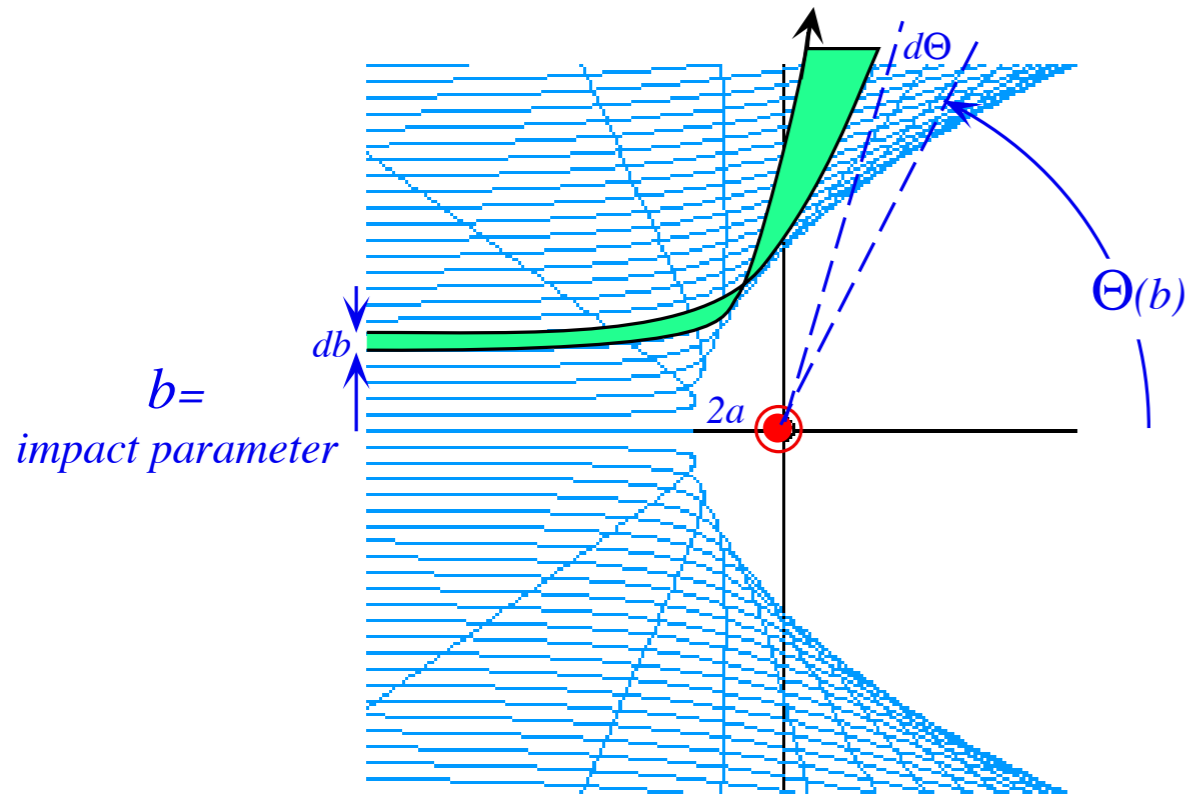
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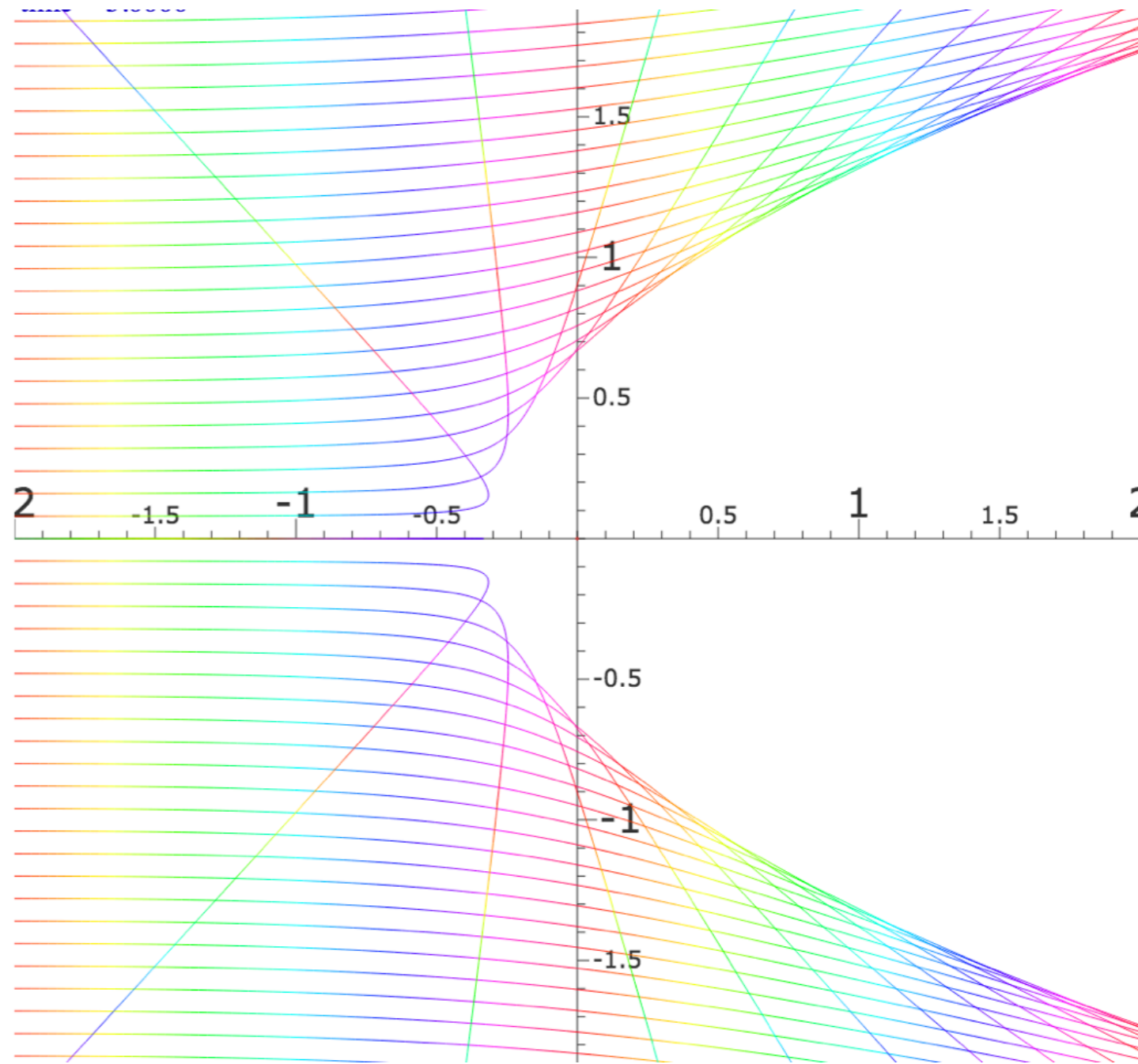
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Rutherford scattering geometry



<http://www.uark.edu/ua/modphys/markup/CoulItWeb.html?scenario=Rutherford>



Chapter 1 Orbit Families and Action

Families of particle orbits are drawn in a varying color which represents the classical action or Hamilton's characteristic function $SH = \int p dq$. (Sometimes SH is called 'reduced action'.) The color is chosen by first calculating $c = SH \text{ modulo } h\text{-bar}$ (You can change Planck's constant from its default value $h/2\pi = 1.0$) The chromatic value c assigns the hue by its position on the color wheel (e.g.; $c=0$ is red, $c=0.2$ is a yellow, $c=0.5$ is a green, etc.).

Chapter 2 Rutherford Scattering

A parallel beam of iso-energetic alpha particles undergo Rutherford scattering from a coulomb field of a nucleus as calculated in these demos. It is also the ideal pattern of paths followed by intergalactic hydrogen in perturbed by the solar wind.

Chapter 3 Coulomb Field (H atom)

Orbits in an attractive Coulomb field are calculated here. You may select the initial position $(x(0), y(0))$ by moving the mouse to a desired launch point, and then select the initial momentum $(p_x(0), p_y(0))$ by pressing the mouse button and dragging.

Chapter 4 Molecular Ion Orbits

Orbits around two fixed nuclei are calculated here. A set of elliptic coordinates are drawn in the background. After running a few trajectories you may notice that their caustics conform to one or two of the elliptic coordinate lines.

Volcanoes of Io (Paths=180, No color quant.) Parabolic Fountain (Uniform)

Space Bomb (Coulomb) Exploding Starlet (IHO)

Synchrotron Motion (Crossed E & B fields)

Rutherford scattering 2-Electron Orbits

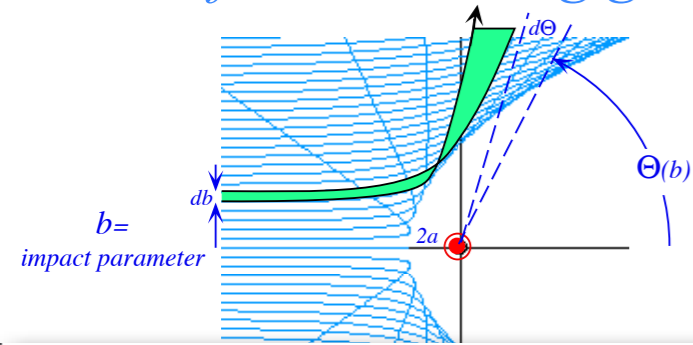
Atomic Orbits

Molecular Ion Orbits

Oscillator Scattering 2-Particle Orbits 2-Particle Collision

Rutherford scattering geometry

time = 3.8700



Terminal time t(off) = 5

Maximum step size dt = 0.03

Start launch angle phi1 = -180

Start launch angle phi2 = 180

Number of burst paths = 221

Charge of Nucleus 1 = 0.2

x-Position of Nucleus 1 = 0

y-Position of Nucleus 1 = 0

Charge of Nucleus 2 = 0

Coulomb (k12) = -1

Core thickness r = 0.000001

x-Stark field Ex = 0

y-Stark field Ey = 0

Zeeman field Bz = 0

Diamagnetic strength k = 0

Plank constant h-bar = 2

Color quantization hues = 64

Color quantization bands = 2

Fractional Error (e^x), x = 8

Particle Size = 6

Fix r(0) Fix p(0) Do swarm Beam

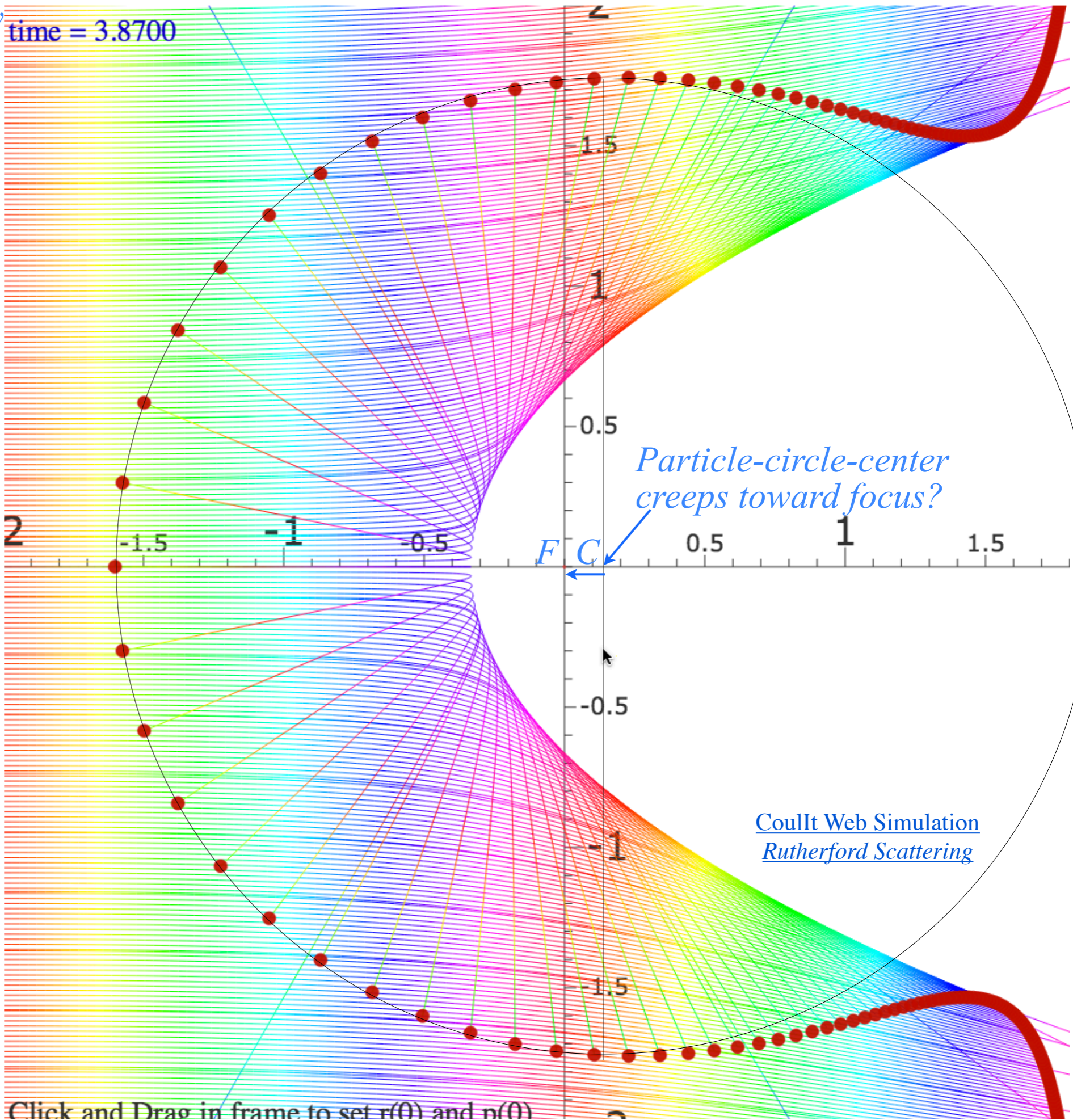
Plot r(t) Plot p(t)

Color action No stops Field vectors Info

Draw masses Axes Coordinates Lenz

Set p by phi Elastic 2 Free

Save to GIF

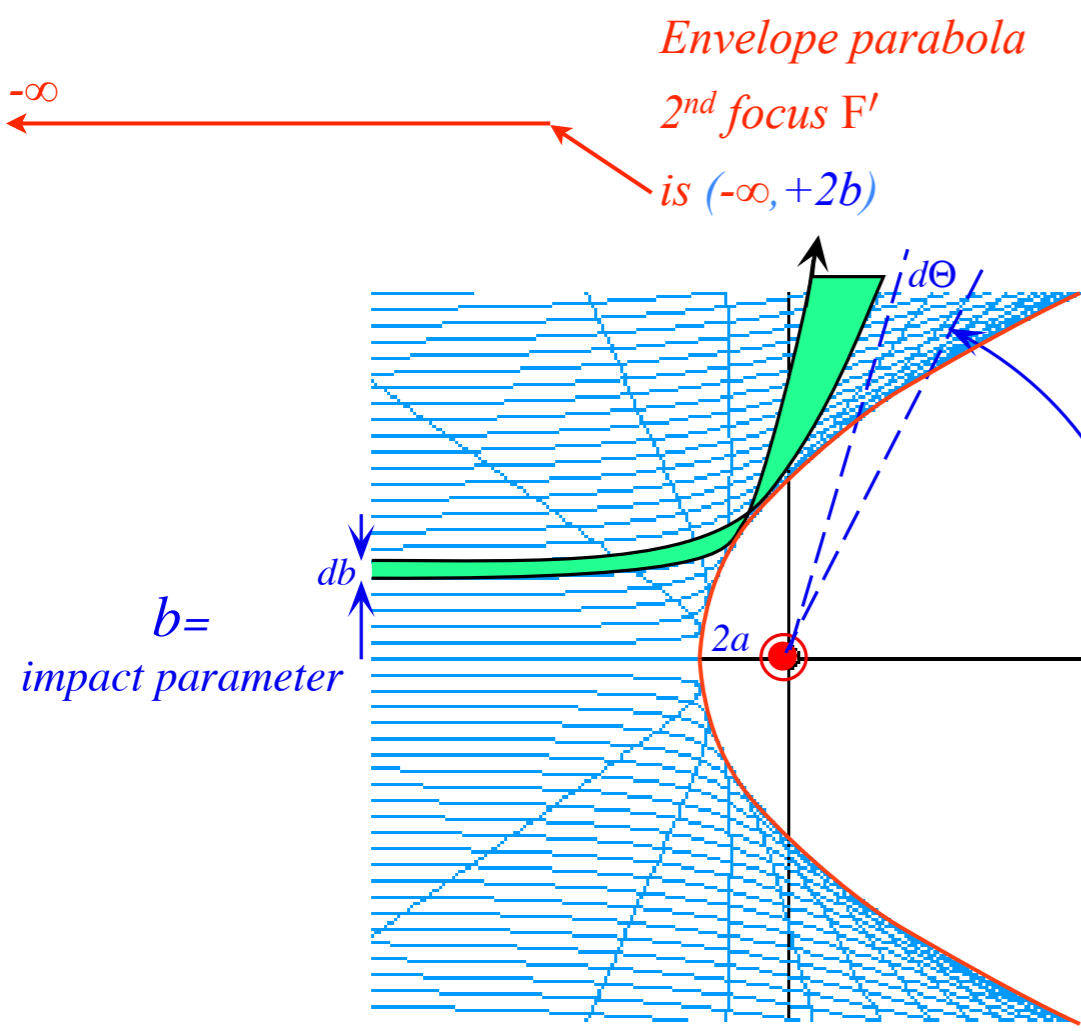


Particle-circle-center creeps toward focus?

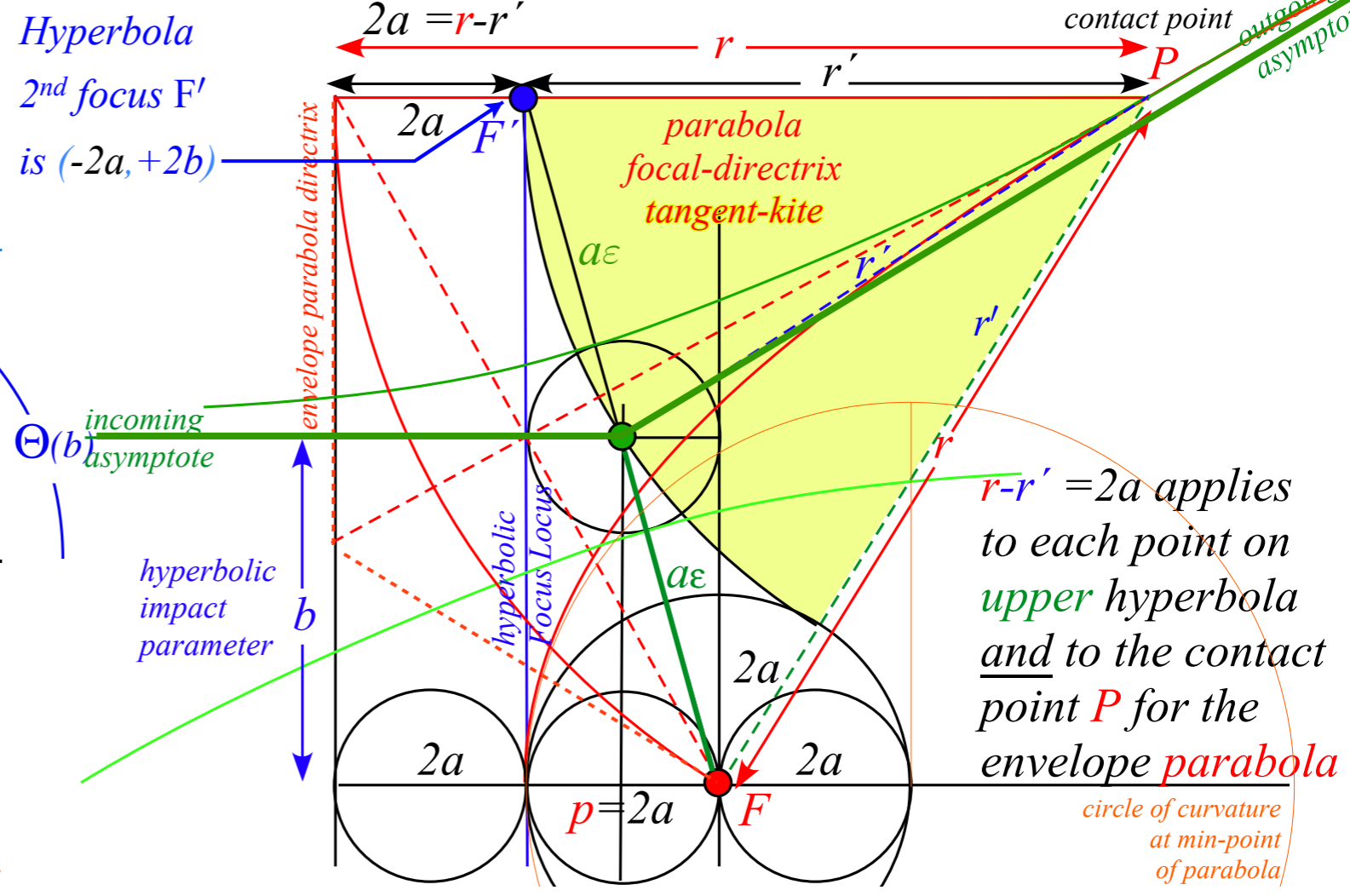
[CoulIt Web Simulation Rutherford Scattering](#)

Click and Drag in frame to set r(0) and p(0)

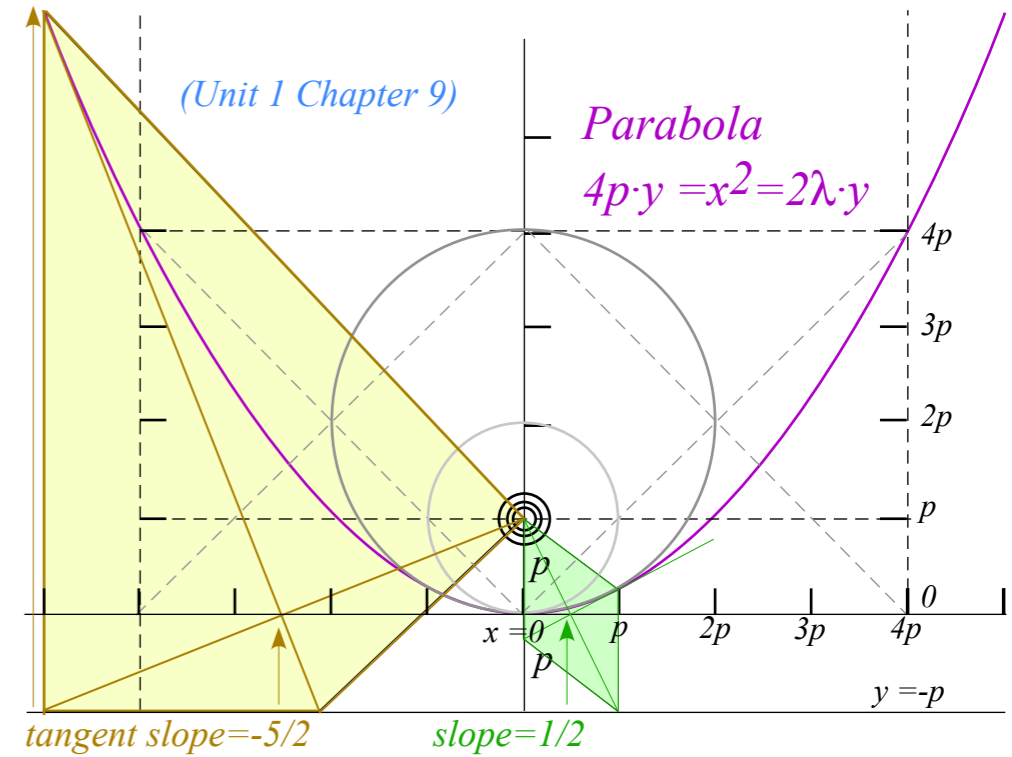
Rutherford scattering geometry



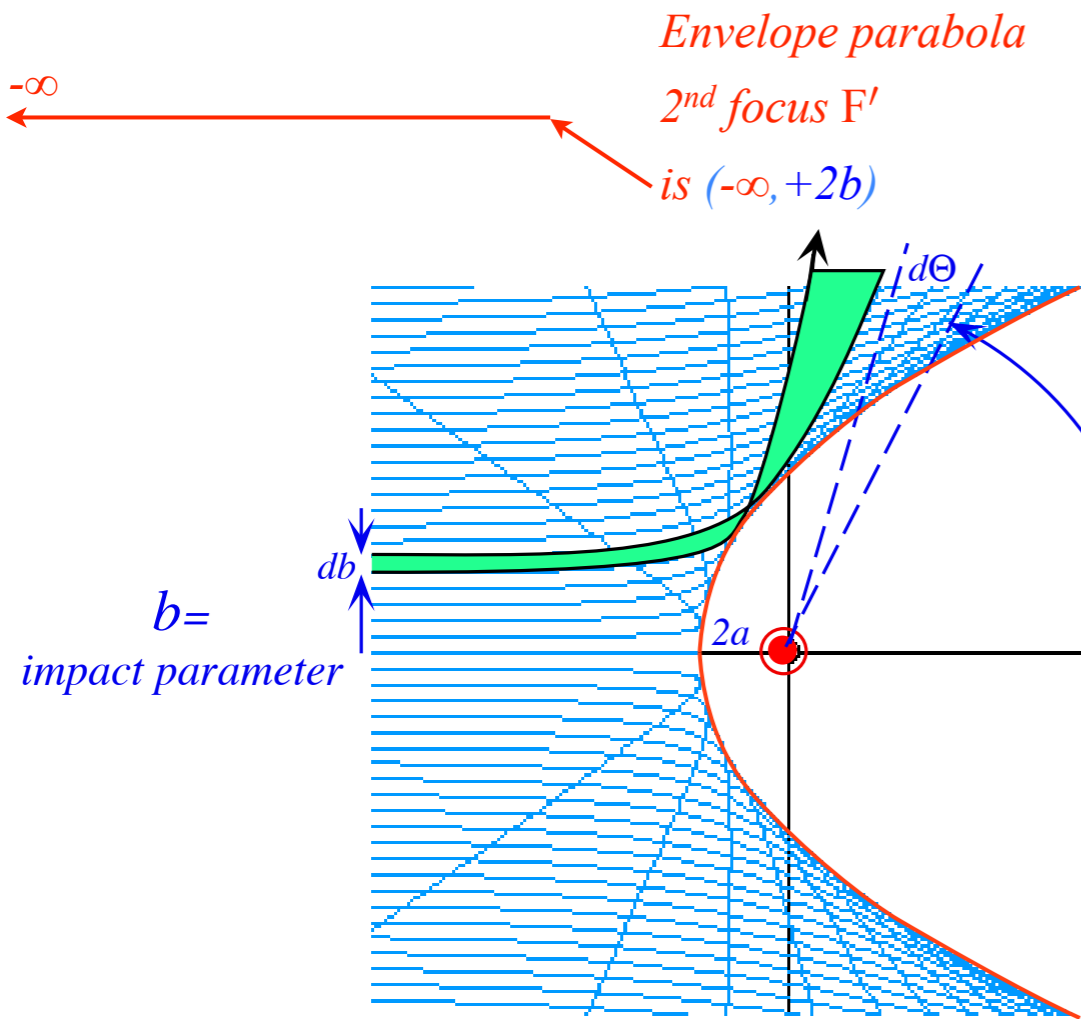
"Kite" geometry of envelope parabola



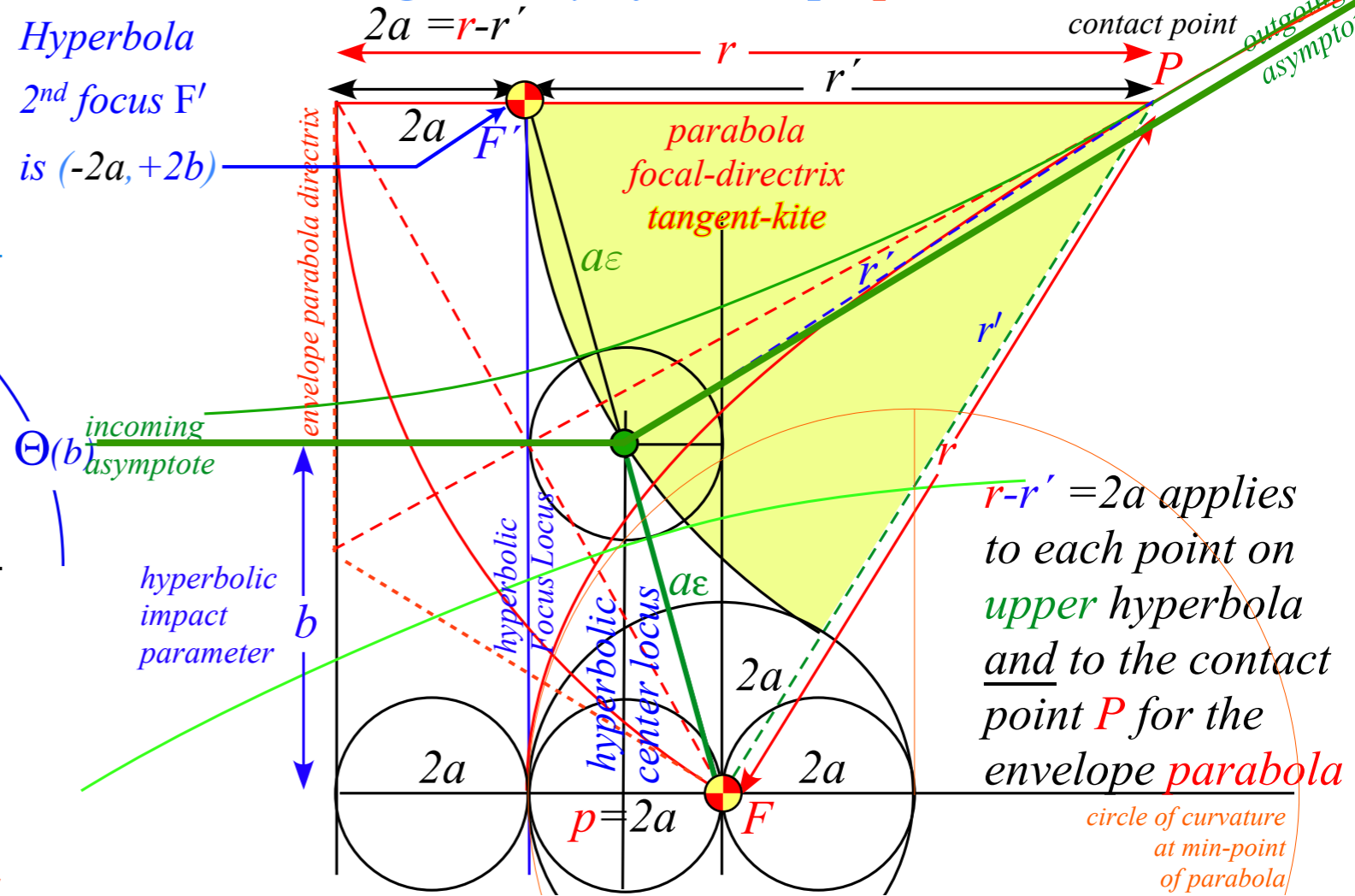
Recall parabolic "kite" geometry



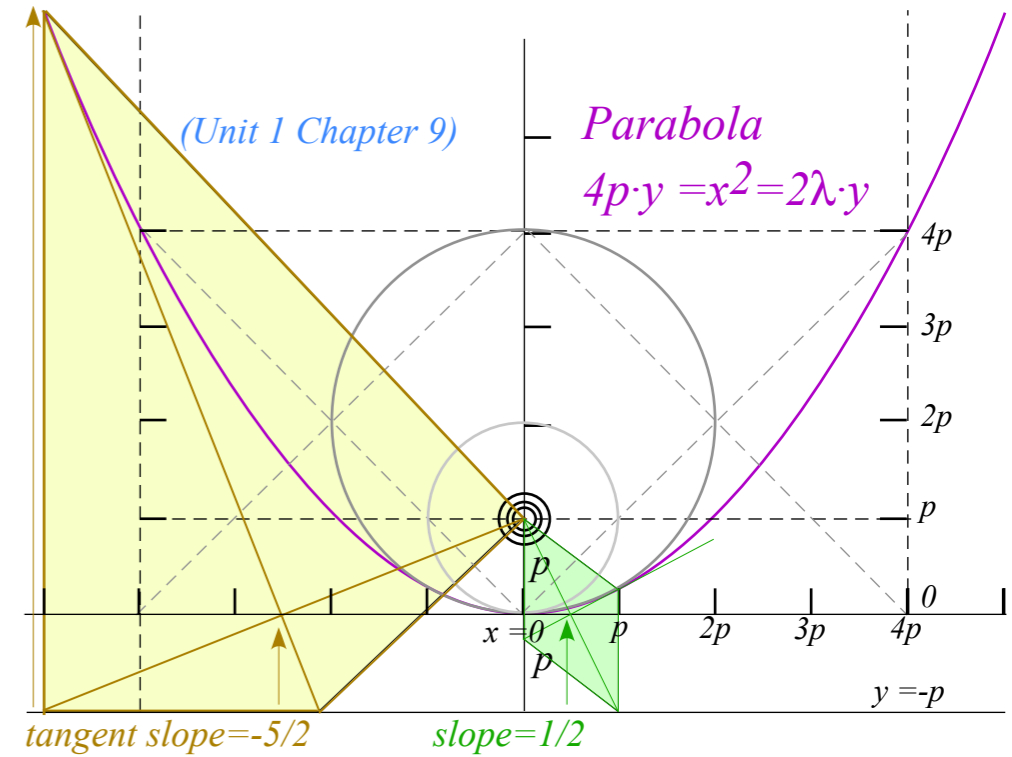
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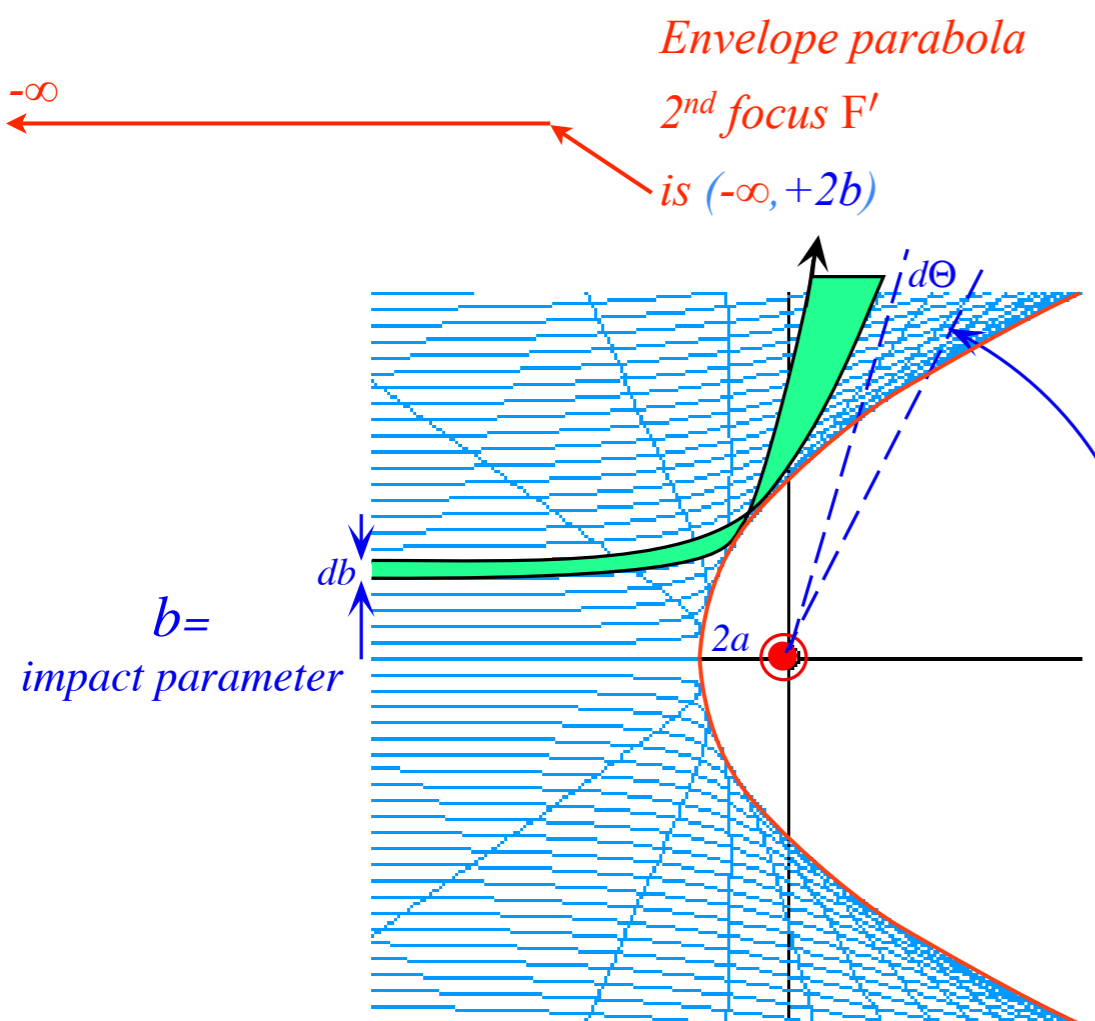
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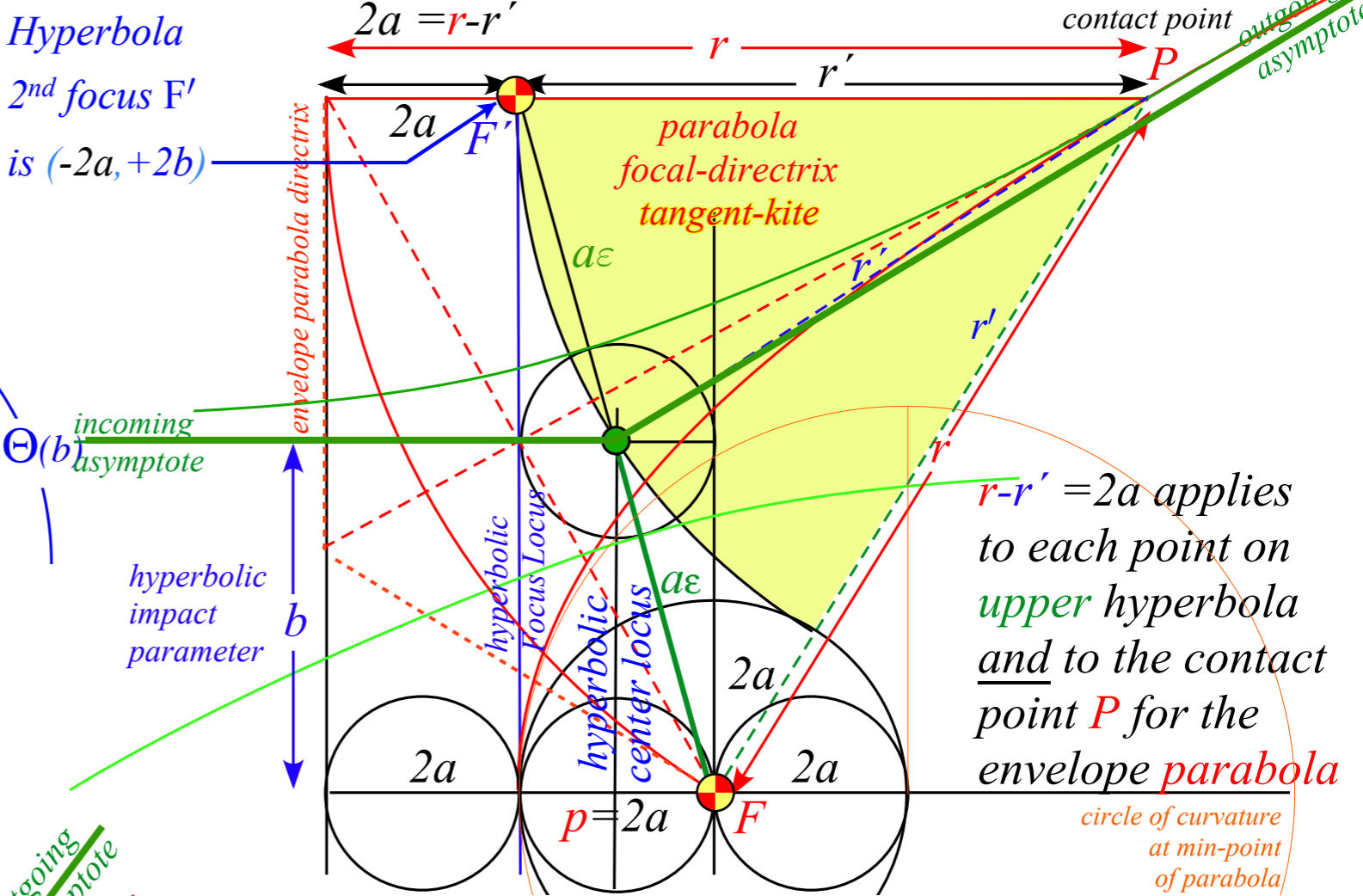
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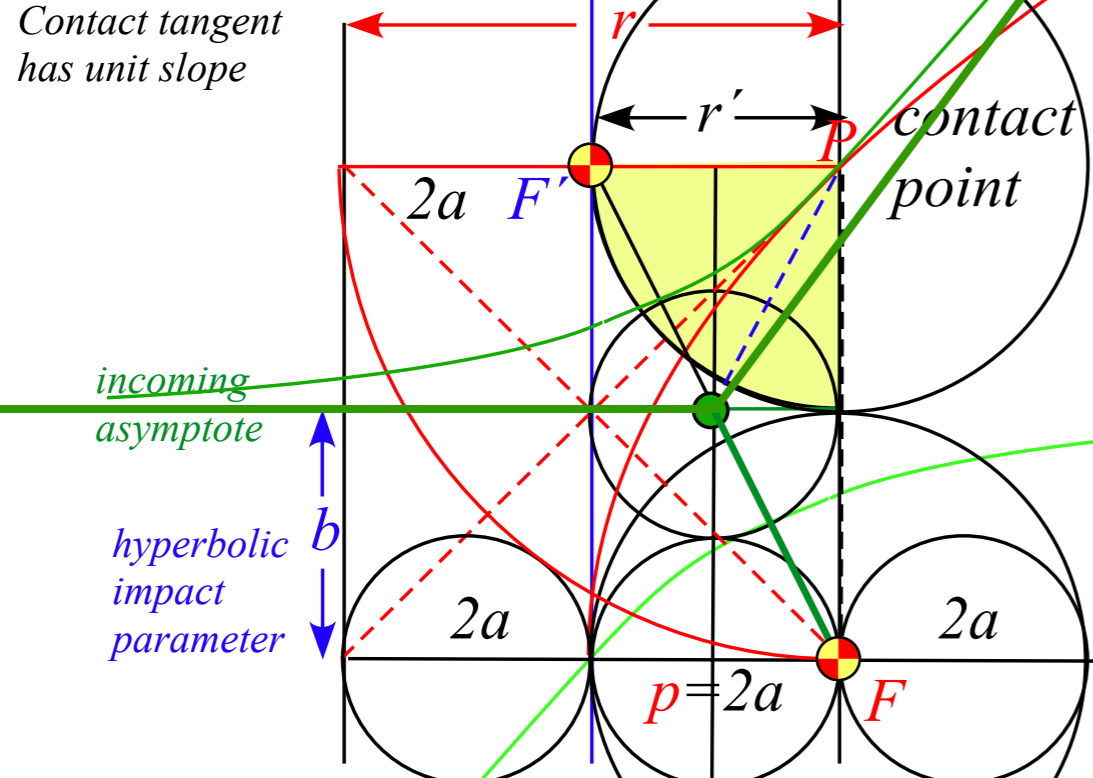
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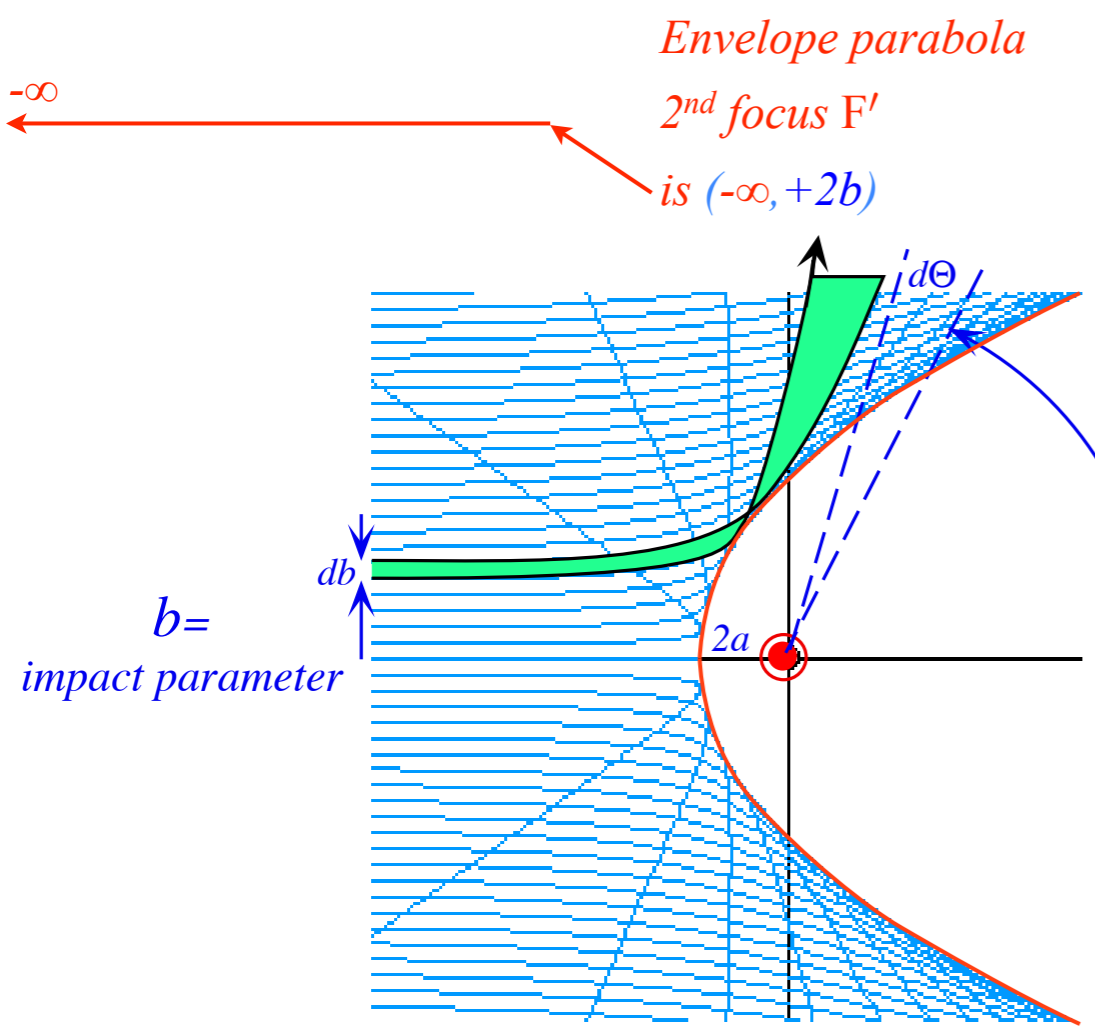


Special case: $b = 2a$

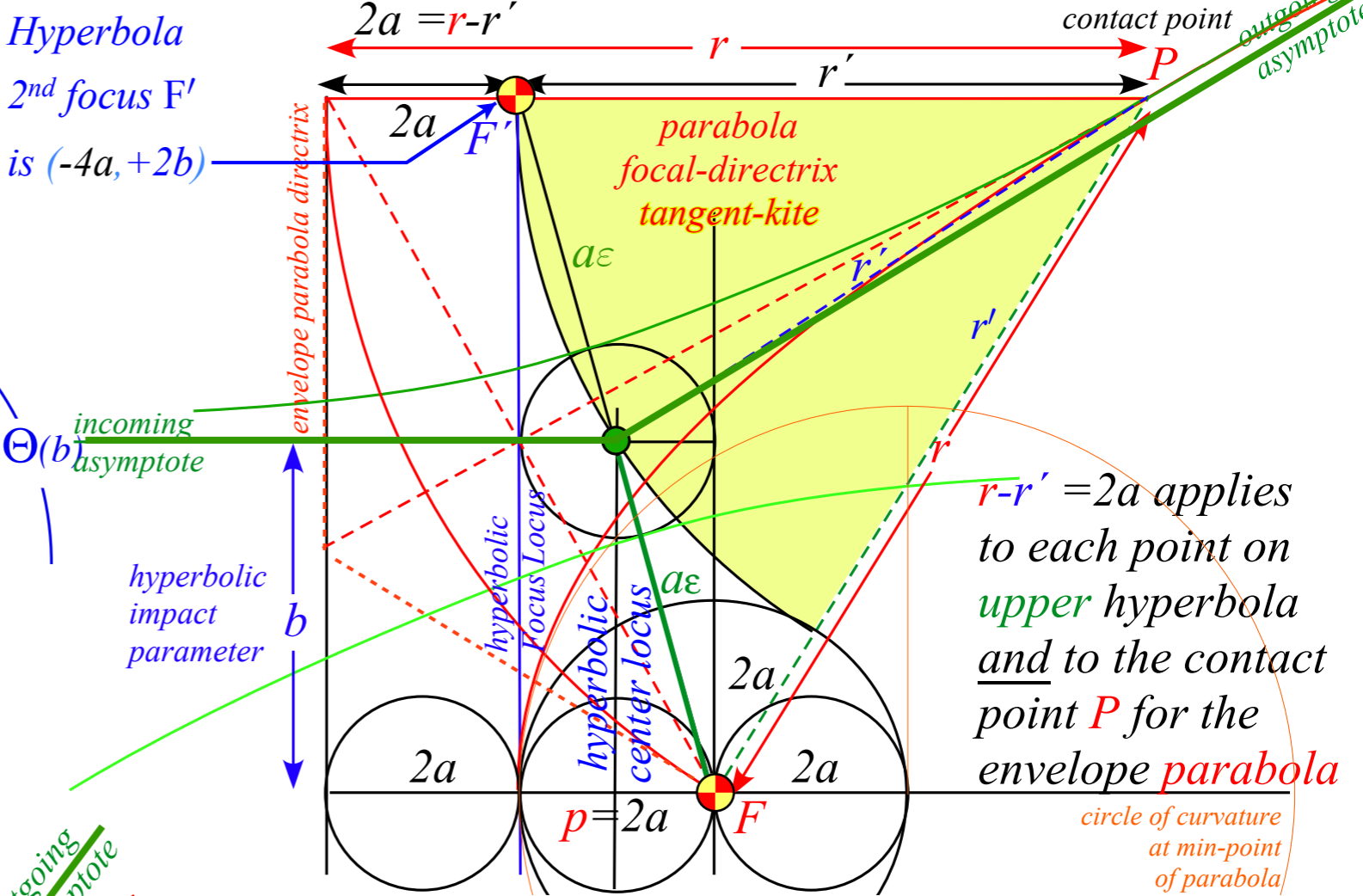


Parabola
contacts Rutherford Hyperbolas of various b at the point where they intersect with equal slope

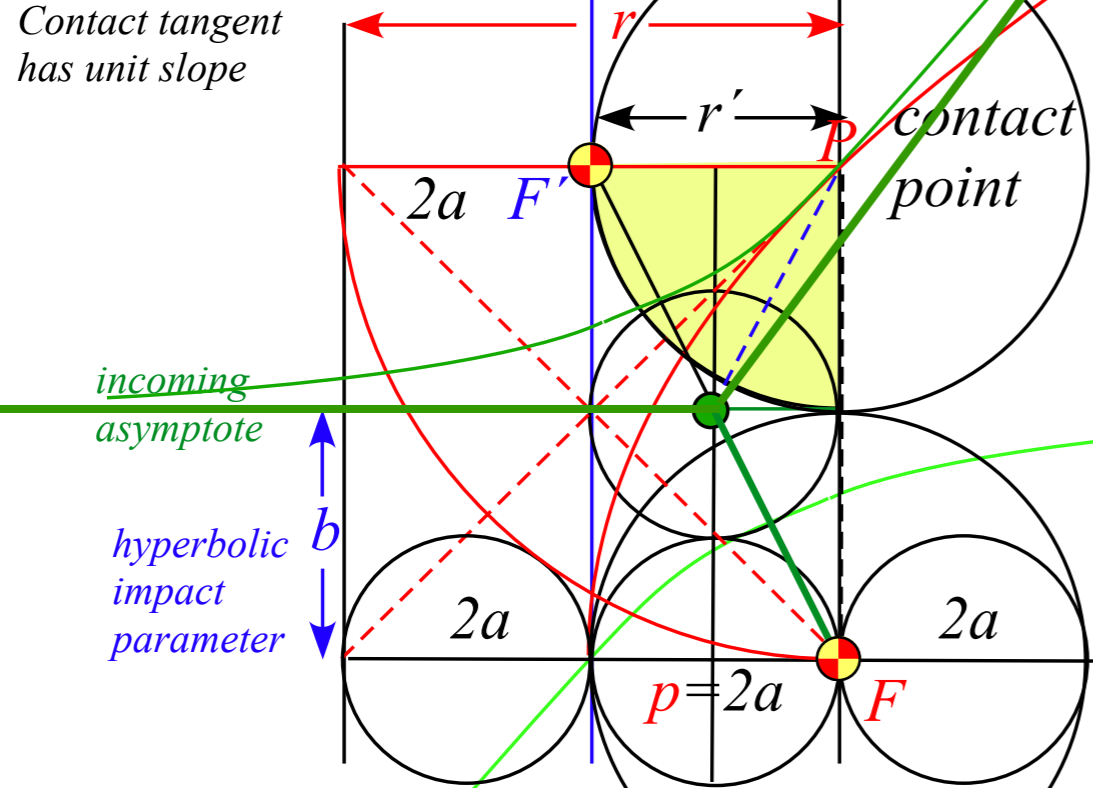
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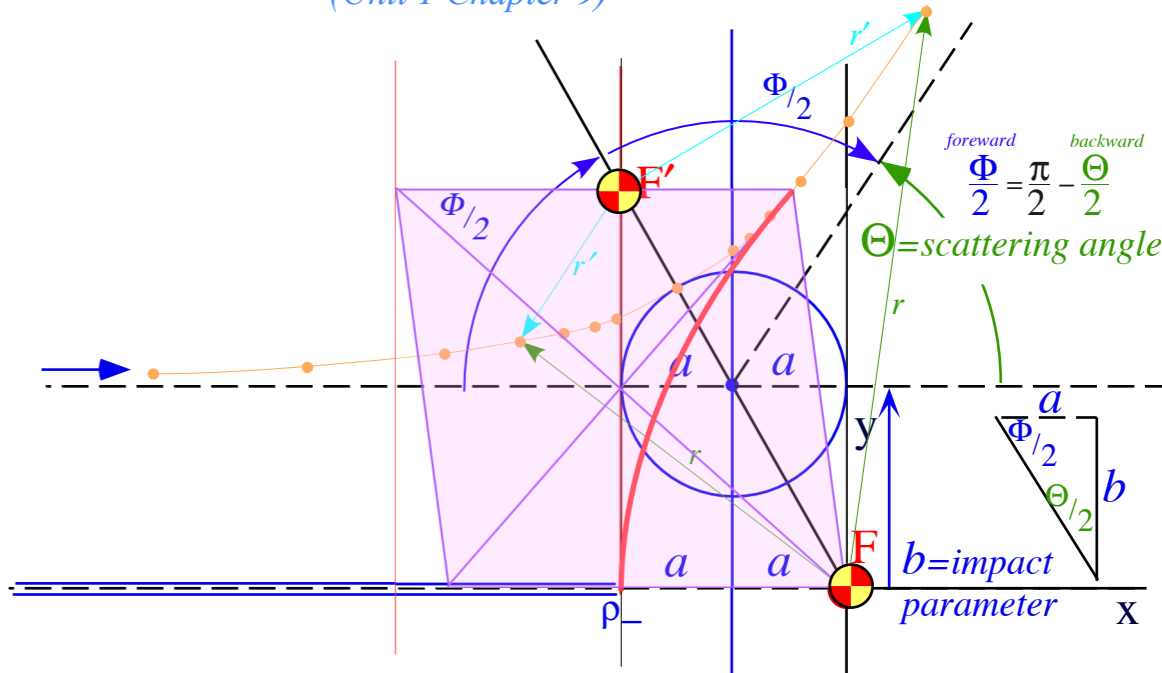
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Parabola contacts Rutherford Hyperbolas of various b at the point where they intersect with equal slope

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(Unit 1 Chapter 9)



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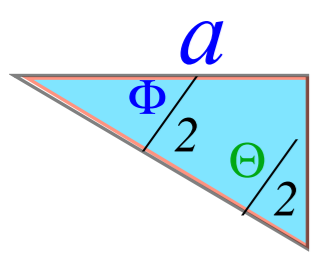
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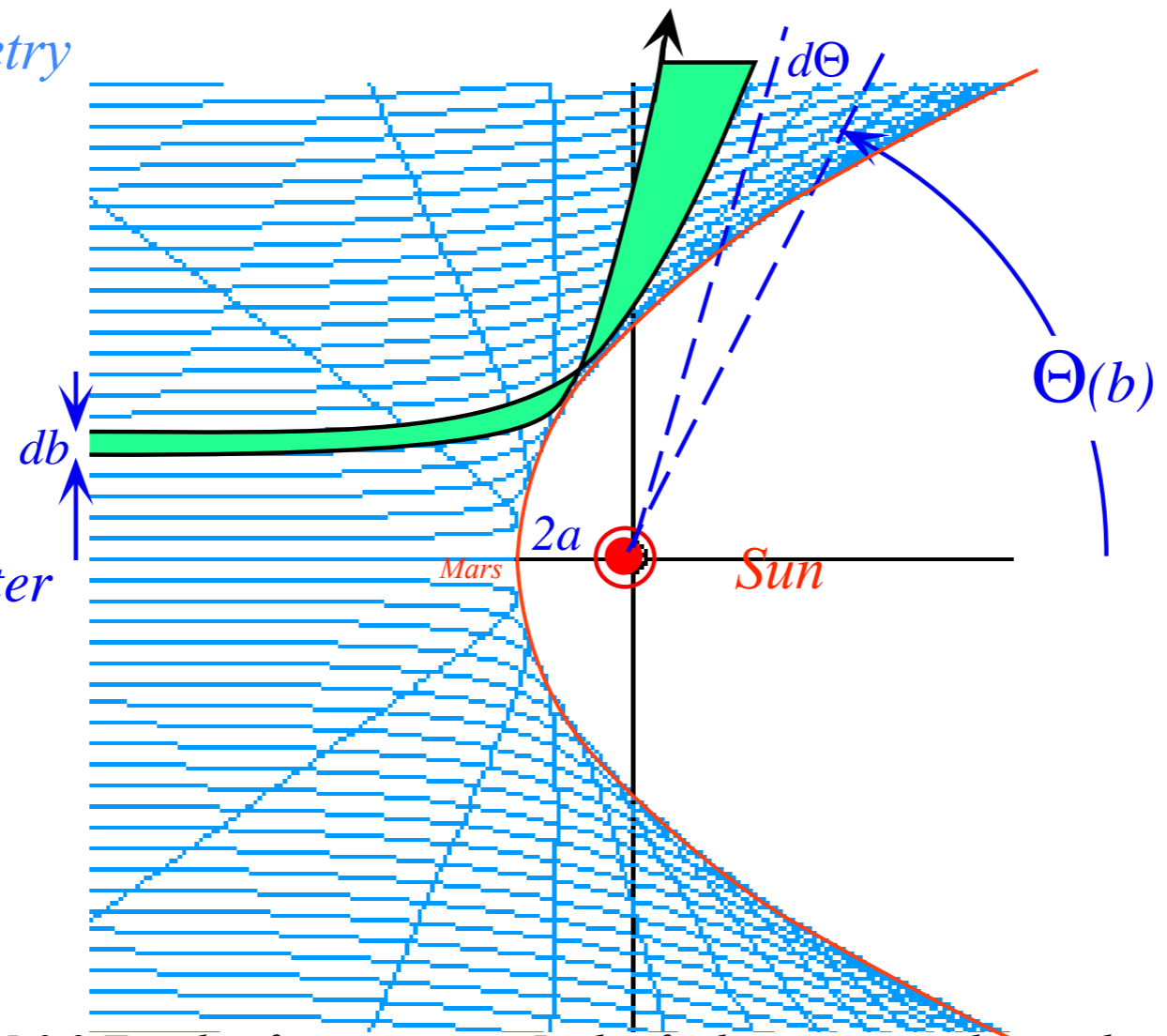
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$b =$
impact parameter

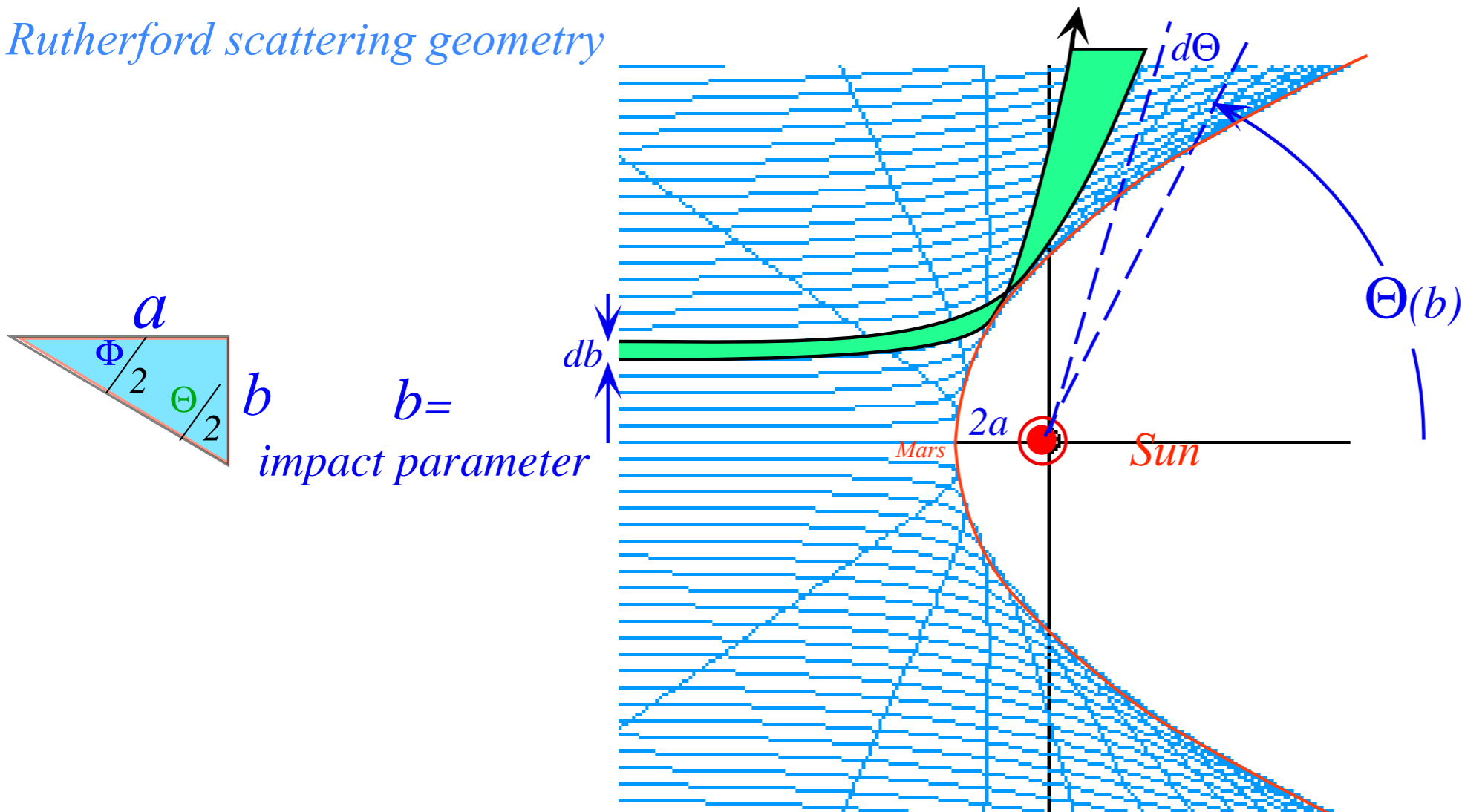


Also: Approximate model of deep-space H-atom scattering from solar wind as our Sun travels around galaxy. Lyman- α shock wave found just inside Mars orbital radius $2a \sim 1.2 \text{ Au}$.

Fig. 5.3.2 Family of iso-energetic Rutherford scattering orbits with varying impact parameter.

Incremental window $d\sigma = b \cdot db$ normal to beam axis at $x = -\infty$ scatters to area $dA = R^2 \sin \Theta d\Theta d\varphi = R^2 d\Omega$ onto a sphere at $R = +\infty$ where is called the *incremental solid angle* $d\Omega = \sin \Theta d\Theta d\varphi$

Rutherford scattering geometry



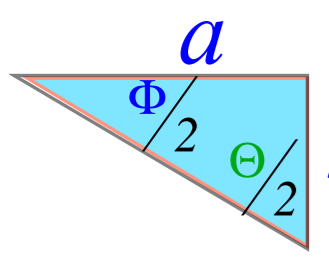
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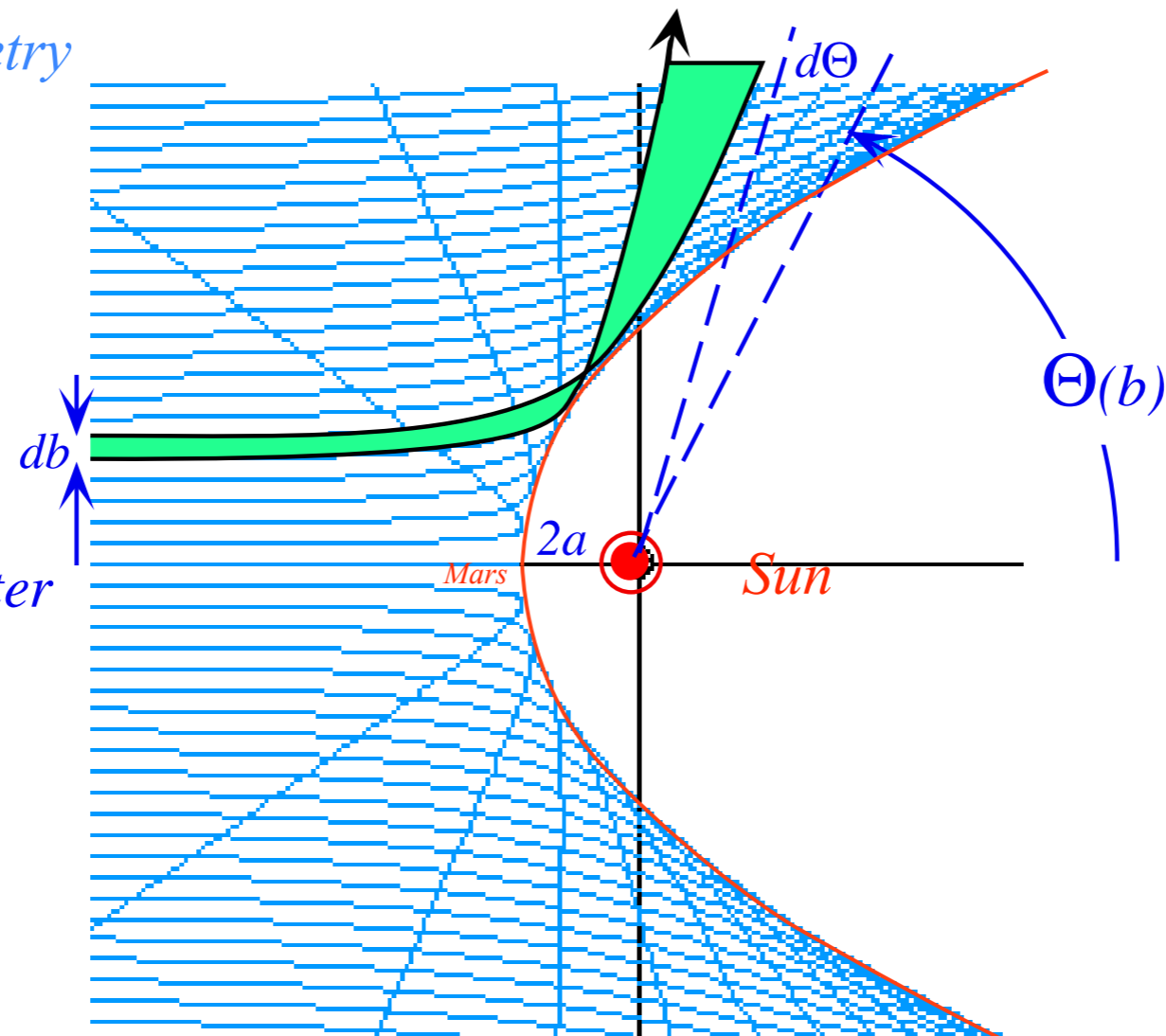
Ratio $\frac{d\sigma}{d\Omega} = \frac{b db d\varphi}{\sin \Theta d\Theta d\varphi} = \frac{b}{\sin \Theta} \frac{db}{d\Theta}$ is called the *differential scattering cross-section (DSC)*

Rutherford scattering geometry



$b =$
impact parameter

$$\frac{b}{a} = \cot \frac{\Theta}{2}$$



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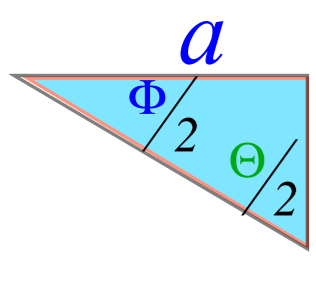
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Ratio $\frac{d\sigma}{d\Omega} = \frac{b db d\varphi}{\sin \Theta d\Theta d\varphi} = \frac{b}{\sin \Theta} \frac{db}{d\Theta}$ is called the *differential scattering cross-section (DSC)*

Geometry: $b = a \cot \frac{\Theta}{2}$

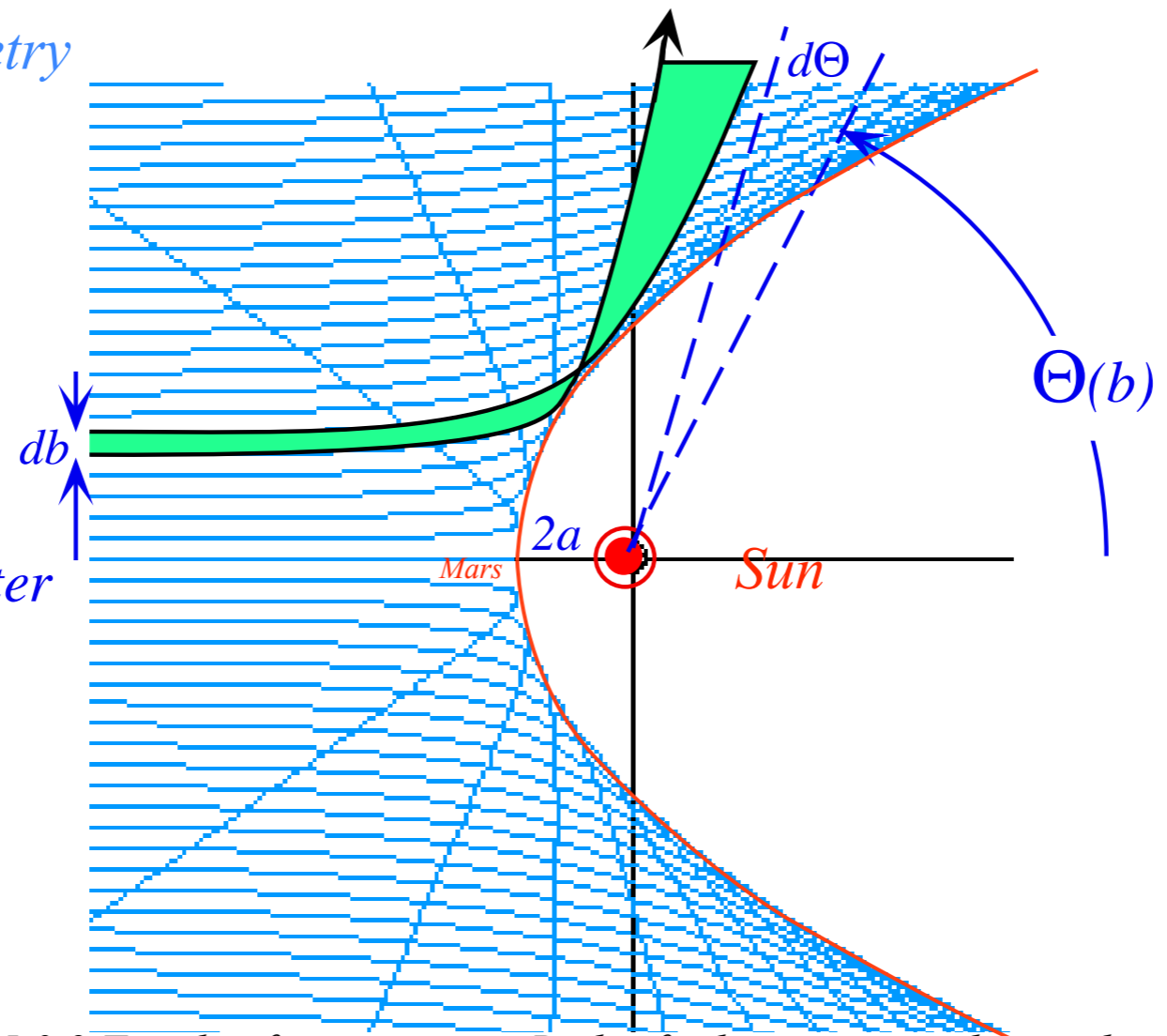
with: $\frac{db}{d\Theta} = \frac{-a}{2} \csc^2 \frac{\Theta}{2}$

Rutherford scattering geometry



$b =$
impact parameter

$$\frac{b}{a} = \cot \frac{\Theta}{2}$$



Also: Approximate model of deep-space H-atom scattering from solar wind as our Sun travels around galaxy. Lyman- α shock wave found just inside Mars orbital radius $2a \sim 1.2Au$.

Fig. 5.3.2 Family of iso-energetic Rutherford scattering orbits with varying impact parameter.

Incremental window $d\sigma = b \cdot db$ normal to beam axis at $x = -\infty$ scatters to area $dA = R^2 \sin \Theta d\Theta d\varphi = R^2 d\Omega$ onto a sphere at $R = +\infty$ where is called the *incremental solid angle* $d\Omega = \sin \Theta d\Theta d\varphi$

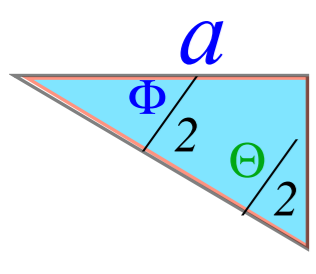
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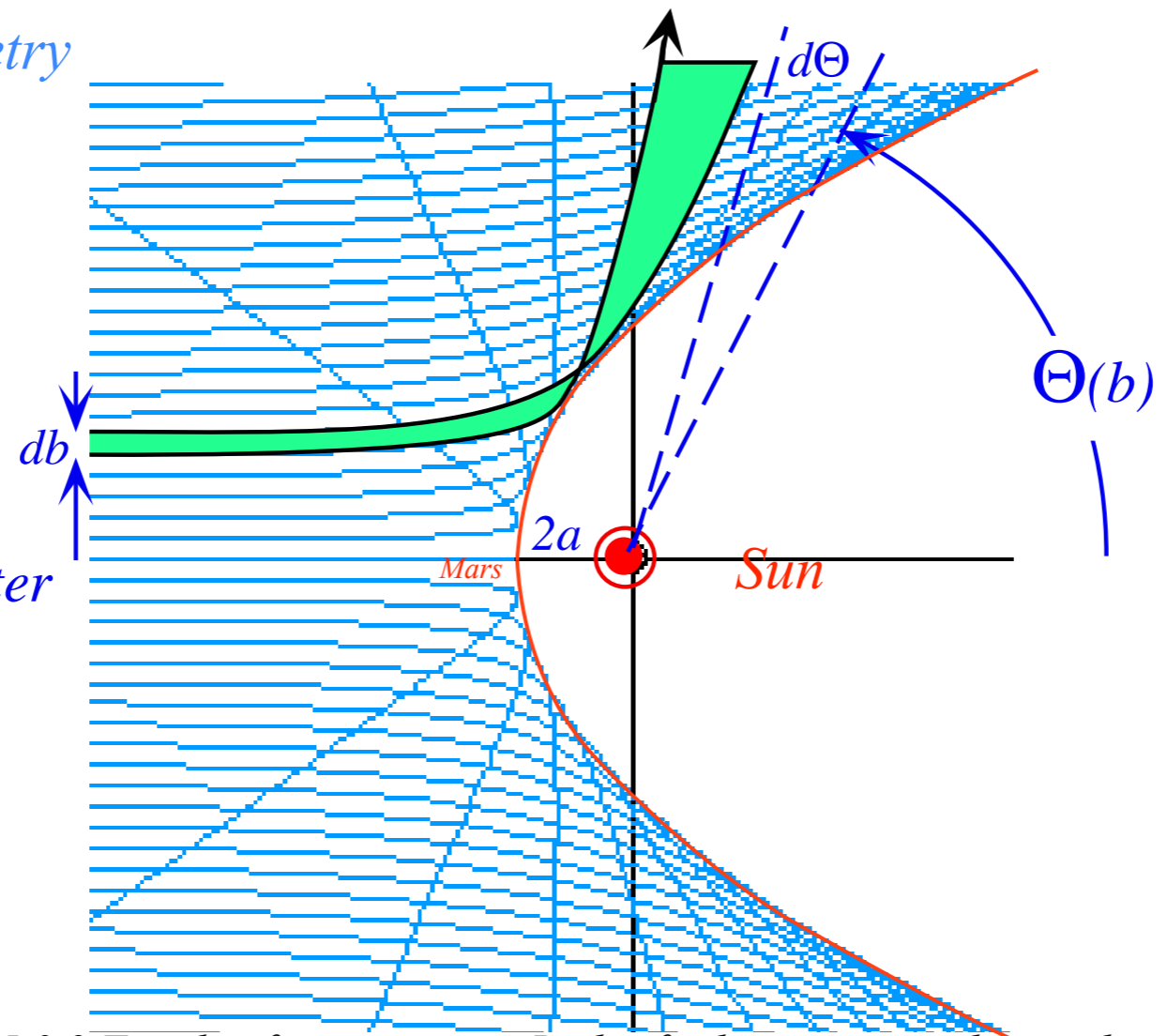
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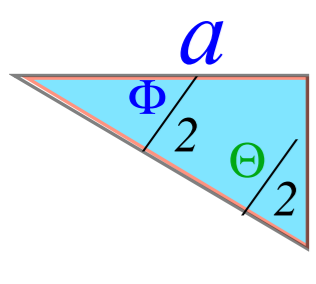
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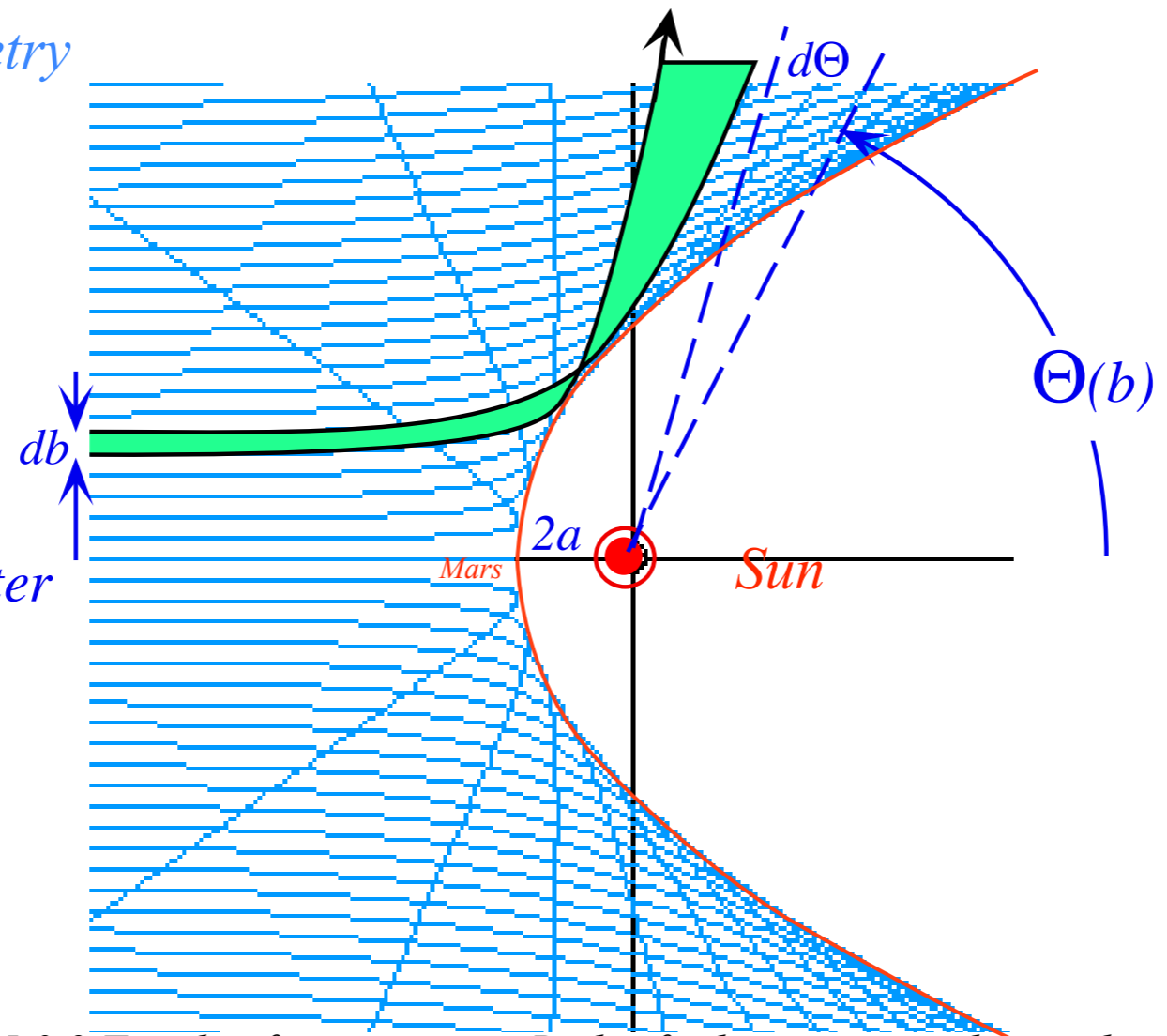
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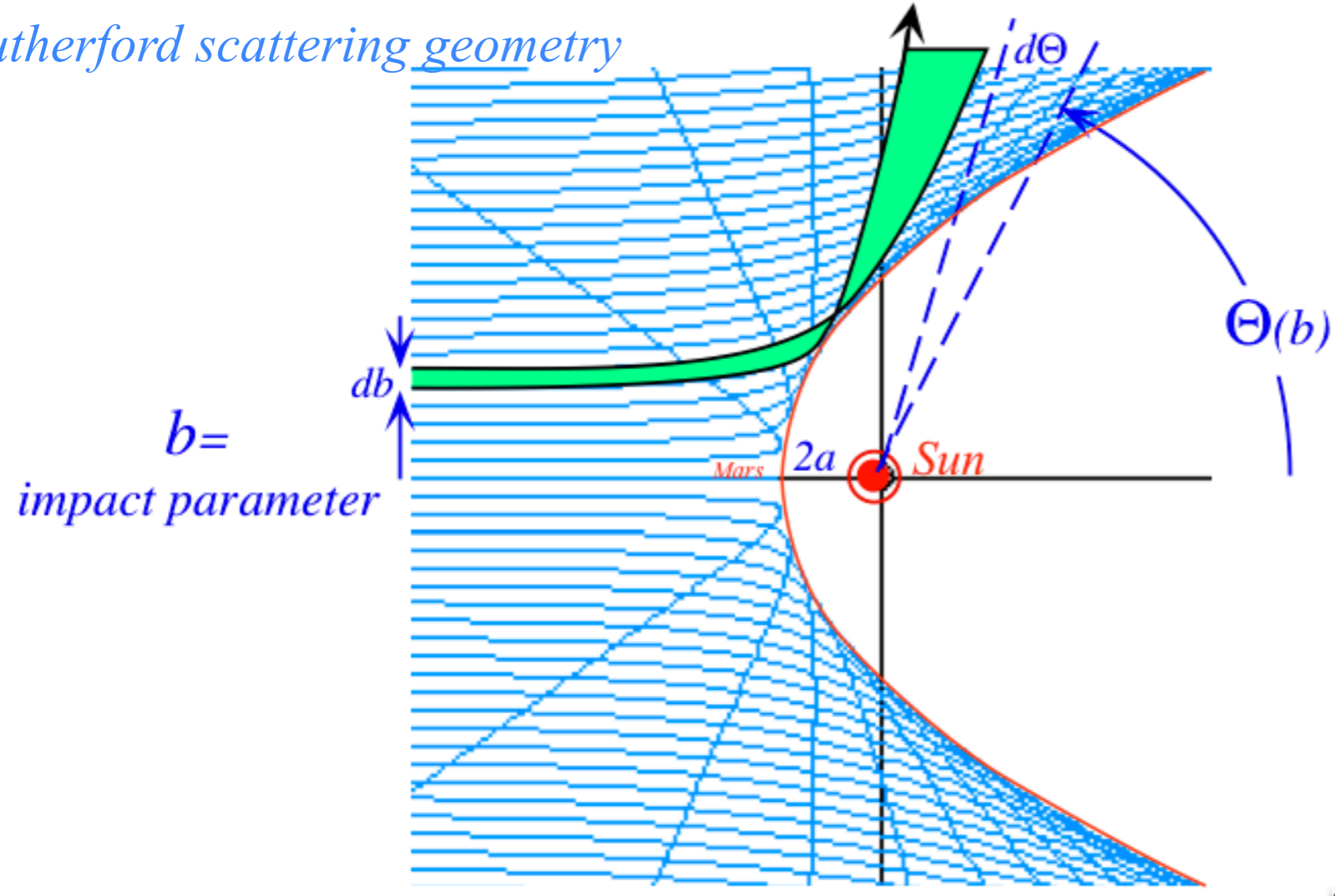
$$\frac{d\sigma}{d\Omega} = \frac{-a^2 \cos \frac{\Theta}{2}}{2 \sin \Theta \sin^3 \frac{\Theta}{2}} = \frac{-k^4}{16E^2} \sin^{-4} \frac{\Theta}{2}$$

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This classical result agrees exactly with 1st Born approximation to quantum Coulomb DSC!

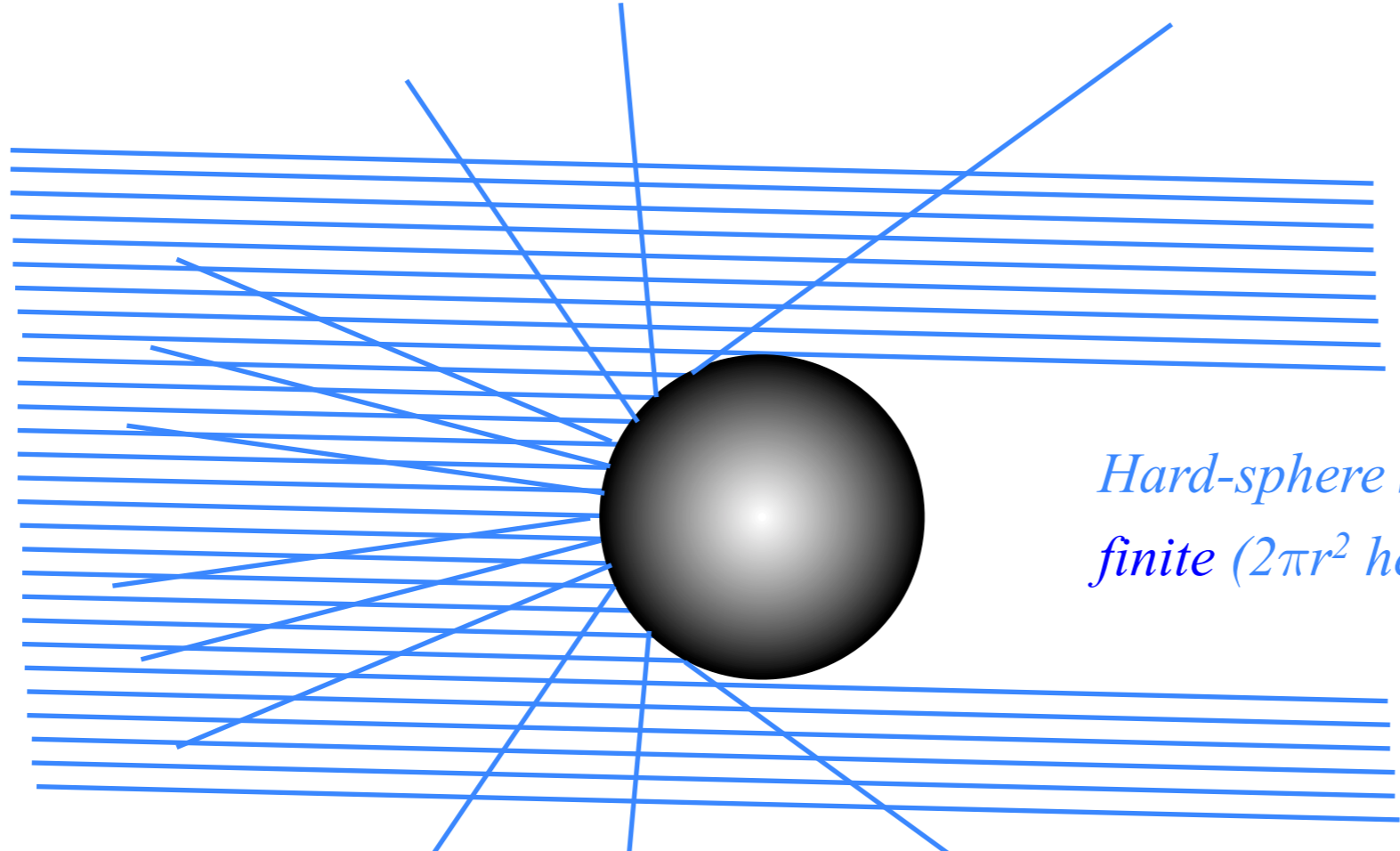
Rutherford scattering geometry



Two Extremes:

Rutherford (Coulomb) scattering has infinite (∞) total cross section

$$\sigma = \int d\Omega \frac{d\sigma}{d\Omega} = \int d\Omega \frac{k^4}{16E^2} \sin^{-4} \frac{\Theta}{2} = \infty$$



Hard-sphere scattering has finite ($2\pi r^2$ here) total cross section

CouIt Web Simulation
Hard Sphere

Rutherford scattering and hyperbolic orbit geometry

Backward vs forward scattering angles and orbit construction example

Parabolic “kite” and orbital envelope geometry

Differential and total scattering cross-sections

➔ *Eccentricity vector $\boldsymbol{\varepsilon}$ and (ε, λ) -geometry of orbital mechanics*

Projection $\boldsymbol{\varepsilon} \cdot \mathbf{r}$ geometry of $\boldsymbol{\varepsilon}$ -vector and orbital radius \mathbf{r}

Review and connection to usual orbital algebra (previous lecture)

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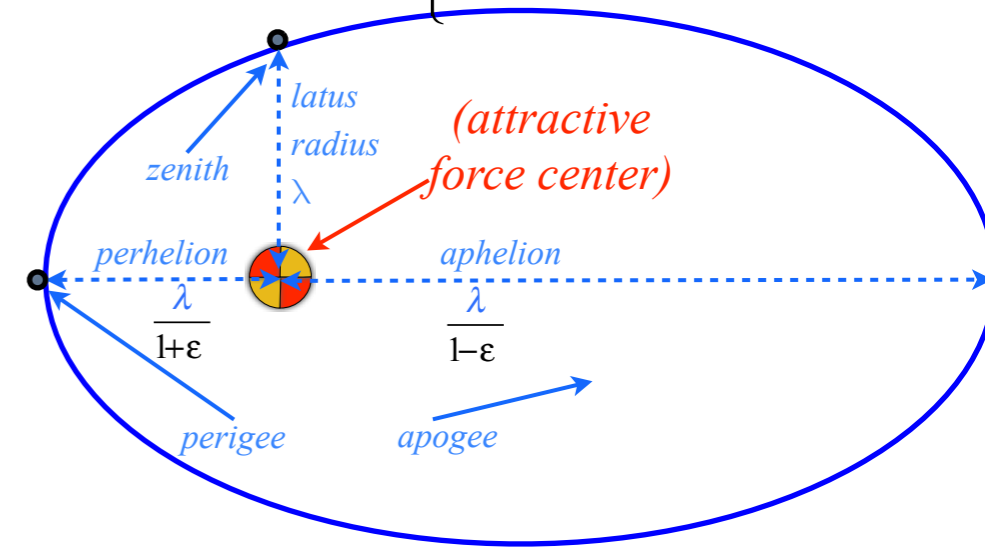
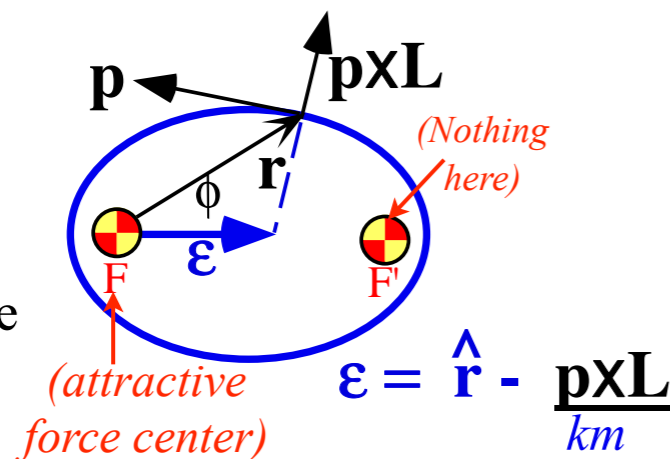
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(a) Attractive ($k > 0$)
Elliptic ($E < 0$)



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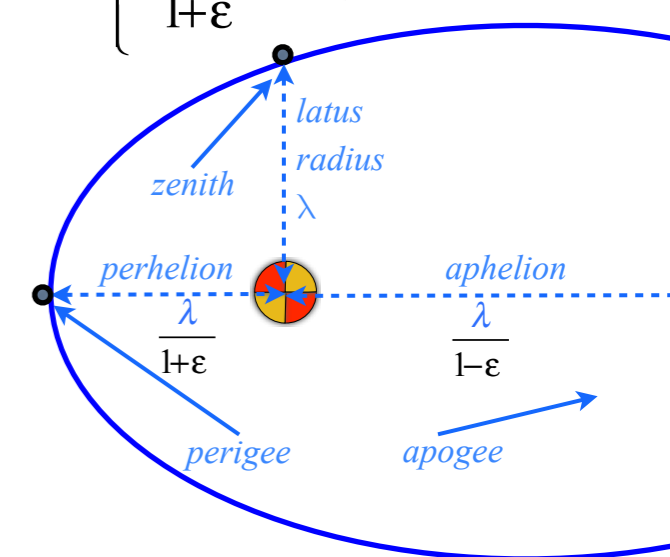
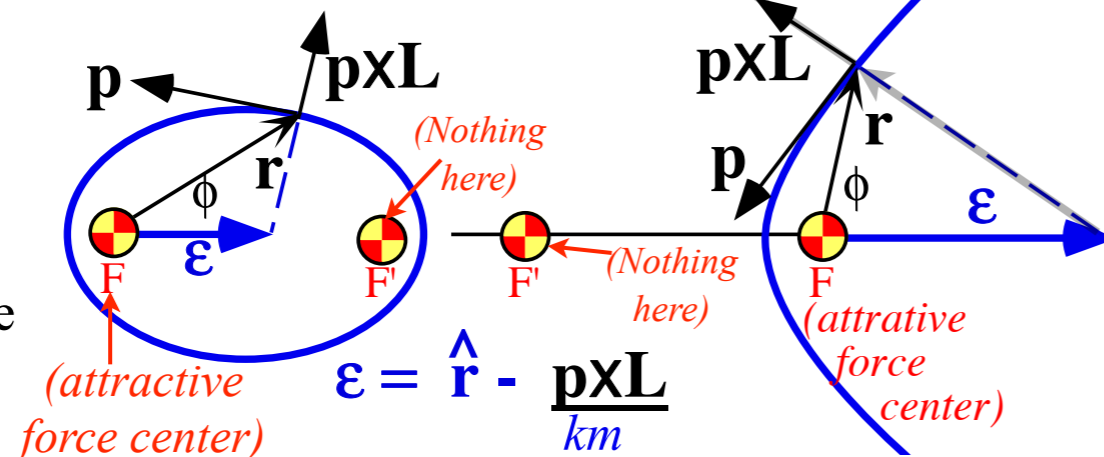
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Elliptic ($E<0$)

(b) Attractive ($k>0$)
Hyperbolic ($E>0$)



(Rotational momentum $\mathbf{L} = \mathbf{r} \times \mathbf{p}$ is normal to the orbit plane.)

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$$\boldsymbol{\varepsilon} \cdot \mathbf{r} = \frac{\mathbf{r} \cdot \mathbf{r}}{r} - \frac{\mathbf{r} \cdot \mathbf{p} \times \mathbf{L}}{km} = r - \frac{\mathbf{r} \times \mathbf{p} \cdot \mathbf{L}}{km} = r - \frac{\mathbf{L} \cdot \mathbf{L}}{km}$$

Let angle ϕ be angle between $\boldsymbol{\varepsilon}$ and radial vector \mathbf{r}

$$\varepsilon r \cos \phi = r - \frac{L^2}{km} \quad \text{or:} \quad r = \frac{L^2/km}{1 - \varepsilon \cos \phi}$$

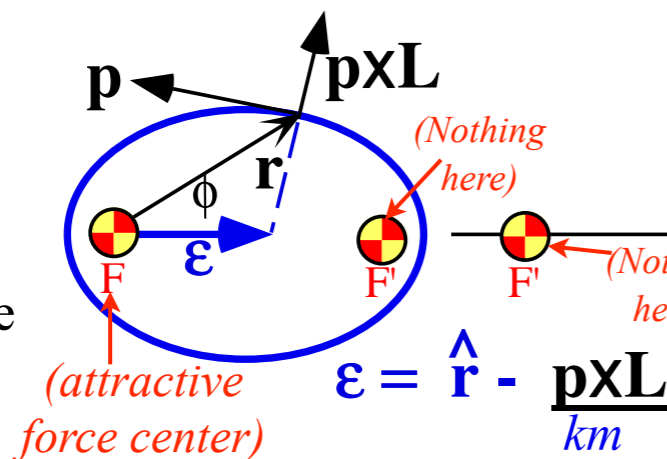
For $\lambda = L^2/km$ that matches: $r = \frac{\lambda}{1 - \varepsilon \cos \phi} =$

...or of $\boldsymbol{\varepsilon}$ with momentum vector \mathbf{p} :

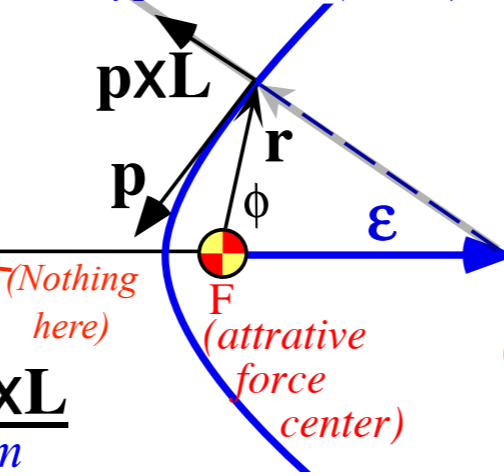
$$\boldsymbol{\varepsilon} \cdot \mathbf{p} = \frac{\mathbf{p} \cdot \mathbf{r}}{r} - \frac{\mathbf{p} \cdot \mathbf{p} \times \mathbf{L}}{km} = \mathbf{p} \cdot \hat{\mathbf{r}} = p_r$$

$$\left. \begin{array}{l} \frac{\lambda}{1-\varepsilon} \text{ if: } \phi=0 \text{ apogee} \\ \lambda \text{ if: } \phi=\frac{\pi}{2} \text{ zenith} \\ \frac{\lambda}{1+\varepsilon} \text{ if: } \phi=\pi \text{ perigee} \end{array} \right\}$$

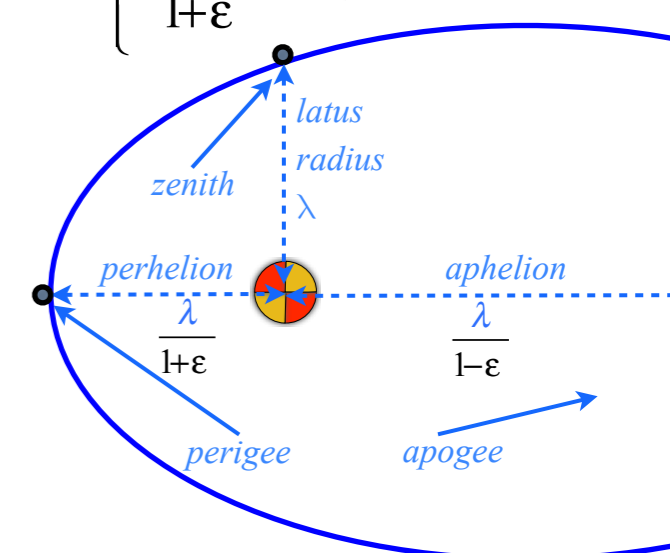
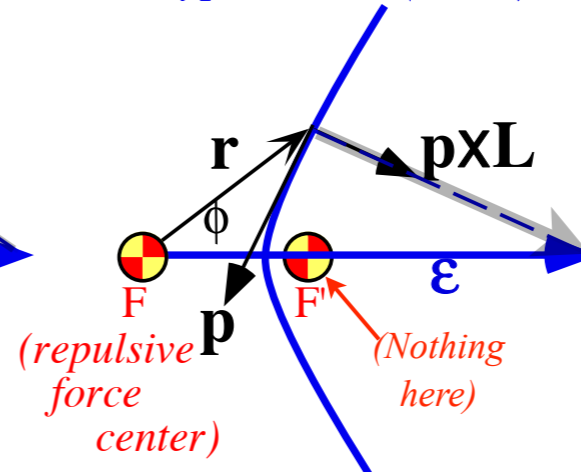
(a) Attractive ($k>0$)
Elliptic ($E<0$)



(b) Attractive ($k>0$)
Hyperbolic ($E>0$)



(c) Repulsive ($k<0$)
Hyperbolic ($E>0$)



(Rotational momentum $\mathbf{L} = \mathbf{r} \times \mathbf{p}$ is normal to the orbit plane.)

Rutherford scattering and hyperbolic orbit geometry

Backward vs forward scattering angles and orbit construction example

Parabolic “kite” and orbital envelope geometry

Differential and total scattering cross-sections

Eccentricity vector $\boldsymbol{\varepsilon}$ and (ε, λ) -geometry of orbital mechanics

Projection $\boldsymbol{\varepsilon} \cdot \mathbf{r}$ geometry of $\boldsymbol{\varepsilon}$ -vector and orbital radius \mathbf{r}

➔ *Review and connection to usual orbital algebra (previous lecture)*

Projection $\boldsymbol{\varepsilon} \cdot \mathbf{p}$ geometry of $\boldsymbol{\varepsilon}$ -vector and momentum $\mathbf{p}=m\mathbf{v}$

General geometric orbit construction using $\boldsymbol{\varepsilon}$ -vector and (γ, R) -parameters

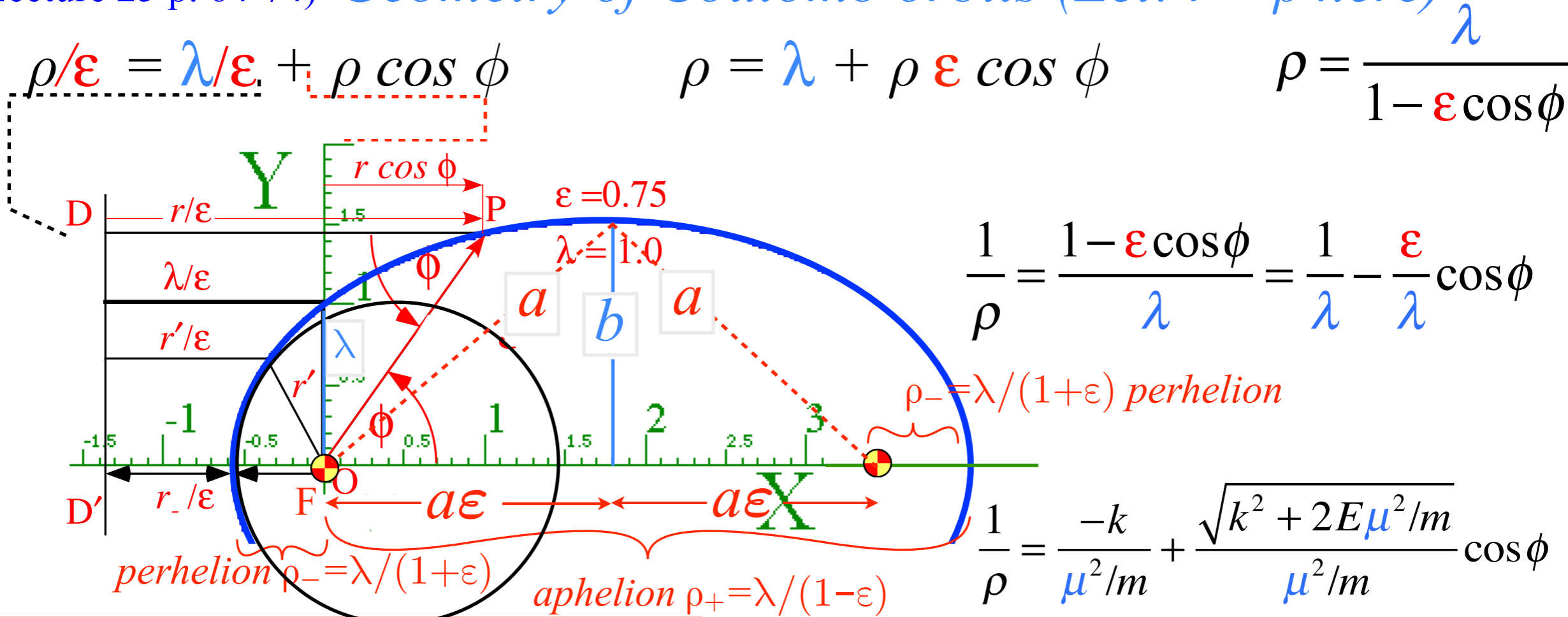
Derivation of $\boldsymbol{\varepsilon}$ -construction by analytic geometry

Coulomb orbit algebra of $\boldsymbol{\varepsilon}$ -vector and Kepler dynamics of momentum $\mathbf{p}=m\mathbf{v}$

Example of complete (\mathbf{r}, \mathbf{p}) -geometry of elliptical orbit

Connection formulas for (γ, R) -parameters with (a, b) and (ε, λ)

(From Lecture 25 p. 64-74) *Geometry of Coulomb orbits (Let: $r = \rho$ here)*



All conics defined by:
Defining eccentricity ϵ
Distance to Focal-point = ϵ · Distance to Directrix-line

Major axis: $\rho_+ + \rho_- = 2a$
 $\rho_+ + \rho_- = [\lambda(1+\epsilon) + \lambda(1-\epsilon)] / (1-\epsilon^2) = 2\lambda / |1-\epsilon^2|$
Focal axis: $\rho_+ - \rho_- = 2a\epsilon$
 $\rho_+ - \rho_- = [\lambda(1+\epsilon) - \lambda(1-\epsilon)] / (1-\epsilon^2) = 2\lambda\epsilon / |1-\epsilon^2|$
Minor radius: $b = \sqrt{a^2 - a^2\epsilon^2} = \sqrt{a\lambda}$ (ellipse: $\epsilon < 1$)
Minor radius: $b = \sqrt{a^2\epsilon^2 - a^2} = \sqrt{\lambda a}$ (hyperb: $\epsilon > 1$)

(x, y) parameters	physical constants	(r, ϕ) parameters
major radius $a = \frac{k}{2E}$	Energy $E = \frac{k}{2a}$	$\epsilon = \sqrt{\frac{k^2 m + 2\mu^2 E}{k^2 m}} = \sqrt{1 \pm \frac{b^2}{a^2}}$
minor radius $b = \frac{\mu}{\sqrt{2m E }}$	Orbital Momentum $\mu = \sqrt{km\lambda}$	eccentricity
		latus radius $\lambda = \frac{\mu^2}{km} = \frac{b^2}{a}$

$\epsilon^2 = 1 - \frac{b^2}{a^2}$ (ellipse: $\epsilon < 1$) $\frac{b}{a} = \sqrt{1 - \epsilon^2}$
 $\epsilon^2 = 1 + \frac{b^2}{a^2}$ (hyperbola: $\epsilon > 1$) $\frac{b}{a} = \sqrt{\epsilon^2 - 1}$
 $\lambda = a(1 - \epsilon^2)$ (ellipse: $\epsilon < 1$)
 $\lambda = a(\epsilon^2 - 1)$ (hyperb: $\epsilon > 1$)

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Review and connection to usual orbital algebra (previous lecture)

➔ *Projection $\boldsymbol{\varepsilon} \cdot \mathbf{p}$ geometry of $\boldsymbol{\varepsilon}$ -vector and momentum $\mathbf{p} = m\mathbf{v}$*

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Connection formulas for (γ, R) -parameters with (a, b) and (ε, λ)

Dot product of $\boldsymbol{\epsilon}$ with momentum vector \mathbf{p} :

$$\boldsymbol{\epsilon} \cdot \mathbf{p} = \frac{\mathbf{p} \cdot \mathbf{r}}{r} - \frac{\mathbf{p} \cdot \mathbf{p} \times \mathbf{L}}{km}$$

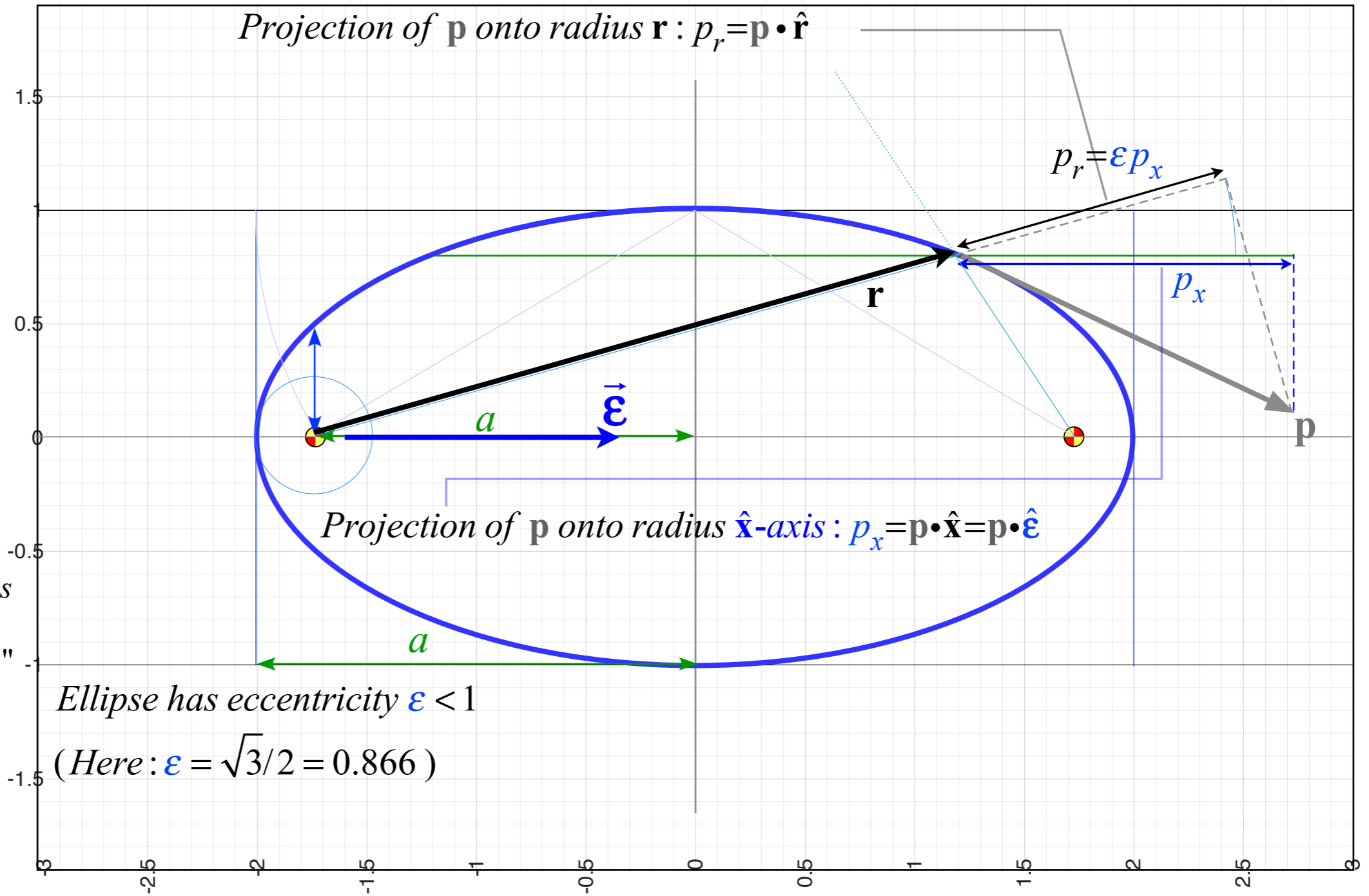
$$= \mathbf{p} \cdot \hat{\mathbf{r}} = p_r = \boldsymbol{\epsilon} p_x$$

This says:

"Projection p_r of \mathbf{p} onto radial \mathbf{r} or \mathbf{r}' lines equals eccentricity $\boldsymbol{\epsilon}$ times projection p_x of \mathbf{p} onto orbit major axis: ($\hat{\mathbf{x}} = \hat{\boldsymbol{\epsilon}}$)"

Ellipse has eccentricity $\boldsymbol{\epsilon} < 1$

(Here: $\boldsymbol{\epsilon} = \sqrt{3}/2 = 0.866$)

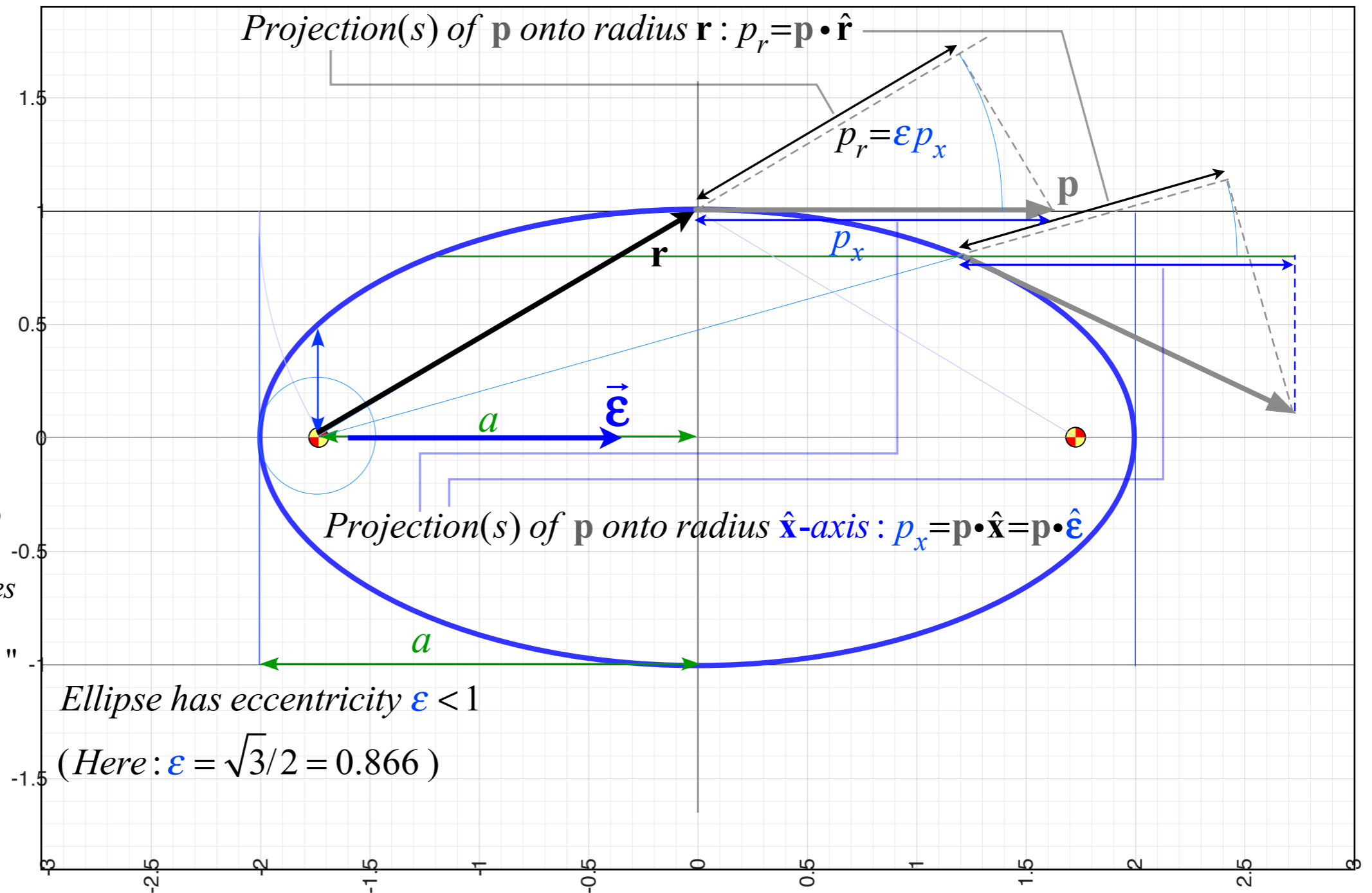


Dot product of $\boldsymbol{\epsilon}$ with momentum vector \mathbf{p} :

$$\begin{aligned} \boldsymbol{\epsilon} \cdot \mathbf{p} &= \frac{\mathbf{p} \cdot \mathbf{r}}{r} - \frac{\mathbf{p} \cdot \mathbf{p} \times \mathbf{L}}{km} \\ &= \mathbf{p} \cdot \hat{\mathbf{r}} = p_r = \boldsymbol{\epsilon} p_x \end{aligned}$$

This says:

"Projection p_r of \mathbf{p} onto radial \mathbf{r} or \mathbf{r}' lines equals eccentricity $\boldsymbol{\epsilon}$ times projection p_x of \mathbf{p} onto orbit major axis: ($\hat{\mathbf{x}} = \hat{\boldsymbol{\epsilon}}$)"



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NOTE: Lengths of vectors \mathbf{p} and $-\mathbf{p}$ are not drawn to correctly show that momentum $\mathbf{p} = m\mathbf{v}$ grows as radial distance $r = |\mathbf{r}|$ falls. (To be shown on p. 85-90)

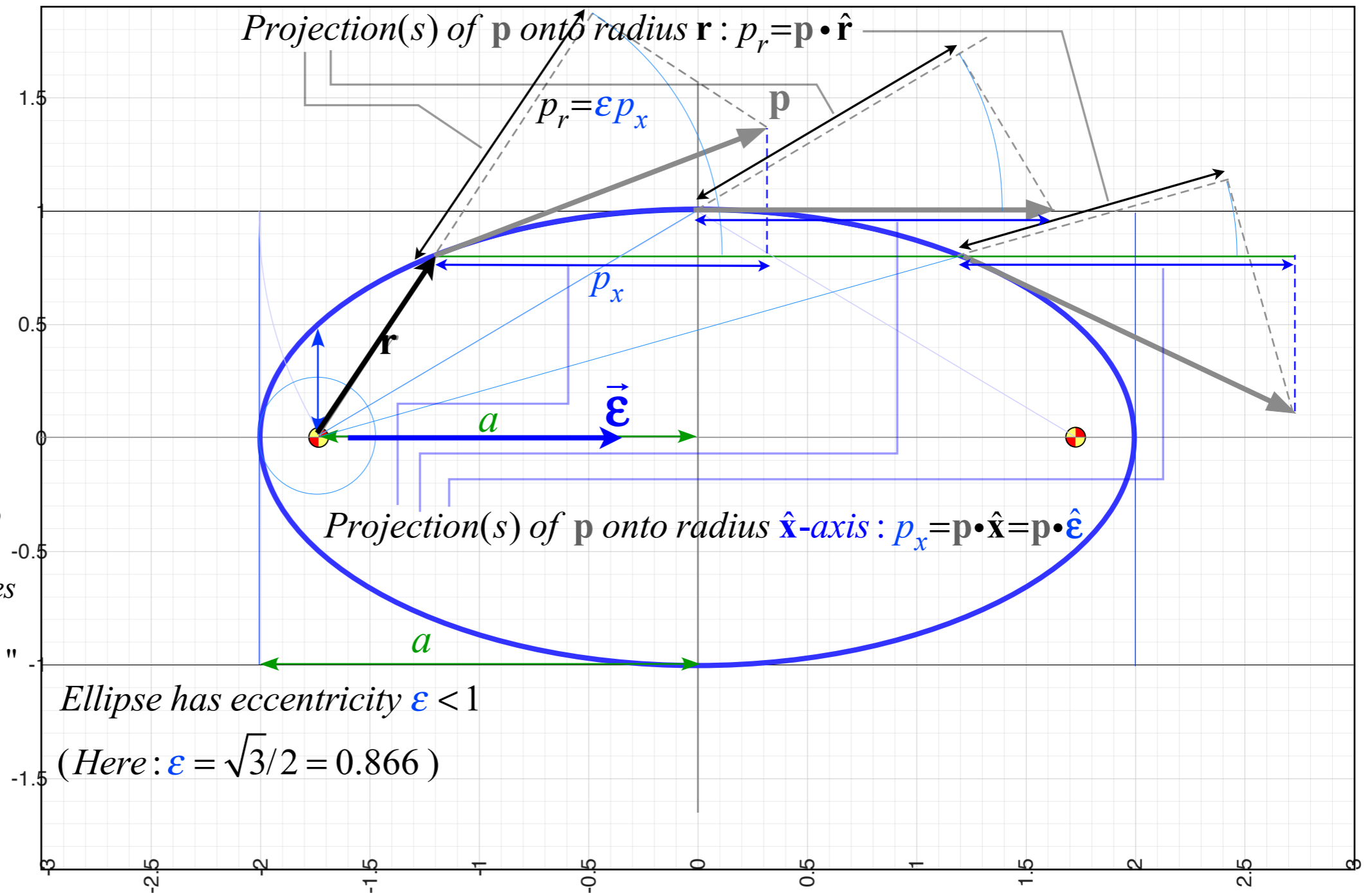
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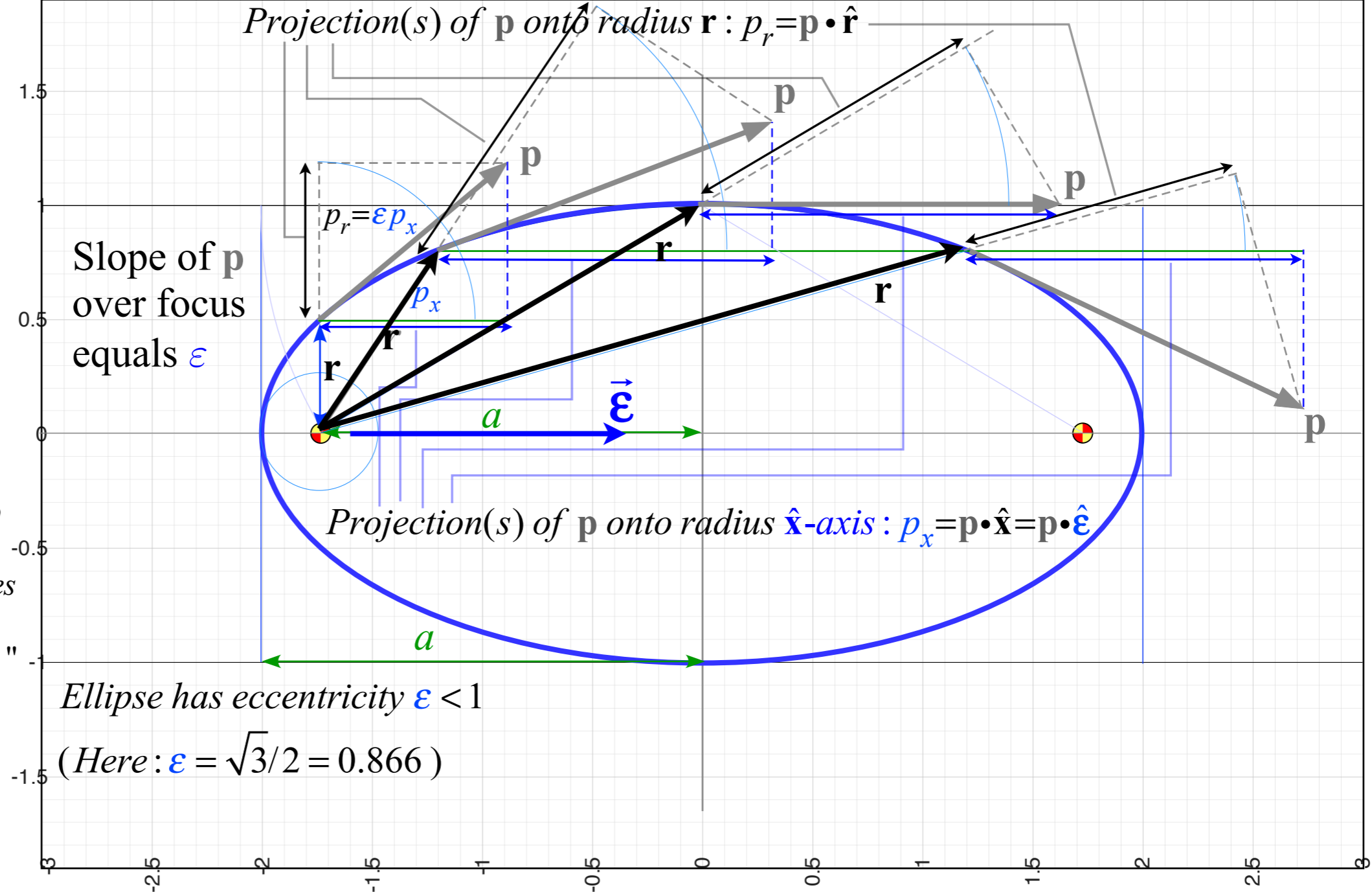
NOTE: Lengths of vectors \mathbf{p} and $-\mathbf{p}$ are not drawn to correctly show that momentum $\mathbf{p} = m\mathbf{v}$ grows as radial distance $r = |\mathbf{r}|$ falls. (To be shown on p. 85-90)

Dot product of ϵ with momentum vector \mathbf{p} :

$$\begin{aligned} \epsilon \cdot \mathbf{p} &= \frac{\mathbf{p} \cdot \mathbf{r}}{r} - \frac{\mathbf{p} \cdot \mathbf{p} \times \mathbf{L}}{km} \\ &= \mathbf{p} \cdot \hat{\mathbf{r}} = p_r = \epsilon p_x \end{aligned}$$

This says:

"Projection p_r of \mathbf{p} onto radial \mathbf{r} or \mathbf{r}' lines equals eccentricity ϵ times projection p_x of \mathbf{p} onto orbit major axis: ($\hat{\mathbf{x}} = \hat{\boldsymbol{\epsilon}}$)"



Ellipse has eccentricity $\epsilon < 1$

(Here: $\epsilon = \sqrt{3}/2 = 0.866$)

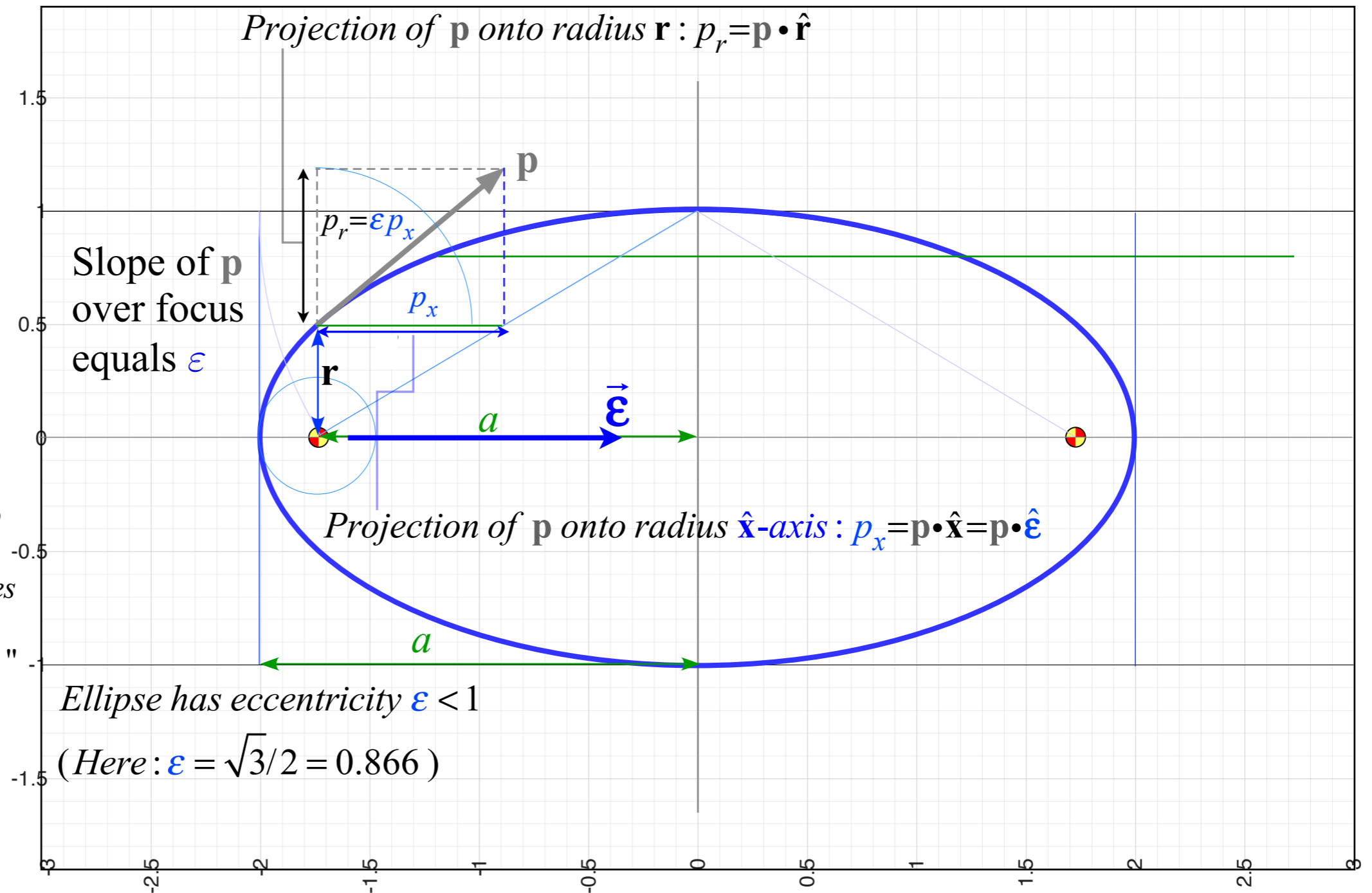
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$$\epsilon \cdot \mathbf{p} = \frac{\mathbf{p} \cdot \mathbf{r}}{r} - \frac{\mathbf{p} \cdot \mathbf{p} \times \mathbf{L}}{km} = \mathbf{p} \cdot \hat{\mathbf{r}} = p_r = \epsilon p_x$$

This says:

"Projection p_r of \mathbf{p} onto radial \mathbf{r} or \mathbf{r}' lines equals eccentricity ϵ times projection p_x of \mathbf{p} onto orbit major axis: ($\hat{\mathbf{x}} = \hat{\mathbf{e}}$)"



NOTE: Lengths of vectors \mathbf{p} and $-\mathbf{p}$ are not drawn to correctly show that momentum $\mathbf{p} = m\mathbf{v}$ grows as radial distance $r = |\mathbf{r}|$ falls. (To be shown on p. 85-90)

Dot product of $\boldsymbol{\varepsilon}$ with momentum vector \mathbf{p} :

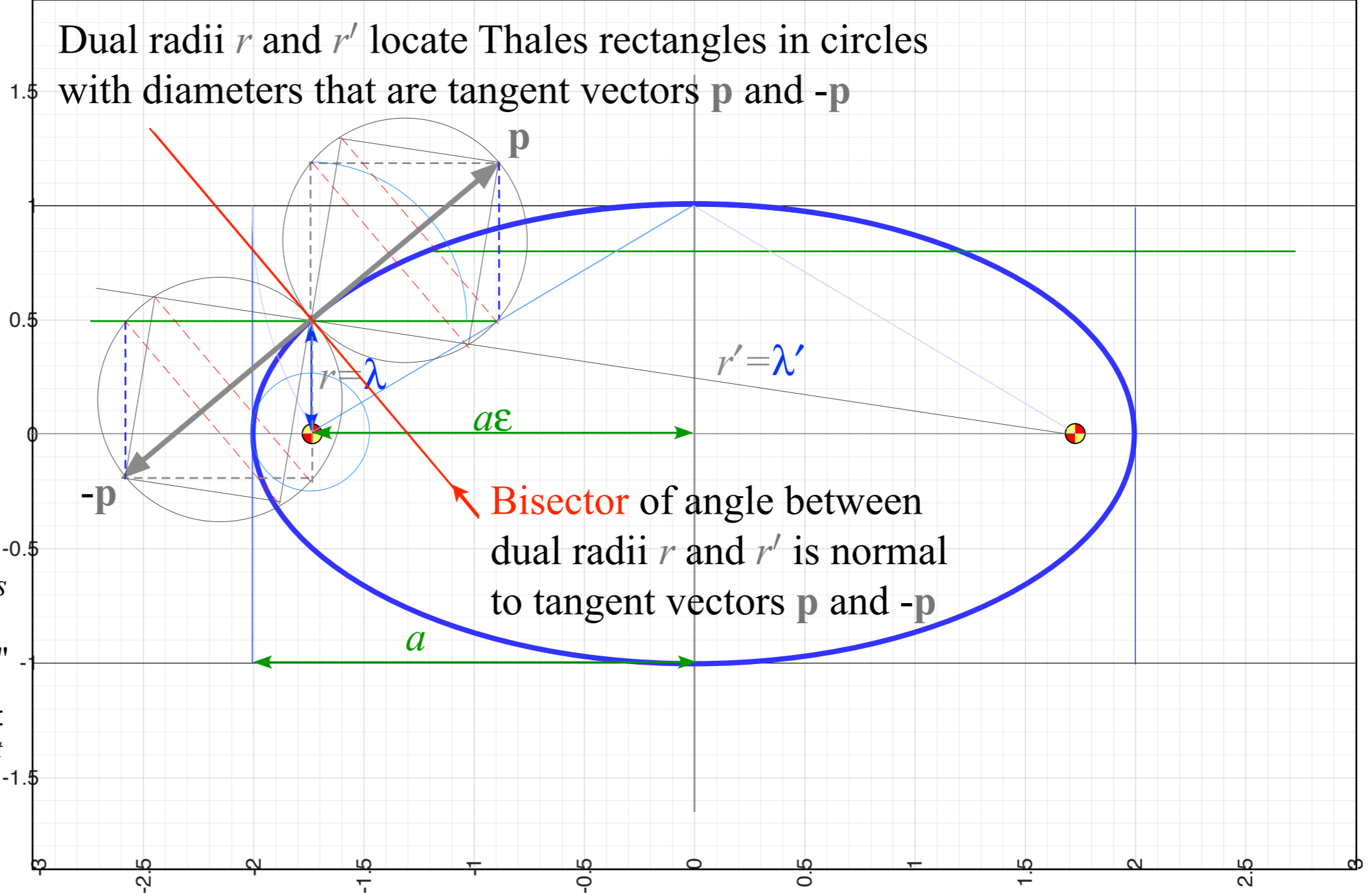
$$\boldsymbol{\varepsilon} \cdot \mathbf{p} = \frac{\mathbf{p} \cdot \mathbf{r}}{r} - \frac{\mathbf{p} \cdot \mathbf{p} \times \mathbf{L}}{km}$$

$$= \mathbf{p} \cdot \hat{\mathbf{r}} = p_r = \boldsymbol{\varepsilon} p_x$$

This says:

"Projection p_r of \mathbf{p} onto radial \mathbf{r} or \mathbf{r}' lines equals *eccentricity* $\boldsymbol{\varepsilon}$ times projection p_x of \mathbf{p} onto orbit major axis: $(\hat{\mathbf{x}} = \hat{\boldsymbol{\varepsilon}})$ "

Focal geometry demands:
"Momentum \mathbf{p} must bisect angle $\angle_{\mathbf{r}}^{\mathbf{r}'}$ between radial \mathbf{r} or \mathbf{r}' lines."



NOTE: Lengths of vectors \mathbf{p} and $-\mathbf{p}$ are not drawn to correctly show that momentum $\mathbf{p} = m\mathbf{v}$ grows as radial distance $r = |\mathbf{r}|$ falls. (To be shown on p. 85-90)

Dot product of $\boldsymbol{\varepsilon}$ with momentum vector \mathbf{p} :

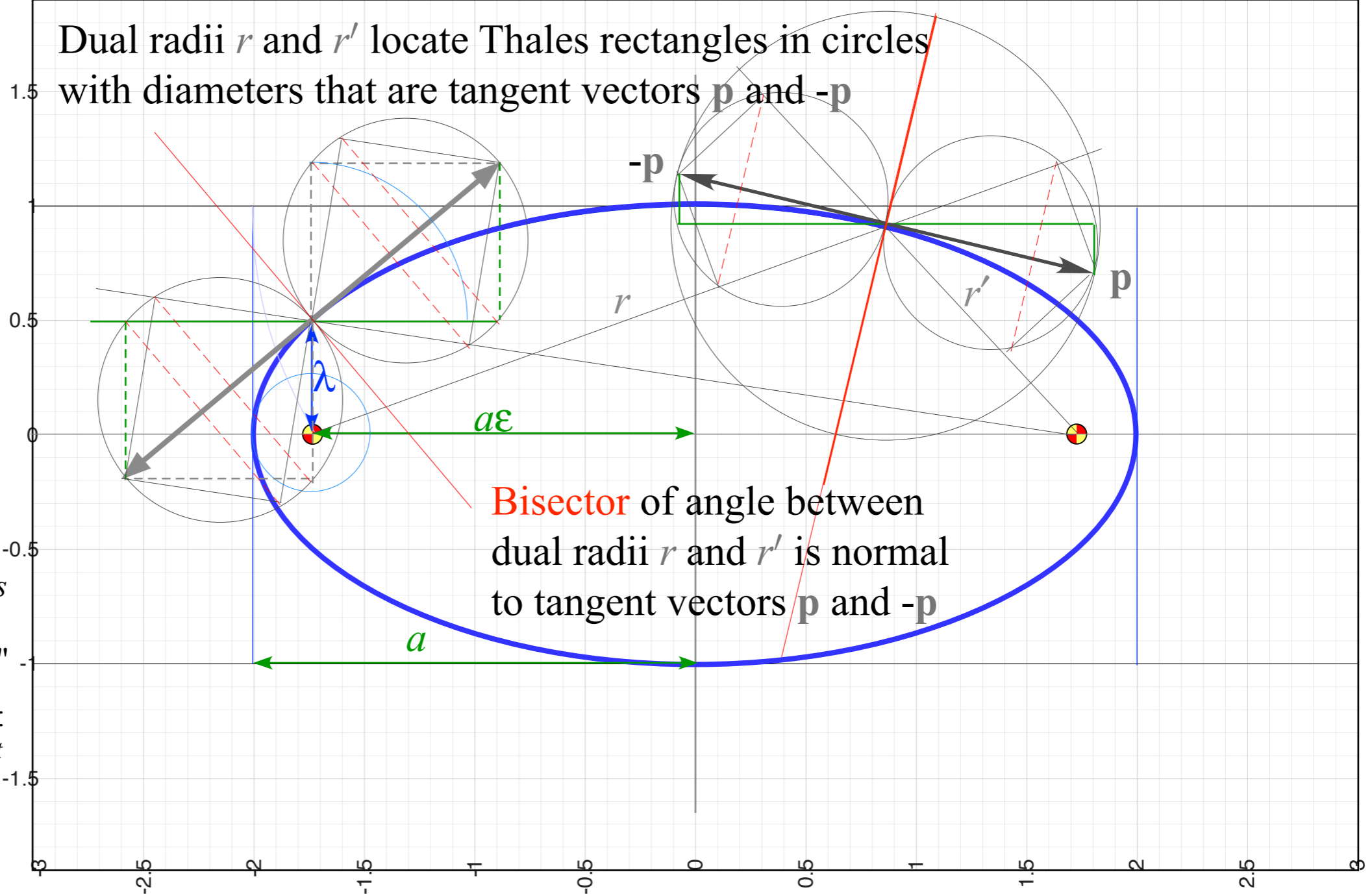
$$\boldsymbol{\varepsilon} \cdot \mathbf{p} = \frac{\mathbf{p} \cdot \mathbf{r}}{r} - \frac{\mathbf{p} \cdot \mathbf{p} \times \mathbf{L}}{km}$$

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This says:

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Focal geometry demands:
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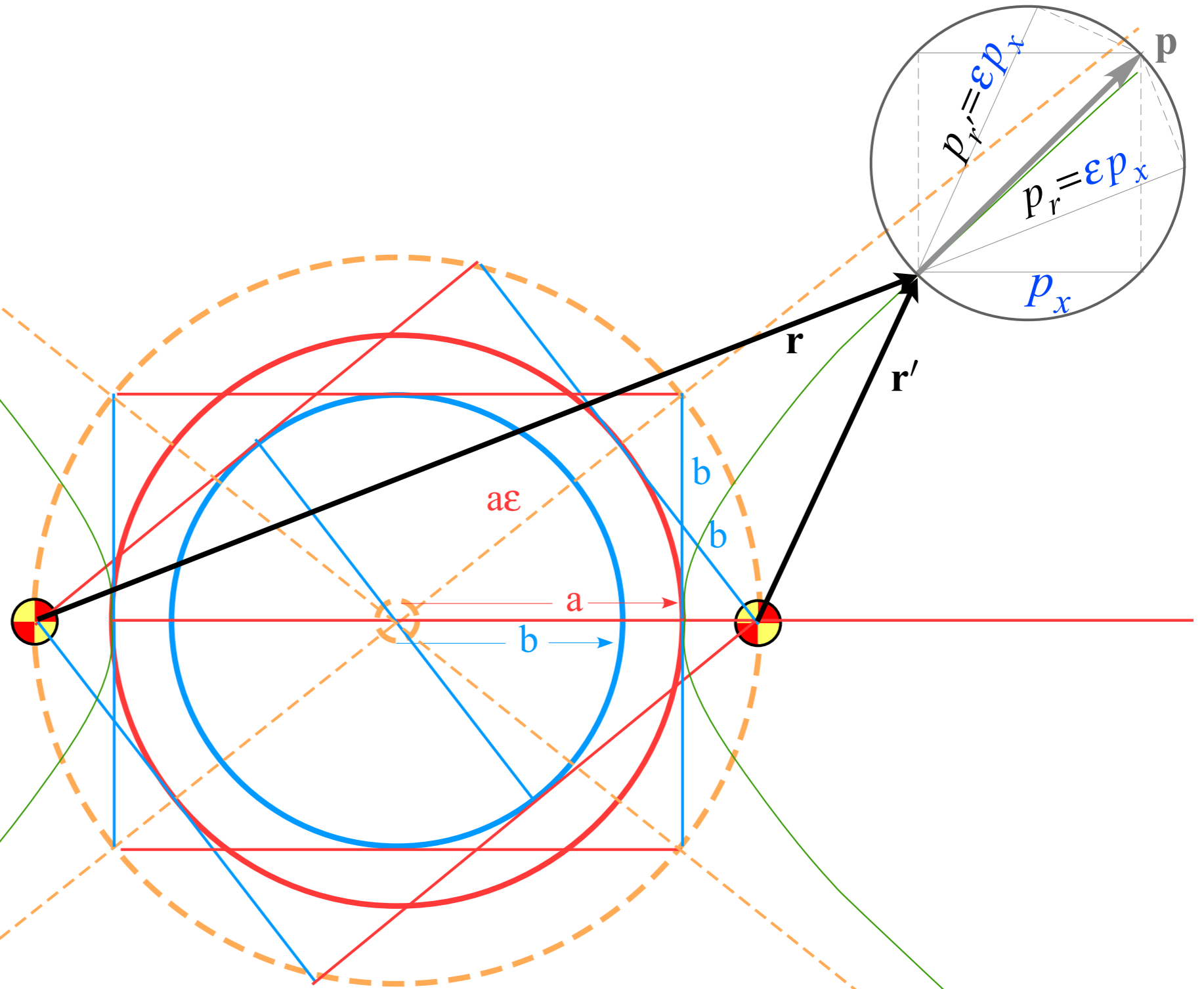
Dot product of $\boldsymbol{\epsilon}$ with momentum vector \mathbf{p} :

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This says:

"Projection p_r of \mathbf{p} onto radial \mathbf{r} or \mathbf{r}' lines equals eccentricity $\boldsymbol{\epsilon}$ times projection p_x of \mathbf{p} onto orbit major axis: $(\hat{\mathbf{x}} = \hat{\boldsymbol{\epsilon}})$ "

Focal geometry demands:
"Momentum \mathbf{p} must bisect angle $\angle_{\mathbf{r}'}^{\mathbf{r}}$ between radial \mathbf{r} or \mathbf{r}' lines."



Hyperbola has eccentricity $\boldsymbol{\epsilon} > 1$
(Here: $\boldsymbol{\epsilon} = 5/4 = 1.25$)

Rutherford scattering and hyperbolic orbit geometry

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Review and connection to usual orbital algebra (previous lecture)

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➔ *General geometric orbit construction using $\boldsymbol{\varepsilon}$ -vector and (γ, R) -parameters*

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Example of complete (\mathbf{r}, \mathbf{p}) -geometry of elliptical orbit

Connection formulas for (γ, R) -parameters with (a, b) and (ε, λ)

Next several pages give step-by-step constructions of ϵ -vector and Coulomb orbit and trajectory physics

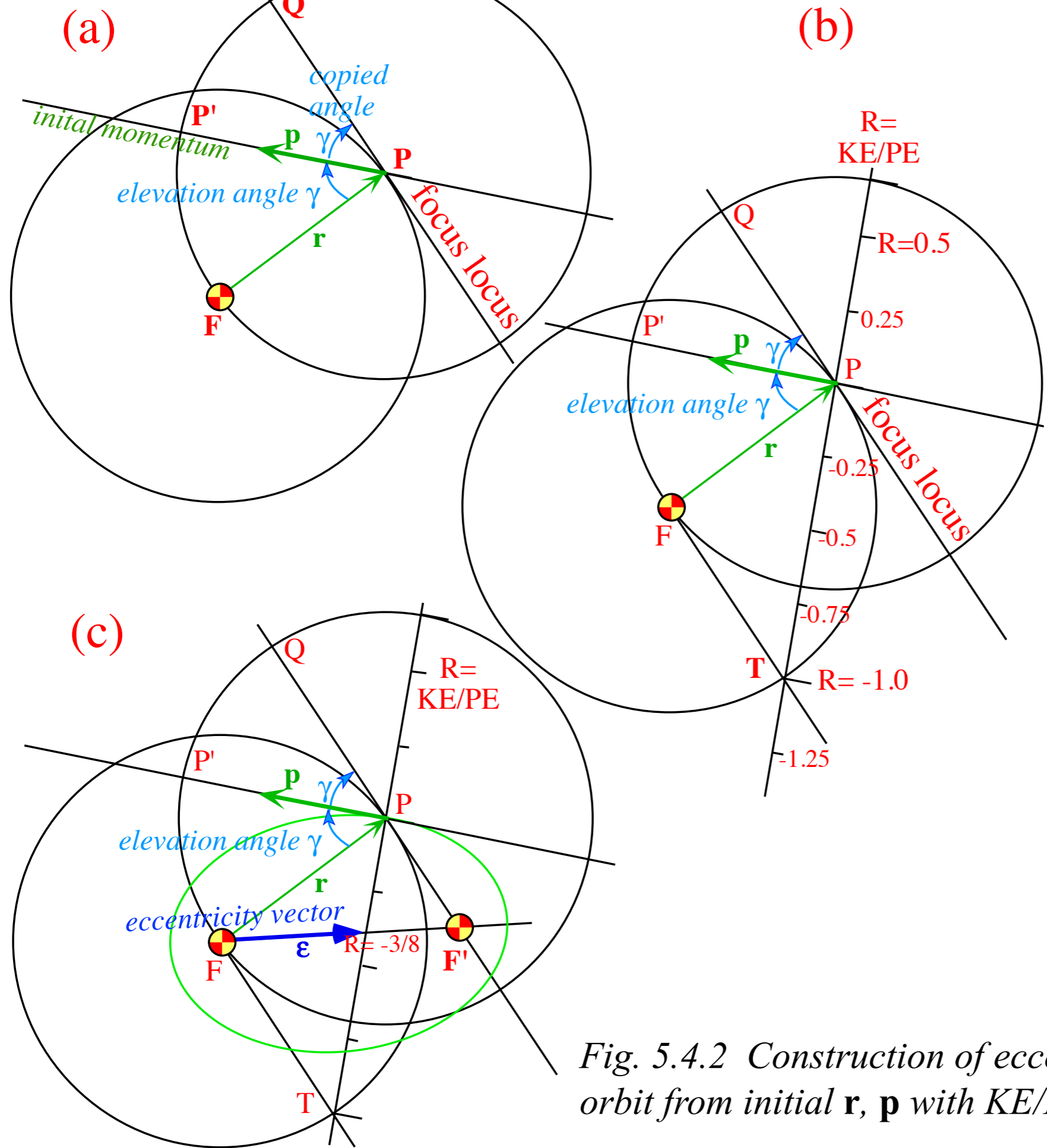
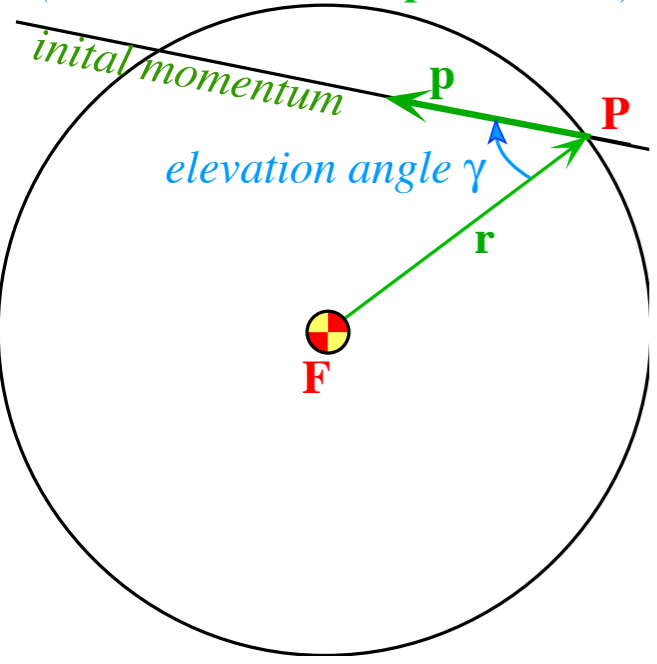


Fig. 5.4.2 Construction of eccentricity vector ϵ and orbit from initial \mathbf{r}, \mathbf{p} with $KE/PE = -3/8$.

General geometric orbit construction using ϵ -vector and (γ, R) -parameters

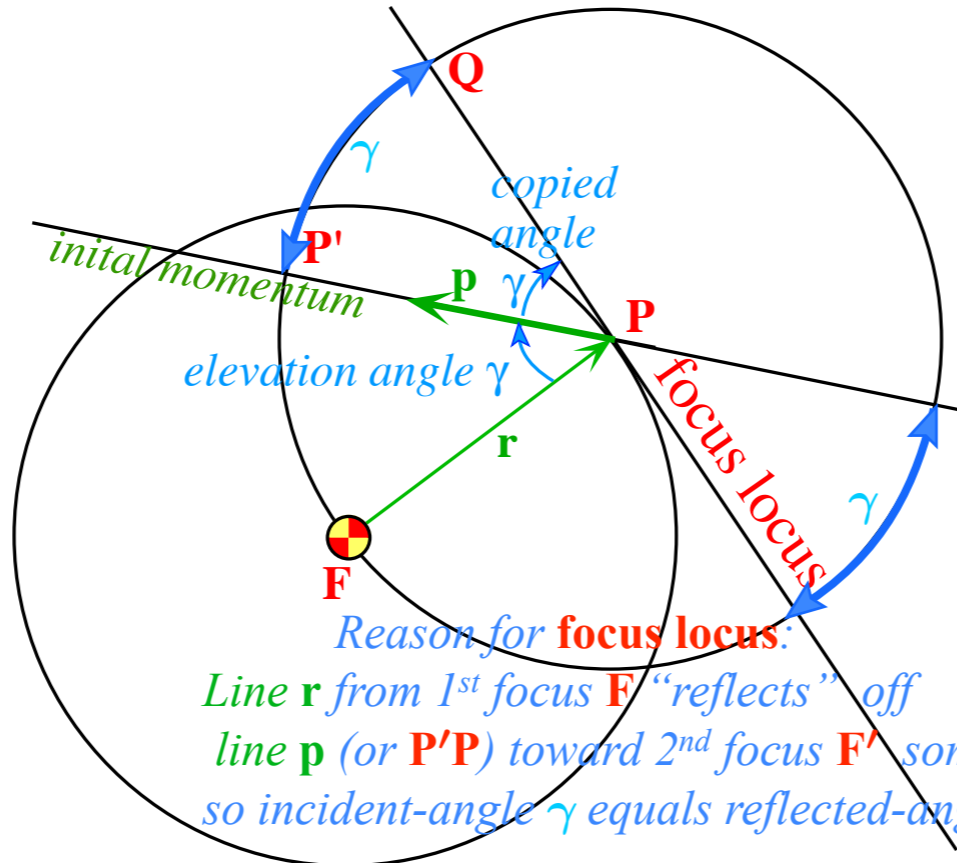
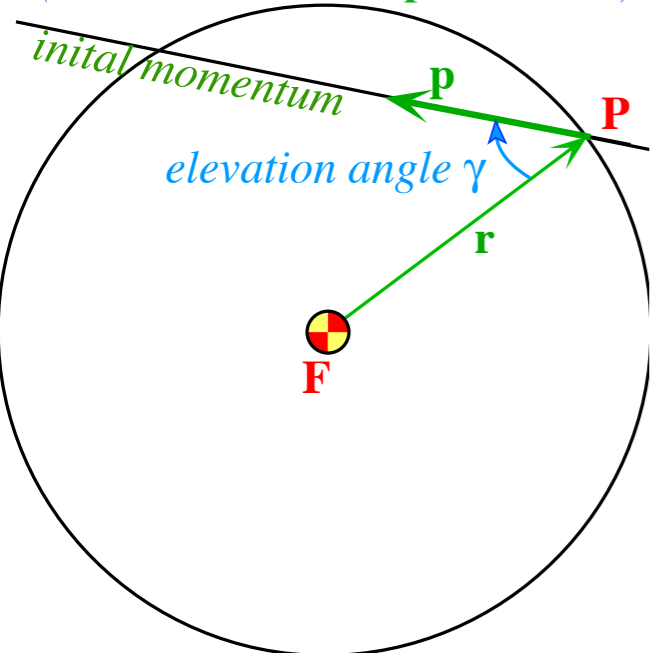
Pick launch point **P**
(radius vector **r**)
and elevation angle γ from radius
(momentum initial **p** direction)



General geometric orbit construction using ϵ -vector and (γ, R) -parameters

Pick launch point **P**
 (radius vector **r**)
 and elevation angle γ from radius
 (momentum initial **p** direction)

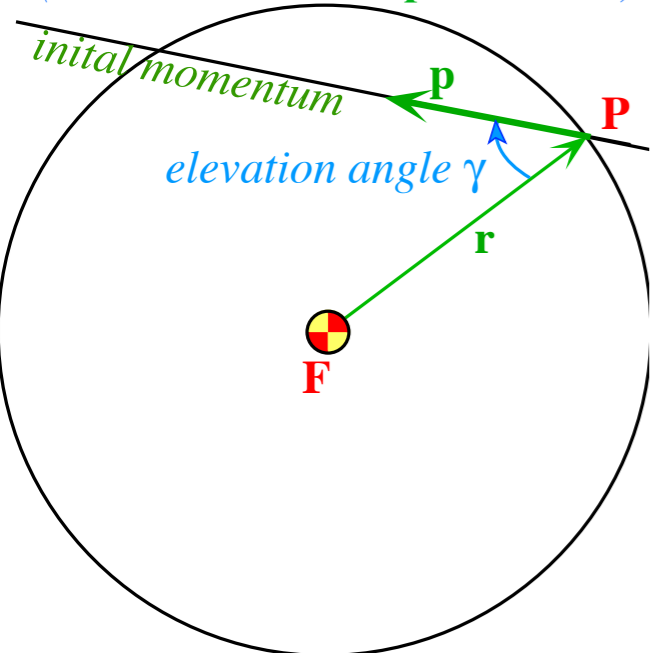
Copy F-center circle around launch point **P**
 Copy elevation angle γ ($\angle FPP'$) onto $\angle P'PQ$
 Extend resulting line **QPQ'** to make **focus locus**



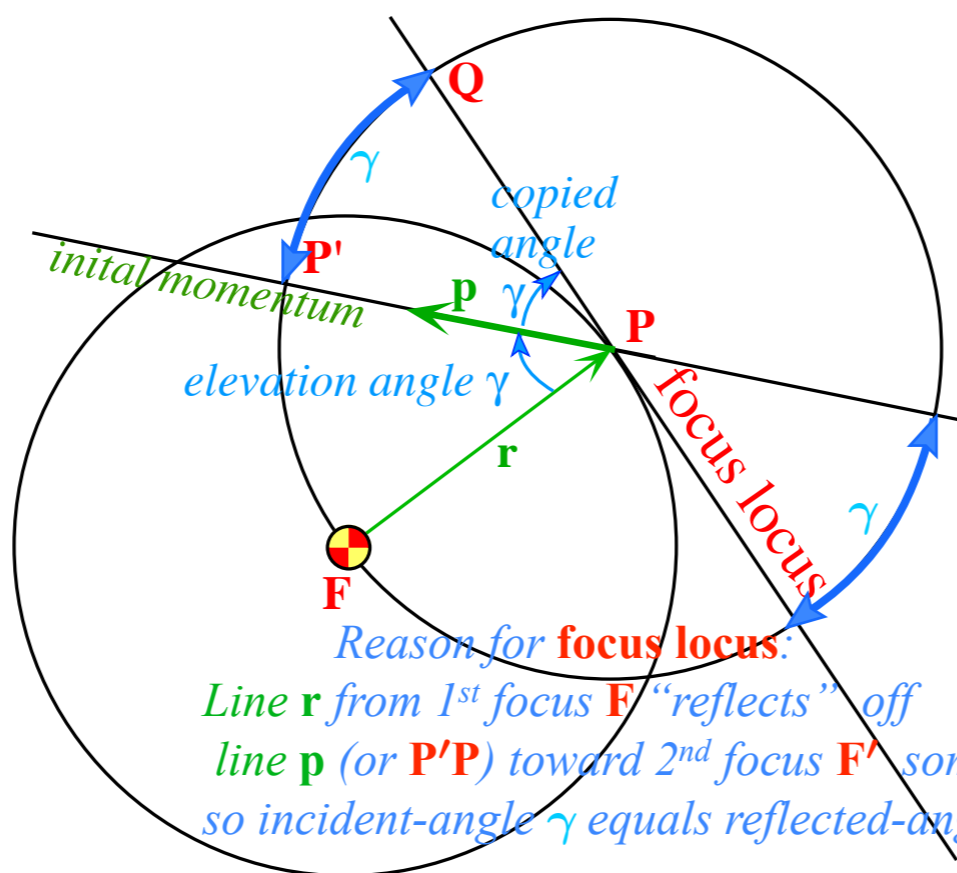
Reason for **focus locus**:
 Line **r** from 1st focus **F** "reflects" off
 line **p** (or **P'P**) toward 2nd focus **F'** somewhere
 so incident-angle γ equals reflected-angle γ

General geometric orbit construction using ϵ -vector and (γ, R) -parameters

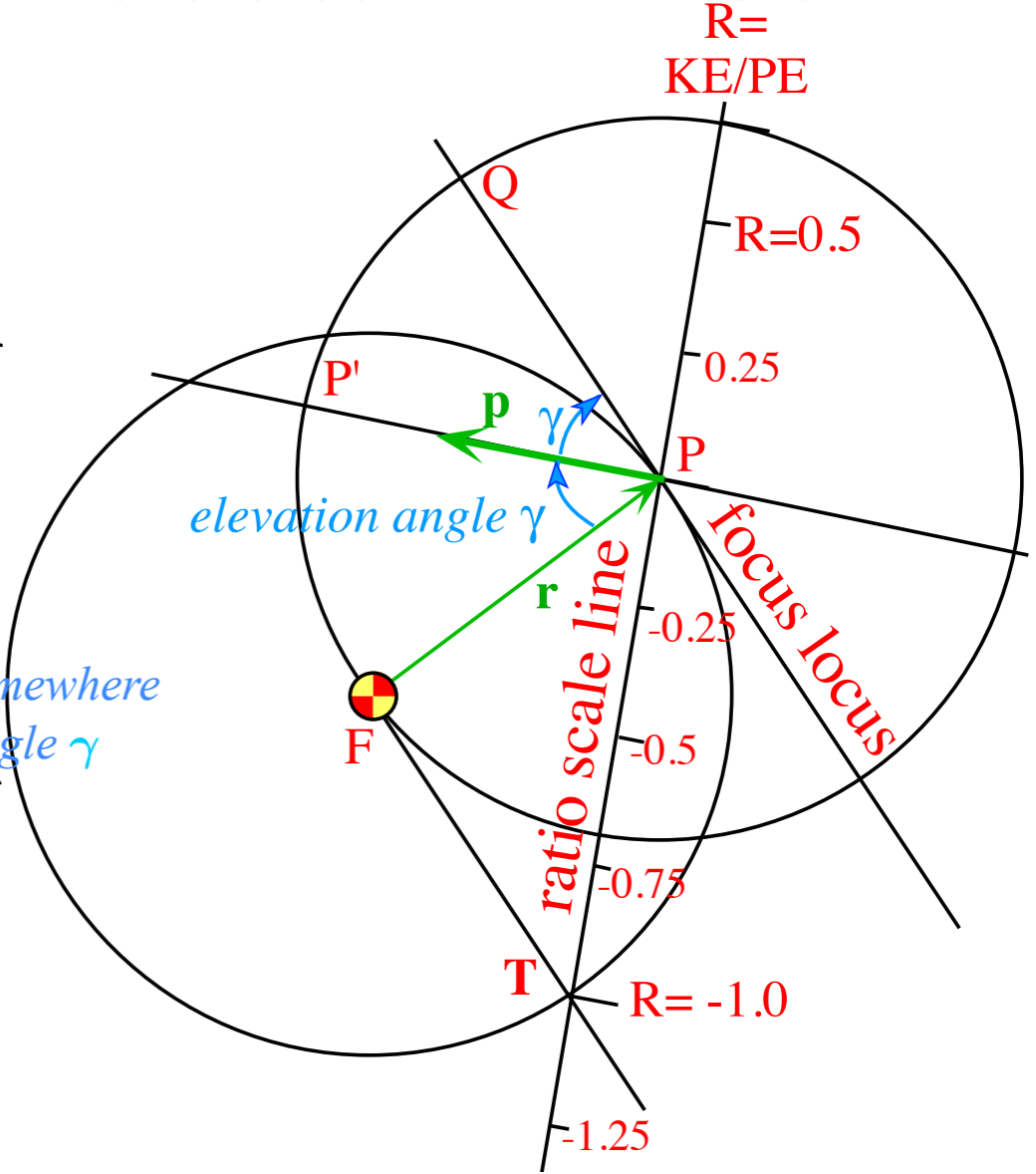
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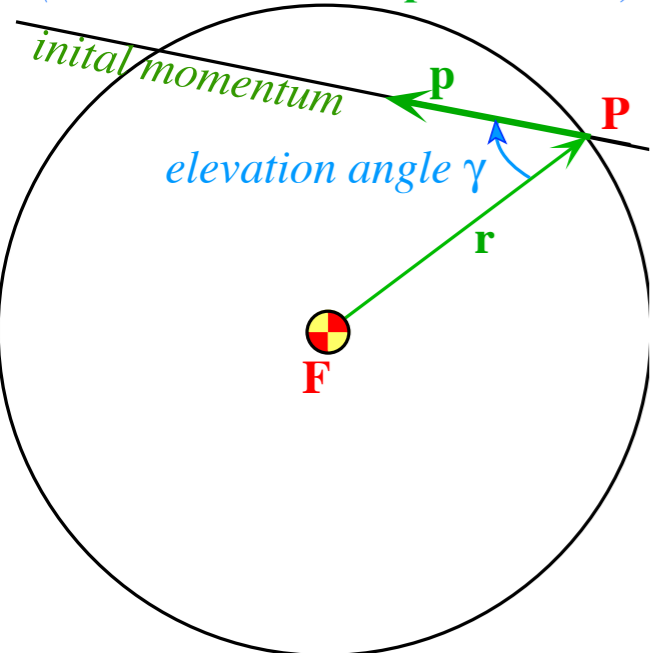


Copy double angle 2γ ($\angle FPQ$) onto $\angle PFT$
 Extend $\angle PFT$ chord **PT** to make **R-ratio scale line**
 Label chord **PT** with $R=0$ at **P** and $R=-1.0$ at **T**.
 Mark **R-line** fractions $R=0, +1/4, +1/2, \dots$ above **P** and
 $R=0, -1/8, -1/4, -1/2, \dots, -3/4$ below **P** and $-5/4, -3/2, \dots$ below **T**.

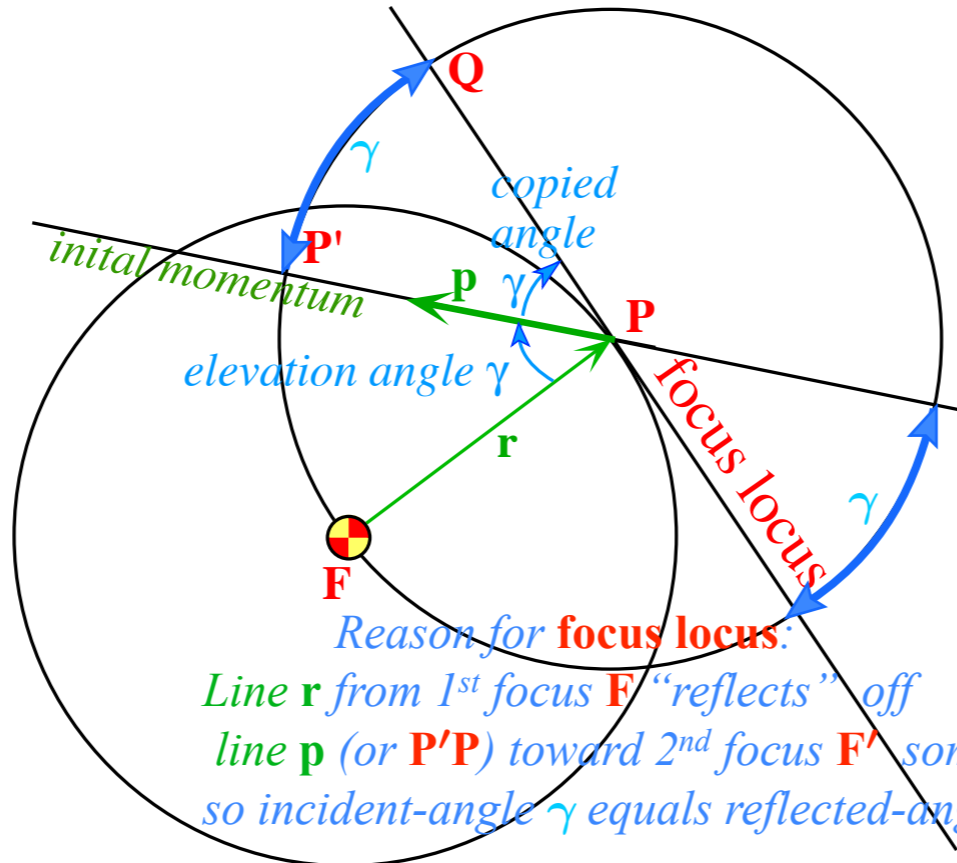


General geometric orbit construction using ϵ -vector and (γ, R) -parameters

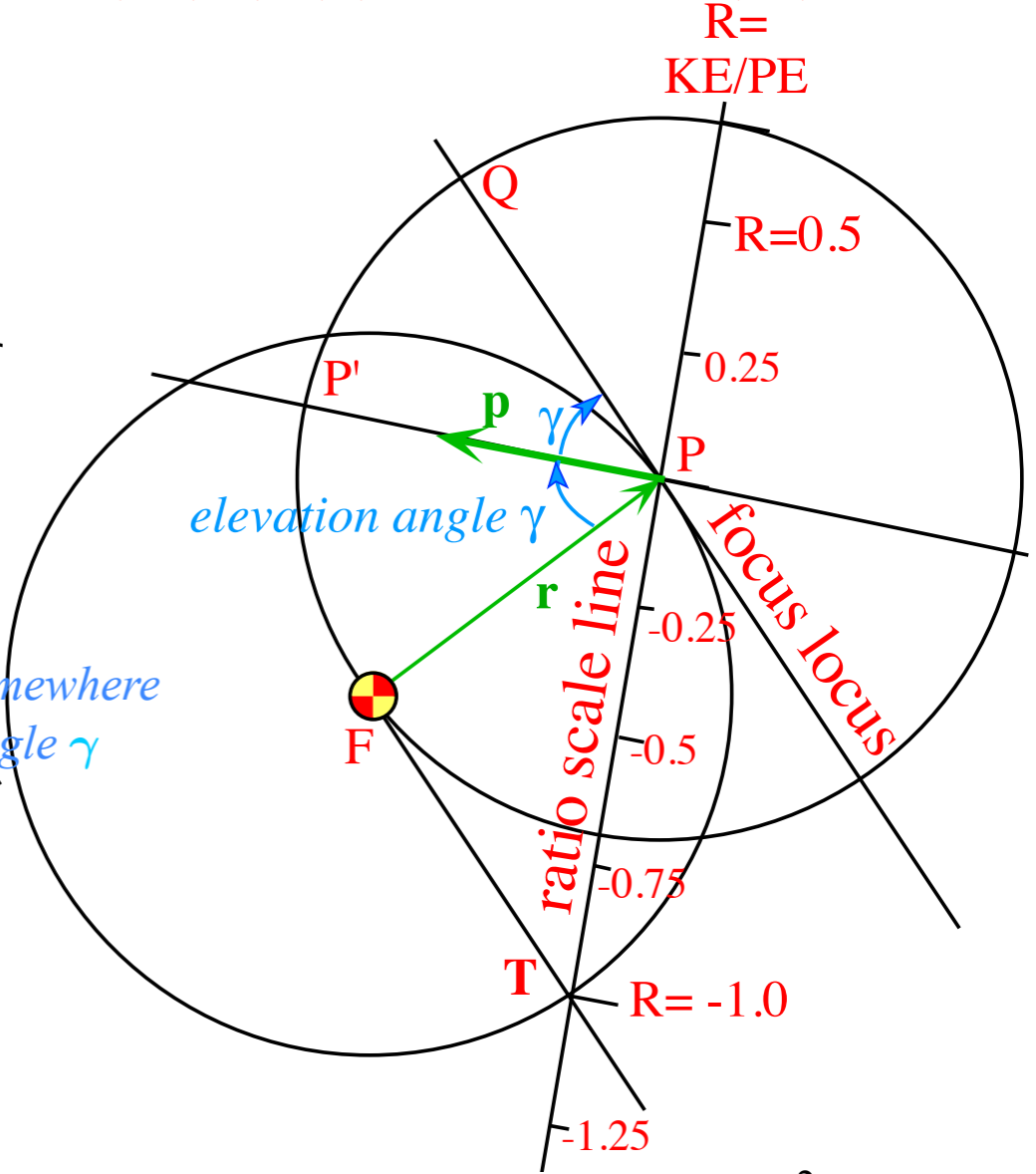
Pick launch point **P**
 (radius vector **r**)
 and elevation angle γ from radius
 (momentum initial **p** direction)



Copy F-center circle around launch point **P**
 Copy elevation angle γ ($\angle FPP'$) onto $\angle P'PQ$
 Extend resulting line **QPQ'** to make **focus locus**



Copy double angle 2γ ($\angle FPQ$) onto $\angle PFT$
 Extend $\angle PFT$ chord **PT** to make **R-ratio scale line**
 Label chord **PT** with $R=0$ at **P** and $R=-1.0$ at **T**.
 Mark **R-line** fractions $R=0, +1/4, +1/2, \dots$ above **P** and
 $R=0, -1/8, -1/4, -1/2, \dots, -3/4$ below **P** and $-5/4, -3/2, \dots$ below **T**.

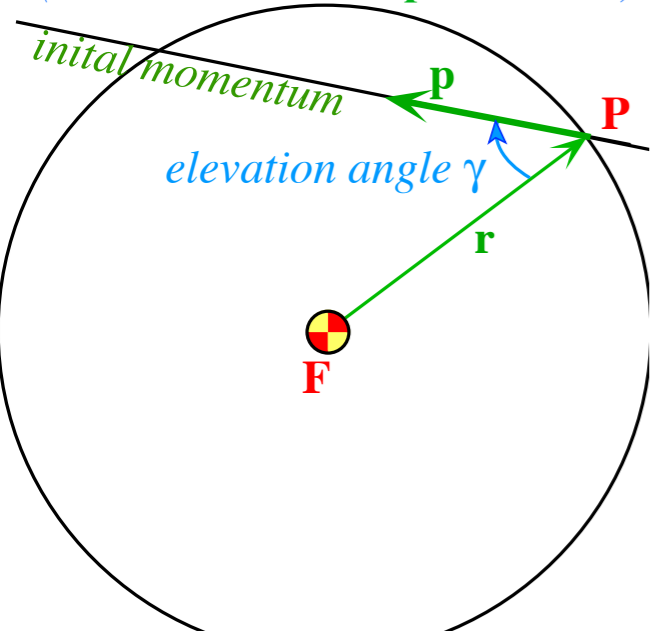


$$R = \frac{\text{Initial KE}}{\text{Initial PE}} = \frac{mv^2(0)/2}{-k/r(0)}$$

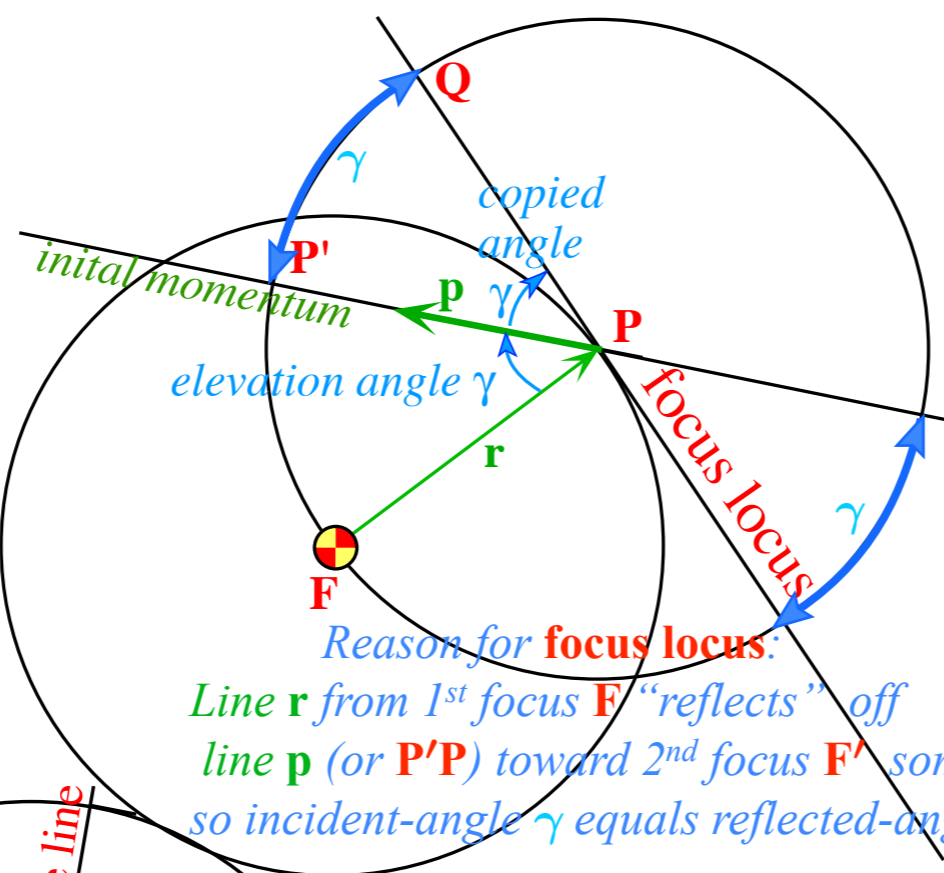
$$= \pm \left(\frac{\text{Initial velocity}}{\text{Escape velocity}} \right)^2 = \pm \frac{v^2(0)}{v^2(\infty)}$$

General geometric orbit construction using ϵ -vector and (γ, R) -parameters

Pick launch point **P**
(radius vector **r**)
and elevation angle γ from radius
(momentum initial **p** direction)

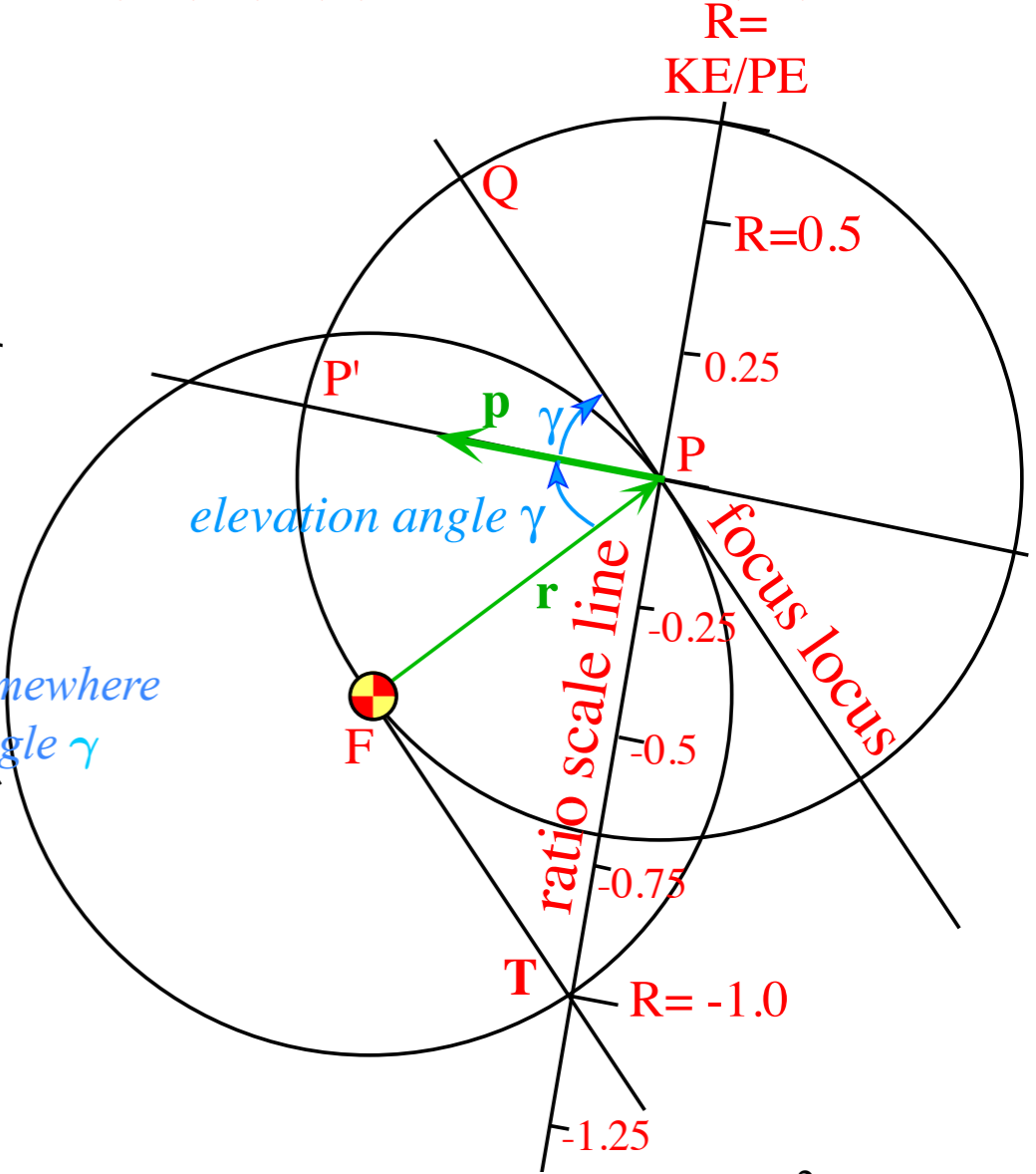


Copy F-center circle around launch point **P**
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Extend resulting line **QPQ'** to make **focus locus**

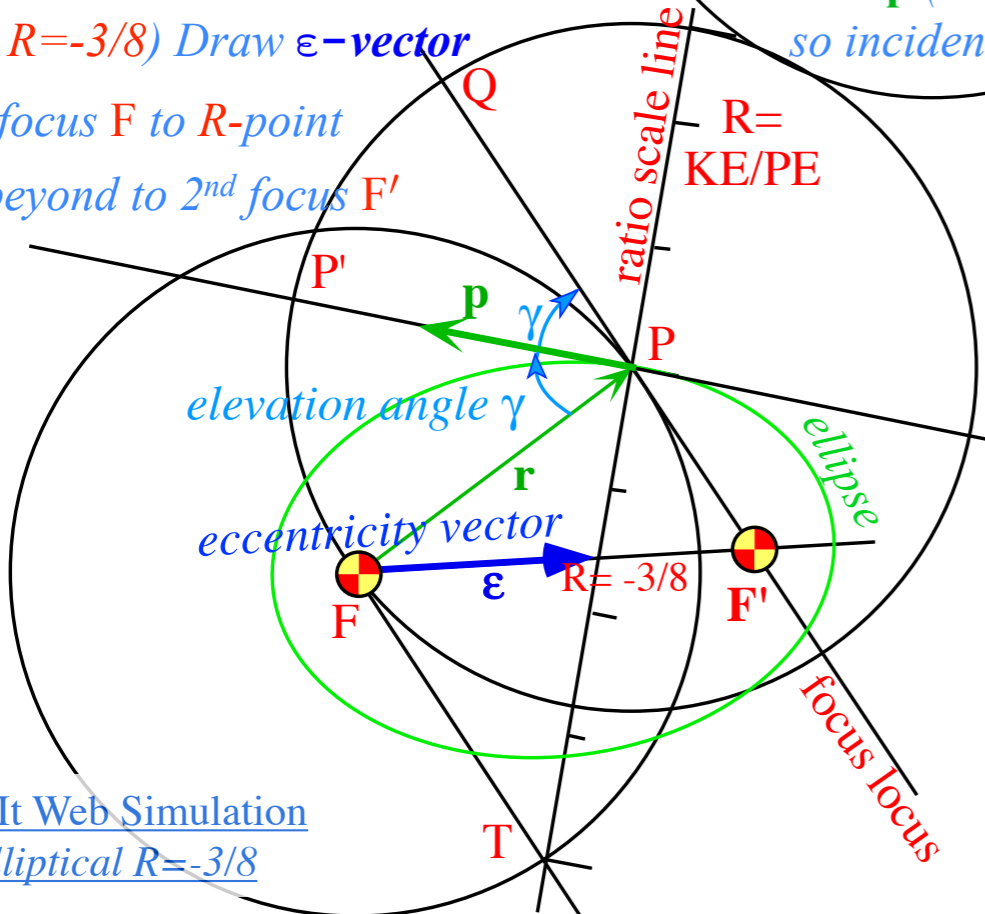


Reason for **focus locus**:
Line **r** from 1st focus **F** "reflects" off
line **p** (or **P'P**) toward 2nd focus **F'** somewhere
so incident-angle γ equals reflected-angle γ

Copy double angle 2γ ($\angle FPQ$) onto $\angle PFT$
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Label chord **PT** with $R=0$ at **P** and $R=-1.0$ at **T**.
Mark **R-line** fractions $R=0, +1/4, +1/2, \dots$ above **P** and
 $R=0, -1/8, -1/4, -1/2, \dots, -3/4$ below **P** and $-5/4, -3/2, \dots$ below **T**.



Pick initial $R=KE/PE$ value
(here $R=-3/8$) Draw ϵ -vector
from focus **F** to **R-point**
and beyond to 2nd focus **F'**



$$R = \frac{\text{Initial KE}}{\text{Initial PE}} = \frac{m v^2(0) / 2}{-k / r(0)}$$

$$= \pm \left(\frac{\text{Initial velocity}}{\text{Escape velocity}} \right)^2 = \pm \frac{v^2(0)}{v^2(\infty)}$$

focus **F** and 2nd focus **F'** allow final
construction of **orbital trajectory**.
Here it is an $R=-3/8$ **ellipse**.

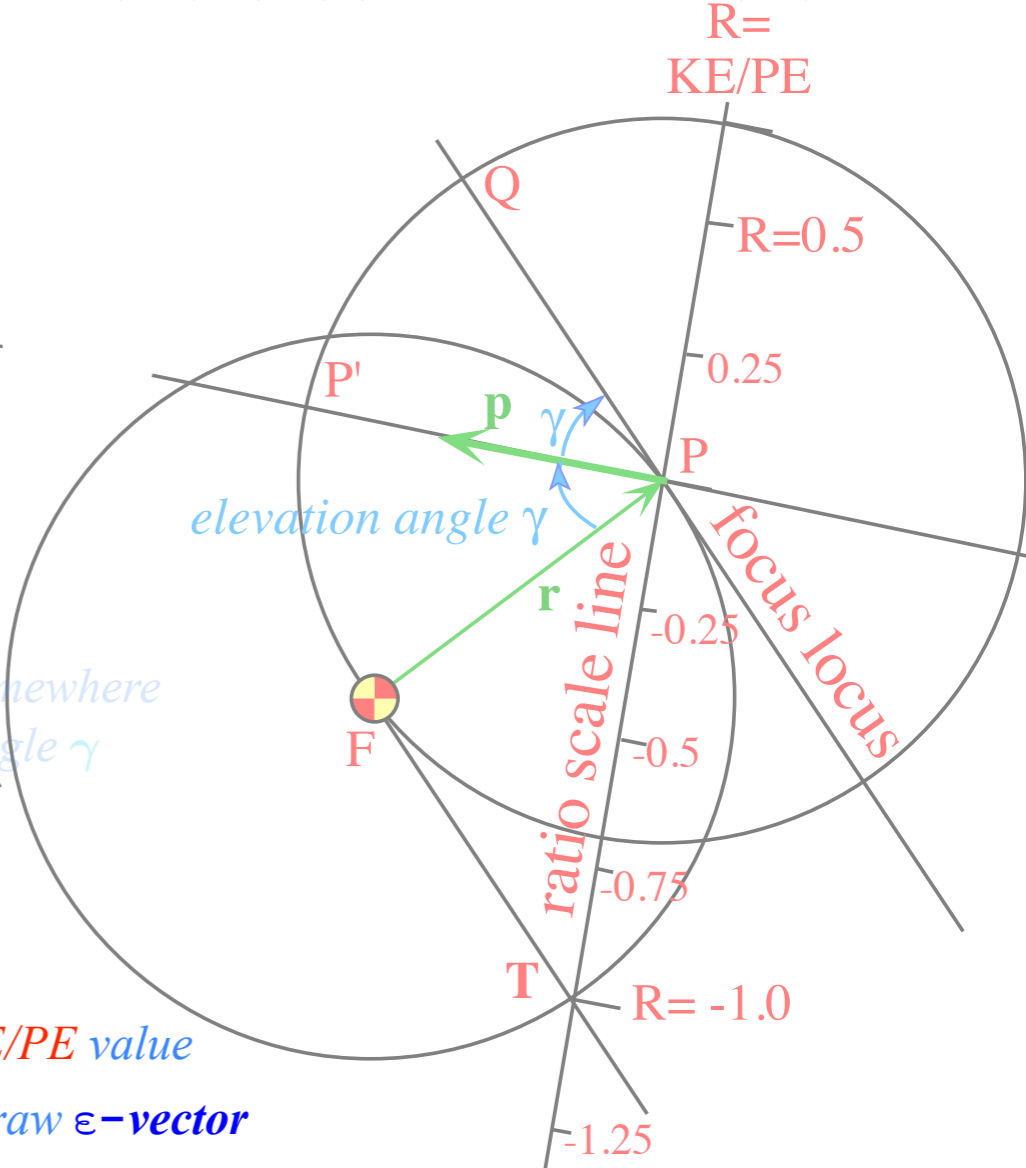
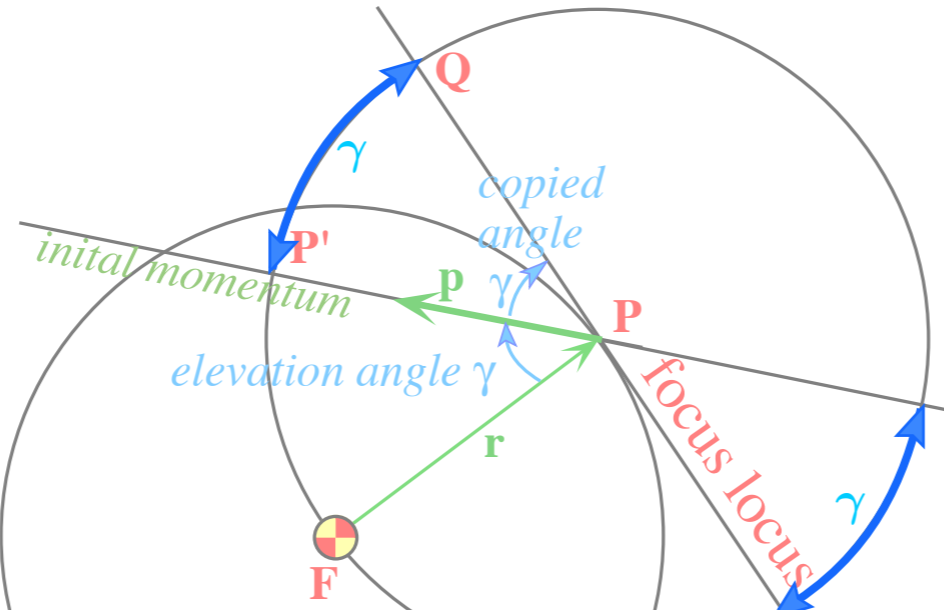
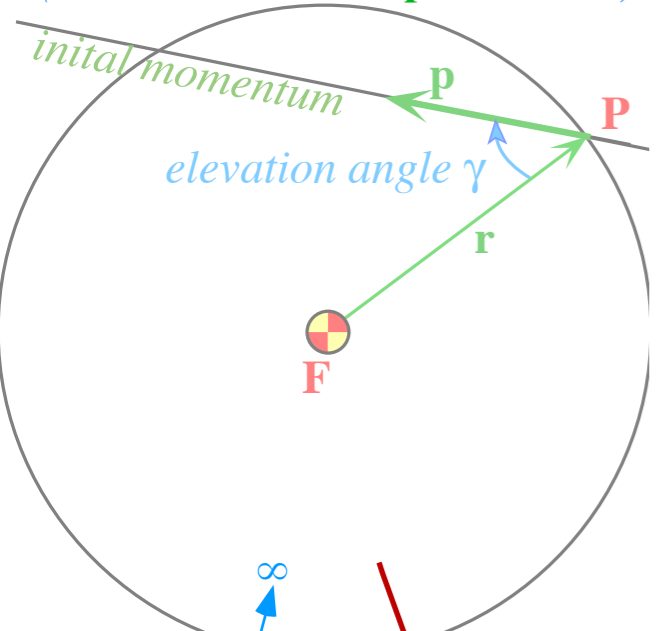
(Detailed Analytic geometry of ϵ -vector follows.)

General geometric orbit construction using ϵ -vector and (γ, R) -parameters

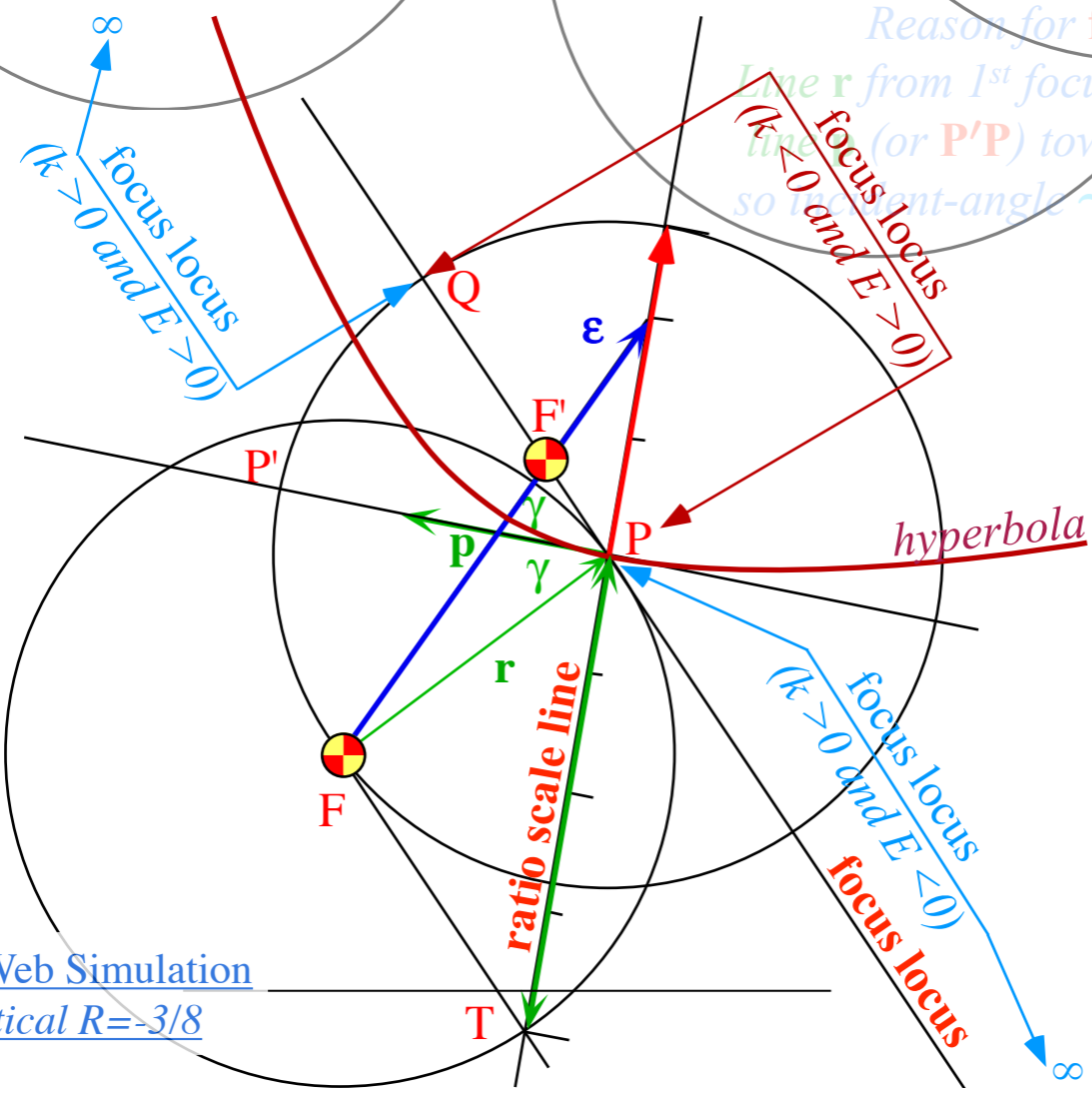
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Copy elevation angle γ ($\angle FPP'$) onto $\angle P'PQ$
Extend resulting line **QPQ'** to make **focus locus**

Copy double angle 2γ ($\angle FPQ$) onto $\angle PFT$
Extend $\angle PFT$ chord **PT** to make **R-ratio scale line**
Label chord **PT** with $R=0$ at **P** and $R=-1.0$ at **T**.
Mark **R-line** fractions $R=0, +1/4, +1/2, \dots$ above **P** and
 $R=0, -1/8, -1/4, -1/2, \dots, -3/4$ below **P** and $-5/4, -3/2, \dots$ below **T**.



Reason for **focus locus**:
Line **r** from 1st focus **F** "reflects" off
line **PP'** toward 2nd focus **F'** somewhere
so incident-angle γ equals reflected-angle γ



Pick initial $R=KE/PE$ value
(here $R=+1/2$) Draw ϵ -vector
from focus **F** to **R**-point
(Here it intersects 2nd focus **F'**)

focus **F** and 2nd focus **F'** allow final
construction of orbital trajectory.
Here it is an $R=+1/2$ hyperbola.

(Detailed Analytic geometry of ϵ -vector follows.)

$$R = \frac{\text{Initial KE}}{\text{Initial PE}} = \frac{m v^2(0) / 2}{-k / r(0)}$$

$$= \pm \left(\frac{\text{Initial velocity}}{\text{Escape velocity}} \right)^2 = \pm \frac{v^2(0)}{v^2(\infty)}$$

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Eccentricity vector $\boldsymbol{\varepsilon}$ and (ε, λ) -geometry of orbital mechanics

Projection $\boldsymbol{\varepsilon} \cdot \mathbf{r}$ geometry of $\boldsymbol{\varepsilon}$ -vector and orbital radius \mathbf{r}

Review and connection to usual orbital algebra (previous lecture)

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➔ *Derivation of $\boldsymbol{\varepsilon}$ -construction by analytic geometry*

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Connection formulas for (γ, R) -parameters with (a, b) and (ε, λ)

Derivation of ϵ -construction by analytic geometry

$$\epsilon = \hat{r} - \frac{\mathbf{p} \times \mathbf{L}}{km} = \hat{r} - \frac{(mv_0)(mv_0 r_0) \sin \gamma}{km} \hat{\mathbf{L}}_{\mathbf{p} \times}$$

where: $\mathbf{L}_{\mathbf{p} \times} \equiv \mathbf{p} \times \mathbf{L}$

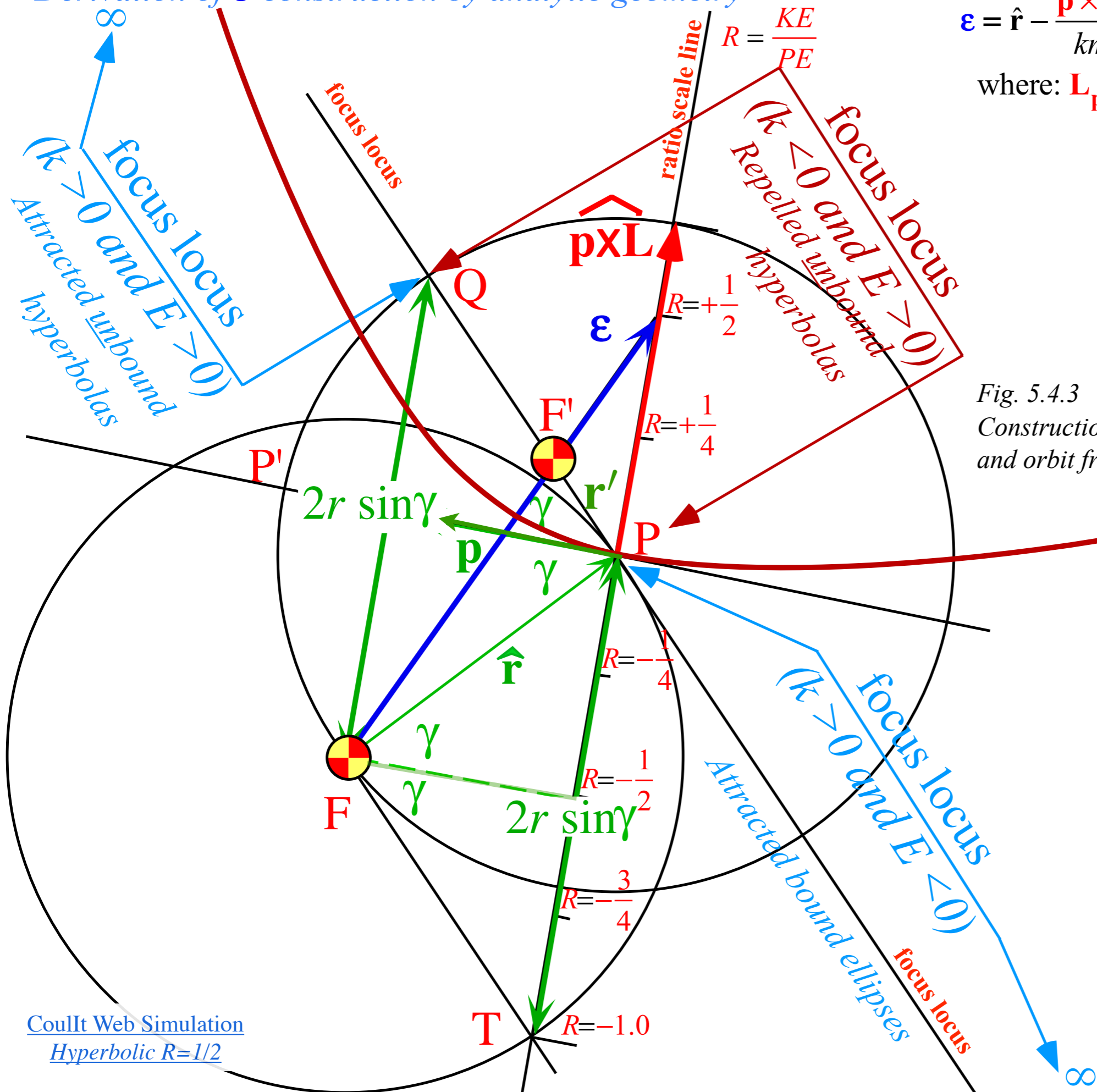
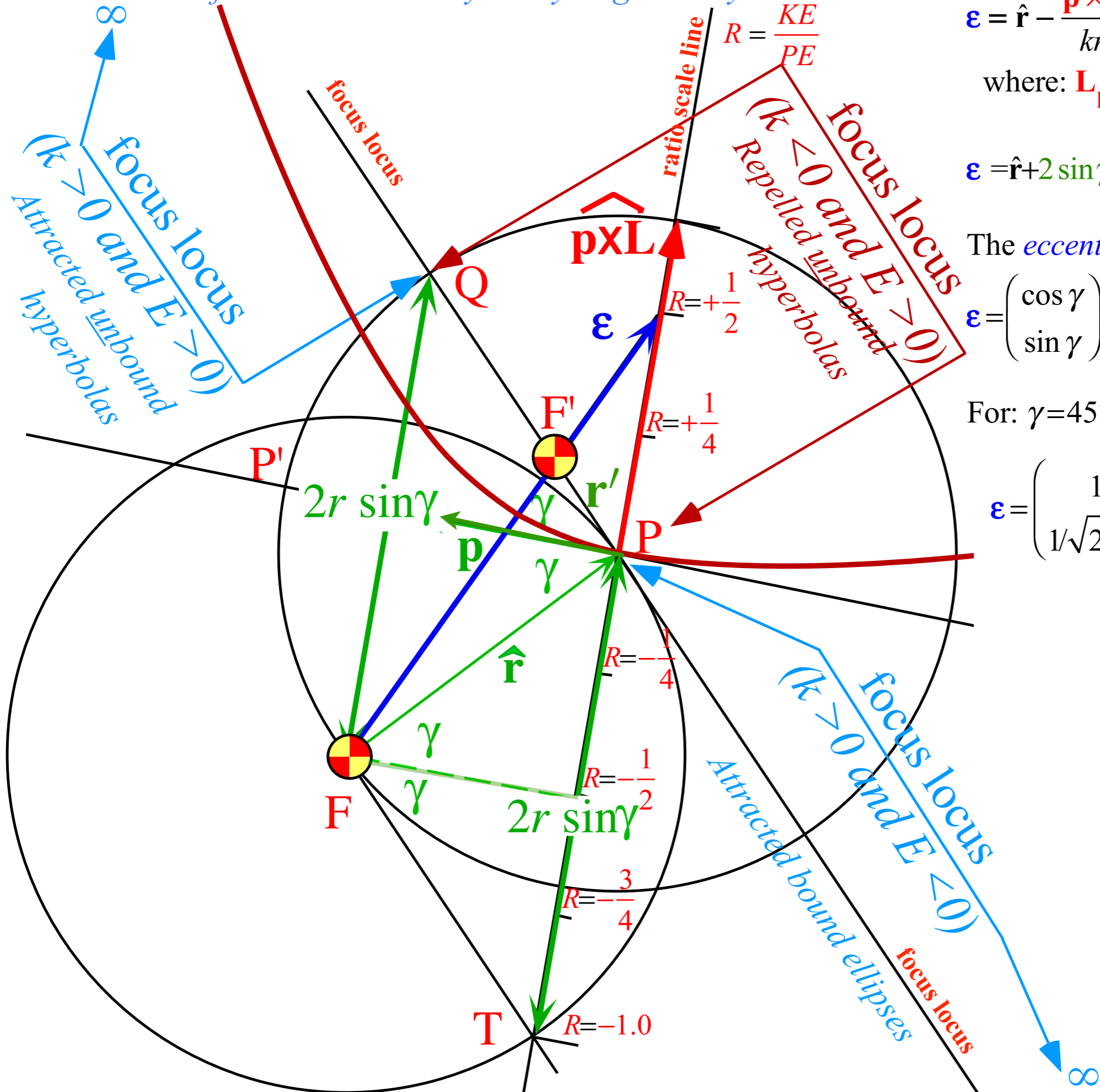


Fig. 5.4.3
Construction of eccentricity vector ϵ
and orbit from initial \mathbf{r}, \mathbf{p} with $KE/PE = +1/2$.

$$R = \frac{\text{Initial KE}}{\text{Initial PE}} = \frac{mv^2(0)/2}{-k/r(0)}$$

$$= \pm \left(\frac{\text{Initial velocity}}{\text{Escape velocity}} \right)^2 = \pm \frac{v^2(0)}{v^2(\infty)}$$

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where: $\mathbf{L}_{\mathbf{p} \times} \equiv \mathbf{p} \times \mathbf{L}$

$$\epsilon = \hat{r} + 2 \sin \gamma \frac{mv_0^2/2}{-k/r_0} \hat{\mathbf{L}}_{\mathbf{p} \times} = \hat{r} + 2 \sin \gamma \frac{KE}{PE} \hat{\mathbf{L}}_{\mathbf{p} \times}$$

The *eccentricity* vector is:

$$\epsilon = \begin{pmatrix} \cos \gamma \\ \sin \gamma \end{pmatrix} + 2 \sin \gamma \begin{pmatrix} 0 \\ 1 \end{pmatrix} R = \begin{pmatrix} \cos \gamma \\ (2R+1) \sin \gamma \end{pmatrix}$$

For: $\gamma = 45^\circ$ and: $R = +\frac{1}{2}$

$$\epsilon = \begin{pmatrix} 1/\sqrt{2} \\ 1/\sqrt{2}(2R+1) \end{pmatrix} = \begin{pmatrix} 1/\sqrt{2} \\ 2/\sqrt{2} \end{pmatrix},$$

$$R = \frac{\text{Initial KE}}{\text{Initial PE}} = \frac{mv^2(0)/2}{-k/r(0)} = \pm \left(\frac{\text{Initial velocity}}{\text{Escape velocity}} \right)^2 = \pm \frac{v^2(0)}{v^2(\infty)}$$

Initial position $x(0) = 0.465648$

Initial position $y(0) = 1.156488$

Initial momentum $p_x(0) = 0.591603$

Initial momentum $p_y(0) = 0.435114$

Terminal time $t(\text{off}) = 20$

Maximum step size $dt = 0.01$

Charge of Nucleus 1 = -1

x-Position of Nucleus 1 = 0

y-Position of Nucleus 1 = 0

Charge of Nucleus 2 = 0

Coulomb (k_{12}) = -1

Core thickness $r = 0.000001$

x-Stark field $E_x = 0$

y-Stark field $E_y = 0$

Zeeman field $B_z = 0$

Diamagnetic strength $k = 0$

Plank constant $\hbar = 2$

Color quantization hues = 64

Color quantization bands = 2

Fractional Error (e^{-x}), $x = 8$

Particle Size = 9

Fix $r(0)$ Fix $p(0)$ Do swarm Beam

Plot $r(t)$ Plot $p(t)$

Color action No stops Field vectors Info

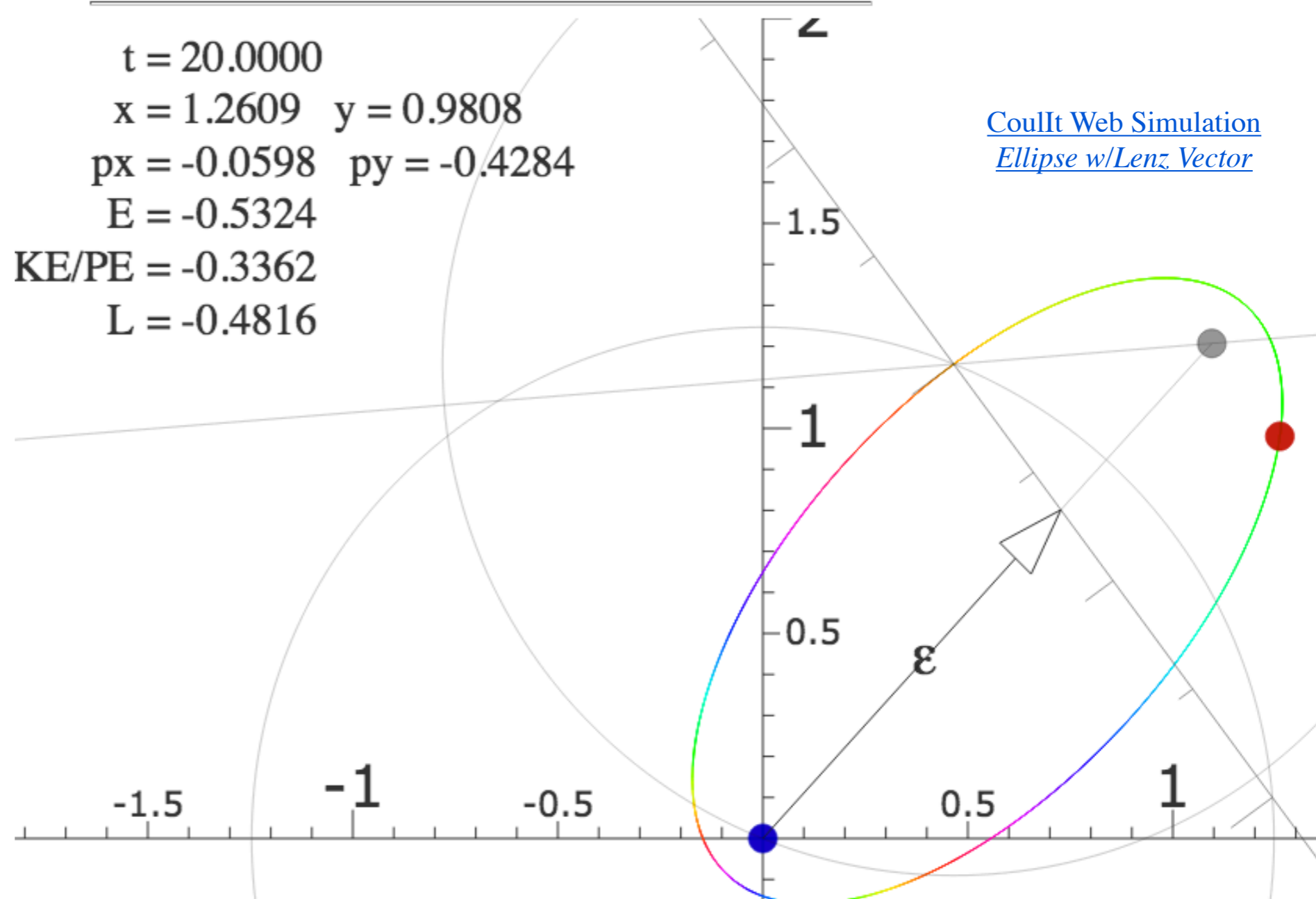
Draw masses Axes Coordinates Lenz

Set p by ϕ Elastic 2 Free

Save to GIF

$t = 20.0000$
 $x = 1.2609$ $y = 0.9808$
 $p_x = -0.0598$ $p_y = -0.4284$
 $E = -0.5324$
 $KE/PE = -0.3362$
 $L = -0.4816$

[CoulIt Web Simulation](#)
[Ellipse w/Lenz Vector](#)



Chapter 1 Orbit Families and Action

Families of particle orbits are drawn in a varying color which represents the classical action or Hamilton's characteristic function $SH = \int p dq$. (Sometimes SH is called 'reduced action'.) The color is chosen by first calculating $c = SH \text{ modulo } \hbar$ (You can change Planck's constant from its default value $\hbar/2\pi = 1.0$) The chromatic value c assigns the hue by its position on the color wheel (e.g.; $c=0$ is red, $c=0.2$ is a yellow, $c=0.5$ is a green, etc.).

Chapter 2 Rutherford Scattering

A parallel beam of iso-energetic alpha particles undergo Rutherford scattering from a coulomb field of a nucleus as calculated in these demos. It is also the ideal pattern of paths followed by intergalactic hydrogen in perturbed by the solar wind.

Chapter 3 Coulomb Field (H atom)

Orbits in an attractive Coulomb field are calculated here. You may select the initial position $(x(0), y(0))$ by moving the mouse to a desired launch point, and then select the initial momentum $(p_x(0), p_y(0))$ by pressing the mouse button and dragging.

Chapter 4 Molecular Ion Orbits

Orbits around two fixed nuclei are calculated here. A set of elliptic coordinates are drawn in the background. After running a few trajectories you may notice that their caustics conform to one or two of the elliptic coordinate lines.

Volcanoes of Io (Paths=180, No color quant.) Parabolic Fountain (Uniform)

Space Bomb (Coulomb) Exploding Starlet (IHO)

Synchrotron Motion (Crossed E & B fields)

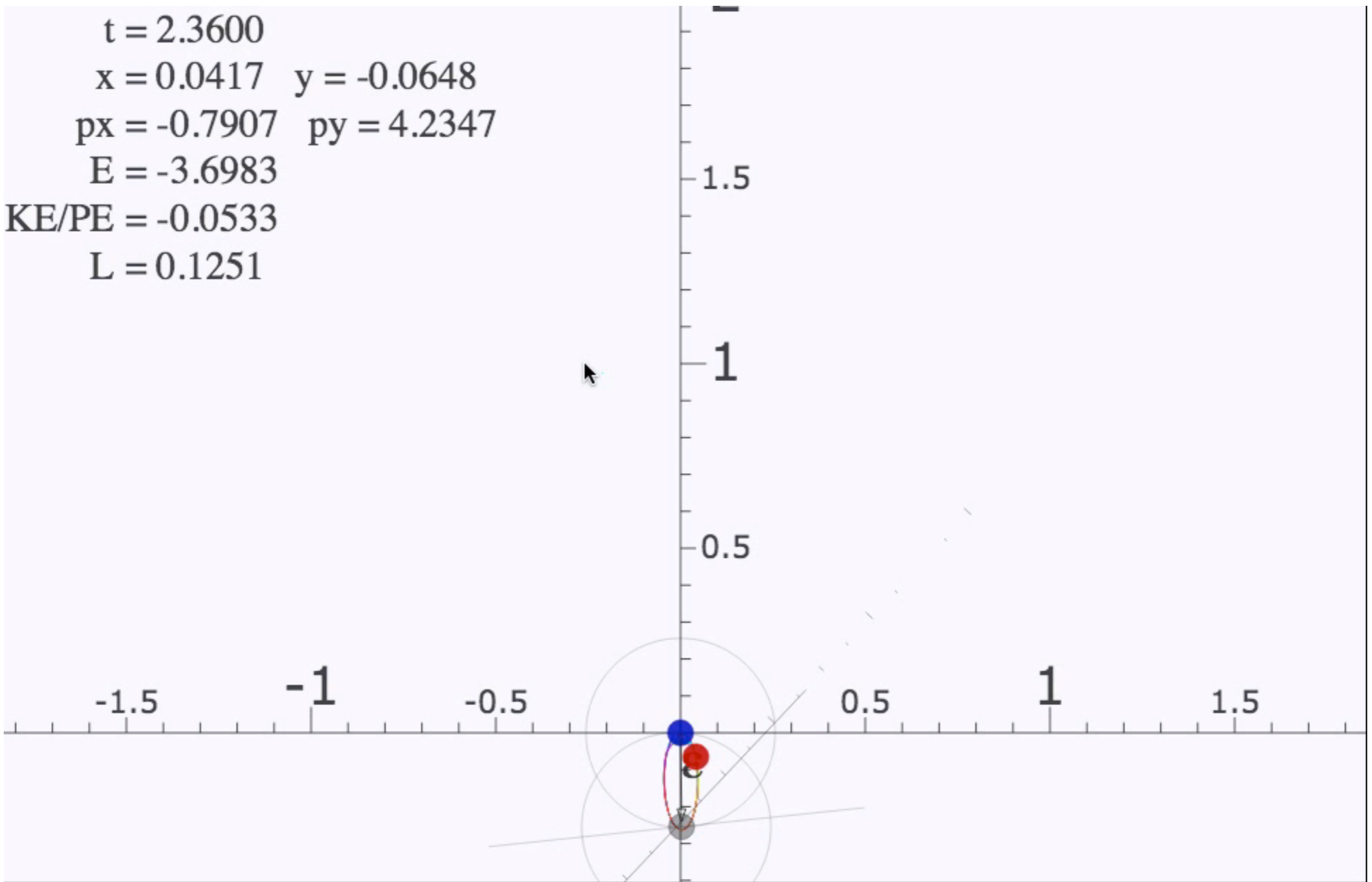
Rutherford scattering 2-Electron Orbits

Atomic Orbits

Molecular Ion Orbits

Oscillator Scattering 2-Particle Orbits 2-Particle Collision

$t = 2.3600$
 $x = 0.0417$ $y = -0.0648$
 $p_x = -0.7907$ $p_y = 4.2347$
 $E = -3.6983$
 $KE/PE = -0.0533$
 $L = 0.1251$



[Play this movie of \$\epsilon\$ -construction by CouItWeb](#)

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Radius r :

$$r = \frac{\lambda}{1 - \epsilon \cos \phi} = \frac{L^2/km}{1 - \epsilon \cos \phi}$$

Polar angle ϕ using: $L = mr^2 \frac{d\phi}{dt} = mr^2 \dot{\phi}$

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$$\text{using: } \frac{1}{r^2} = \left(\frac{km}{L^2} \right)^2 (1 - \epsilon \cos \phi)^2$$

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$$r\dot{\phi} = \frac{L}{mr}$$

$$\text{using: } \frac{1}{r^2} = \left(\frac{km}{L^2} \right)^2 (1 - \epsilon \cos \phi)^2$$

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$$r\dot{\phi} = \frac{L}{mr} = \frac{L}{m} \frac{1}{r} = \frac{L}{m} \left(\frac{km}{L^2} \right) (1 - \epsilon \cos \phi) = \frac{k}{L} (1 - \epsilon \cos \phi)$$

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$$\dot{r} = -\frac{k}{L^2} m r^2 \dot{\phi} \epsilon \sin \phi = -\frac{k}{L} \epsilon \sin \phi$$

Polar angle ϕ using: $L = m r^2 \frac{d\phi}{dt} = m r^2 \dot{\phi}$

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$$\dot{x} = \frac{dx}{dt} = \dot{r} \cos \phi - \sin \phi r \dot{\phi}$$

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again using: $L = mr^2 \dot{\phi}$

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Polar angle ϕ using: $L = mr^2 \frac{d\phi}{dt} = mr^2 \dot{\phi}$

$$\dot{\phi} = \frac{L}{mr^2} = \frac{L}{m} \frac{1}{r^2} = \frac{L}{m} \left(\frac{km}{L^2}\right)^2 (1 - \epsilon \cos \phi)^2$$

$$r \dot{\phi} = \frac{L}{mr} = \frac{L}{m} \frac{1}{r} = \frac{L}{m} \left(\frac{km}{L^2}\right) (1 - \epsilon \cos \phi) = \frac{k}{L} (1 - \epsilon \cos \phi)$$

$$\text{using: } \frac{1}{r^2} = \left(\frac{km}{L^2}\right)^2 (1 - \epsilon \cos \phi)^2$$

$$\text{using: } \frac{1}{(1 - \epsilon \cos \phi)^2} = \left(\frac{km}{L^2}\right)^2 r^2$$

again using: $L = mr^2 \dot{\phi}$

Cartesian $y = r \sin \phi$:

$$\begin{aligned} \dot{y} &= \frac{dy}{dt} = \dot{r} \sin \phi + \cos \phi r \dot{\phi} \\ &= -\frac{k}{L} \epsilon \sin \phi \sin \phi + \cos \phi \frac{k}{L} (1 - \epsilon \cos \phi) \\ &= \frac{k}{L} (\cos \phi - \epsilon) \end{aligned}$$

Coulomb orbit algebra of ϵ -vector and Kepler dynamics of momentum $\mathbf{p}=m\mathbf{v}$

Finding time derivatives of orbital coordinates r, ϕ, x, y , and eventually velocity \mathbf{v} or momentum $\mathbf{p}=m\mathbf{v}$

Radius r :

$$r = \frac{\lambda}{1 - \epsilon \cos \phi} = \frac{L^2/km}{1 - \epsilon \cos \phi}$$

$$\dot{r} = \frac{dr}{dt} = \frac{L^2}{km} \frac{-\frac{d}{dt}(-\epsilon \cos \phi)}{(1 - \epsilon \cos \phi)^2}$$

$$\dot{r} = \frac{L^2}{km} \frac{-\epsilon \sin \phi \dot{\phi}}{(1 - \epsilon \cos \phi)^2}$$

$$\dot{r} = -\frac{L^2}{km} \left(\frac{km}{L^2}\right)^2 r^2 \dot{\phi} \epsilon \sin \phi$$

$$\dot{r} = -\frac{k}{L^2} m r^2 \dot{\phi} \epsilon \sin \phi = -\frac{k}{L} \epsilon \sin \phi$$

Cartesian $x = r \cos \phi$:

$$\dot{x} = \frac{dx}{dt} = \dot{r} \cos \phi - \sin \phi r \dot{\phi}$$

$$= -\frac{k}{L} \sin \phi$$

$$p_x = m\dot{x} = -\frac{mk}{L} \sin \phi$$

Velocity:

Momentum:

Polar angle ϕ using: $L = m r^2 \frac{d\phi}{dt} = m r^2 \dot{\phi}$

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$$= \frac{k}{L} (\cos \phi - \epsilon)$$

$$p_y = m\dot{y} = \frac{mk}{L} (\cos \phi - \epsilon)$$

\mathbf{p} traces an off-center circle!

Rutherford scattering and hyperbolic orbit geometry

Backward vs forward scattering angles and orbit construction example

Parabolic “kite” and orbital envelope geometry

Differential and total scattering cross-sections

Eccentricity vector $\boldsymbol{\varepsilon}$ and (ε, λ) -geometry of orbital mechanics

Projection $\boldsymbol{\varepsilon} \cdot \mathbf{r}$ geometry of $\boldsymbol{\varepsilon}$ -vector and orbital radius \mathbf{r}

Review and connection to usual orbital algebra (previous lecture)

Projection $\boldsymbol{\varepsilon} \cdot \mathbf{p}$ geometry of $\boldsymbol{\varepsilon}$ -vector and momentum $\mathbf{p}=m\mathbf{v}$

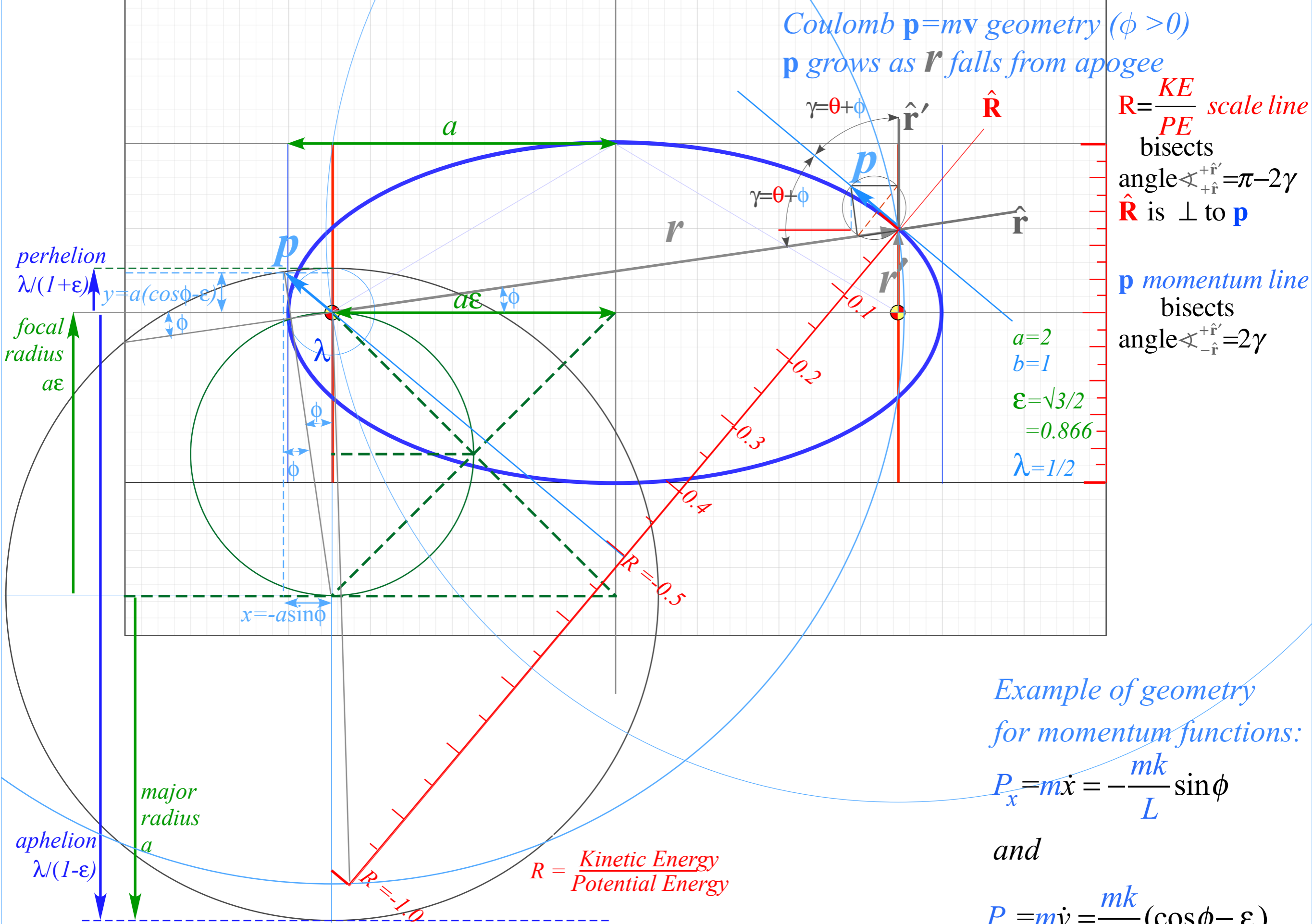
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➔ *Example of complete (\mathbf{r}, \mathbf{p}) -geometry of elliptical orbit*

Connection formulas for (γ, R) -parameters with (a, b) and (ε, λ)



Coulomb $\mathbf{p}=m\mathbf{v}$ geometry ($\phi > 0$)
 \mathbf{p} grows as \mathbf{r} falls from apogee

$R = \frac{KE}{PE}$ scale line

bisects
 angle $\angle_{+\hat{r}}^{+\hat{r}'} = \pi - 2\gamma$
 \hat{R} is \perp to \mathbf{p}

\mathbf{p} momentum line
 bisects
 angle $\angle_{-\hat{r}}^{+\hat{r}'} = 2\gamma$

$a=2$
 $b=1$
 $\epsilon = \sqrt{3}/2 = 0.866$
 $\lambda = 1/2$

Example of geometry for momentum functions:

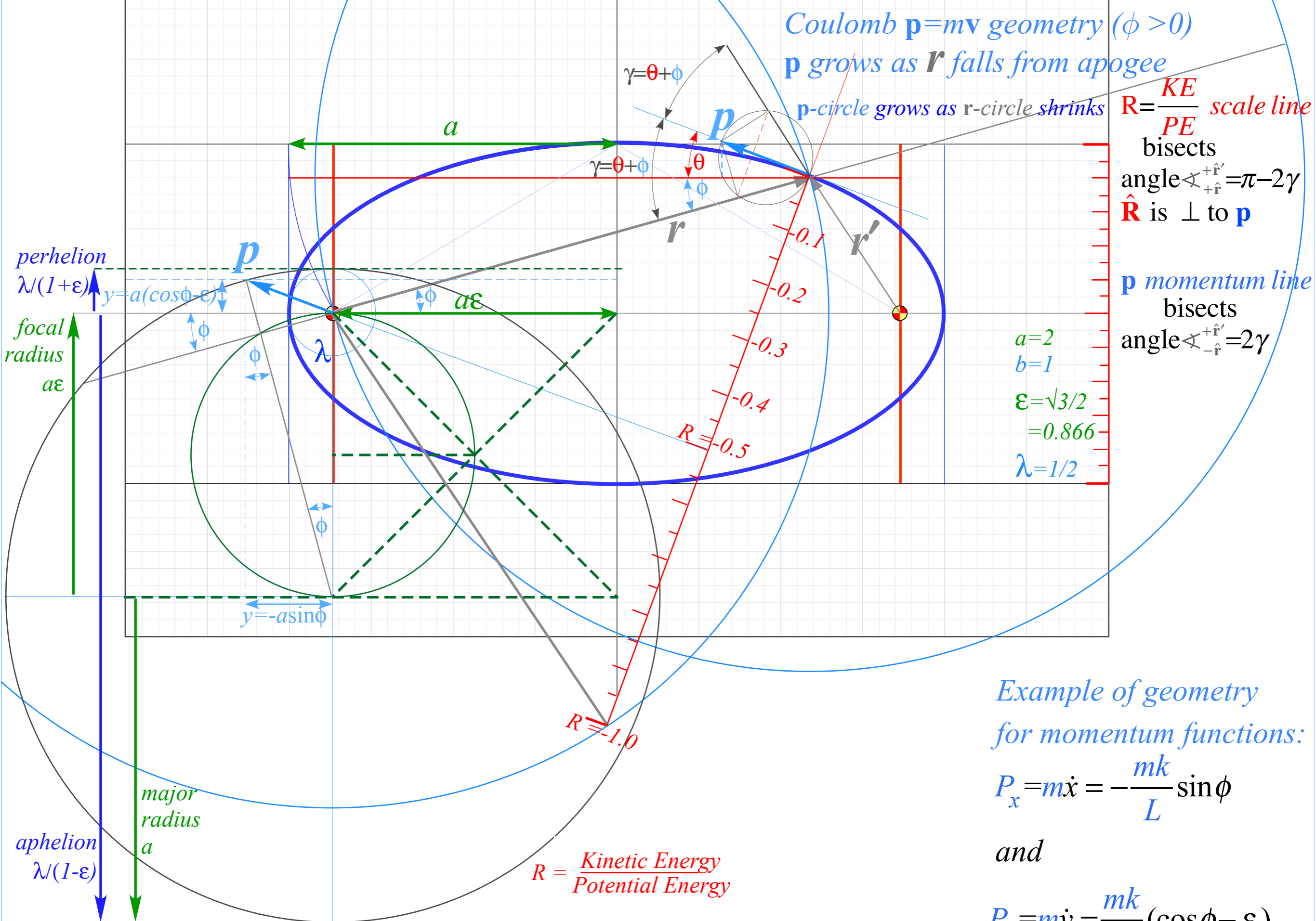
$$P_x = m\dot{x} = -\frac{mk}{L} \sin\phi$$

and

$$P_y = m\dot{y} = \frac{mk}{L} (\cos\phi - \epsilon)$$

$0 > R = KE/PE > -1$ scale subtends angle 2γ with length $2r \sin\gamma$ as is derived before on p. 67-71.

Note similarity of (\mathbf{R}, \mathbf{r}) -triangle in \mathbf{r} -circle of radius r to that in \mathbf{p} -circle of diameter p above.



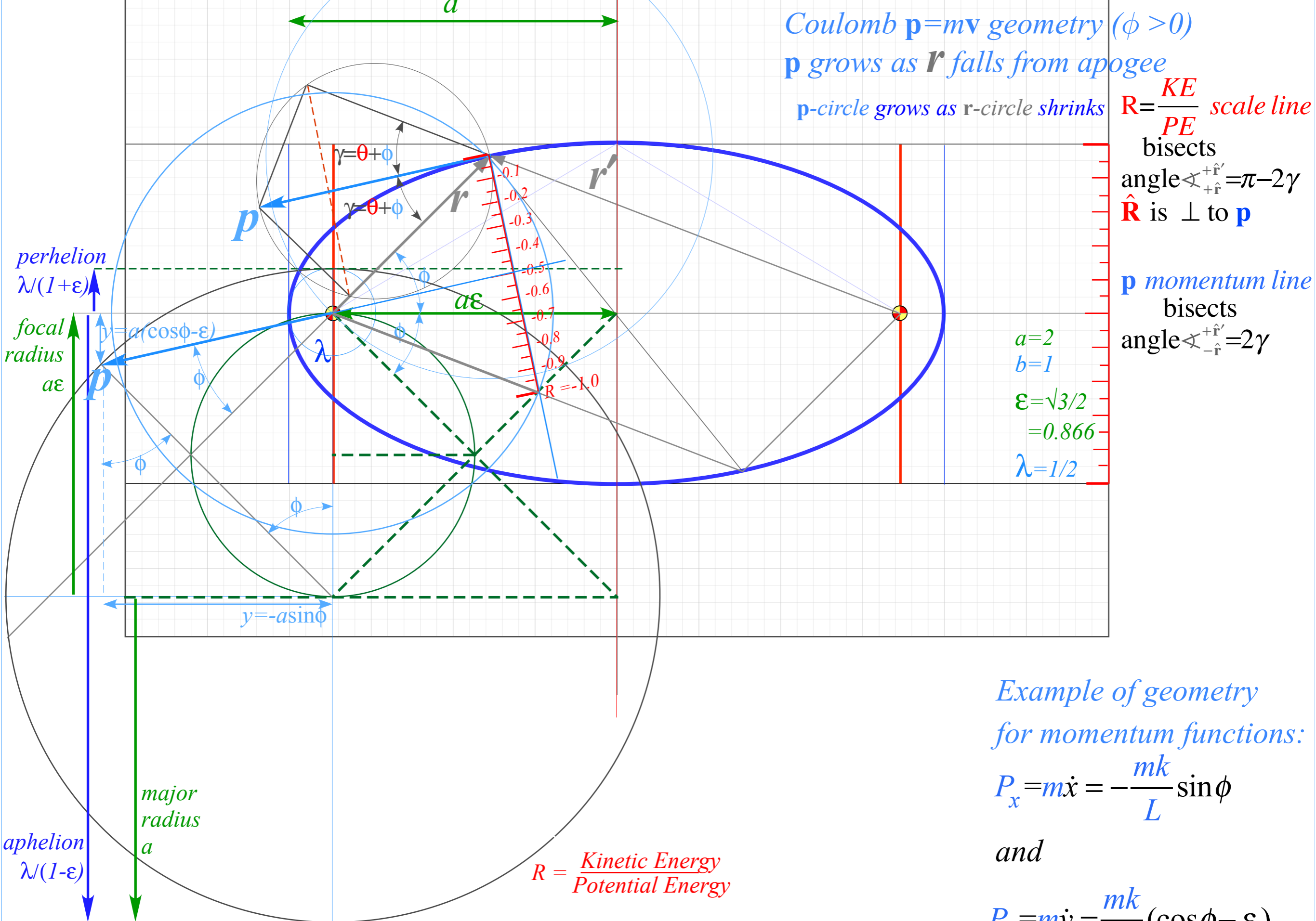
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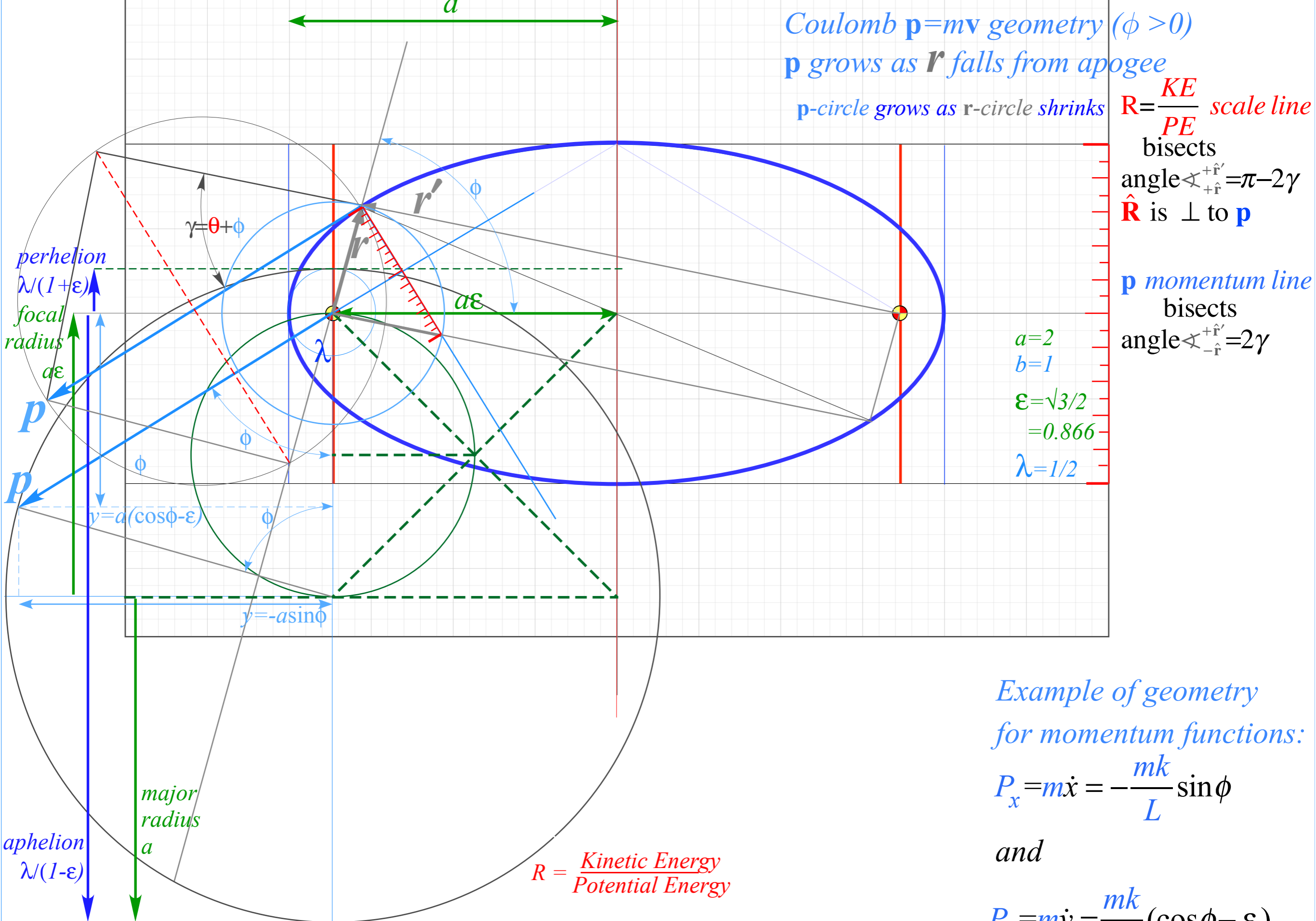
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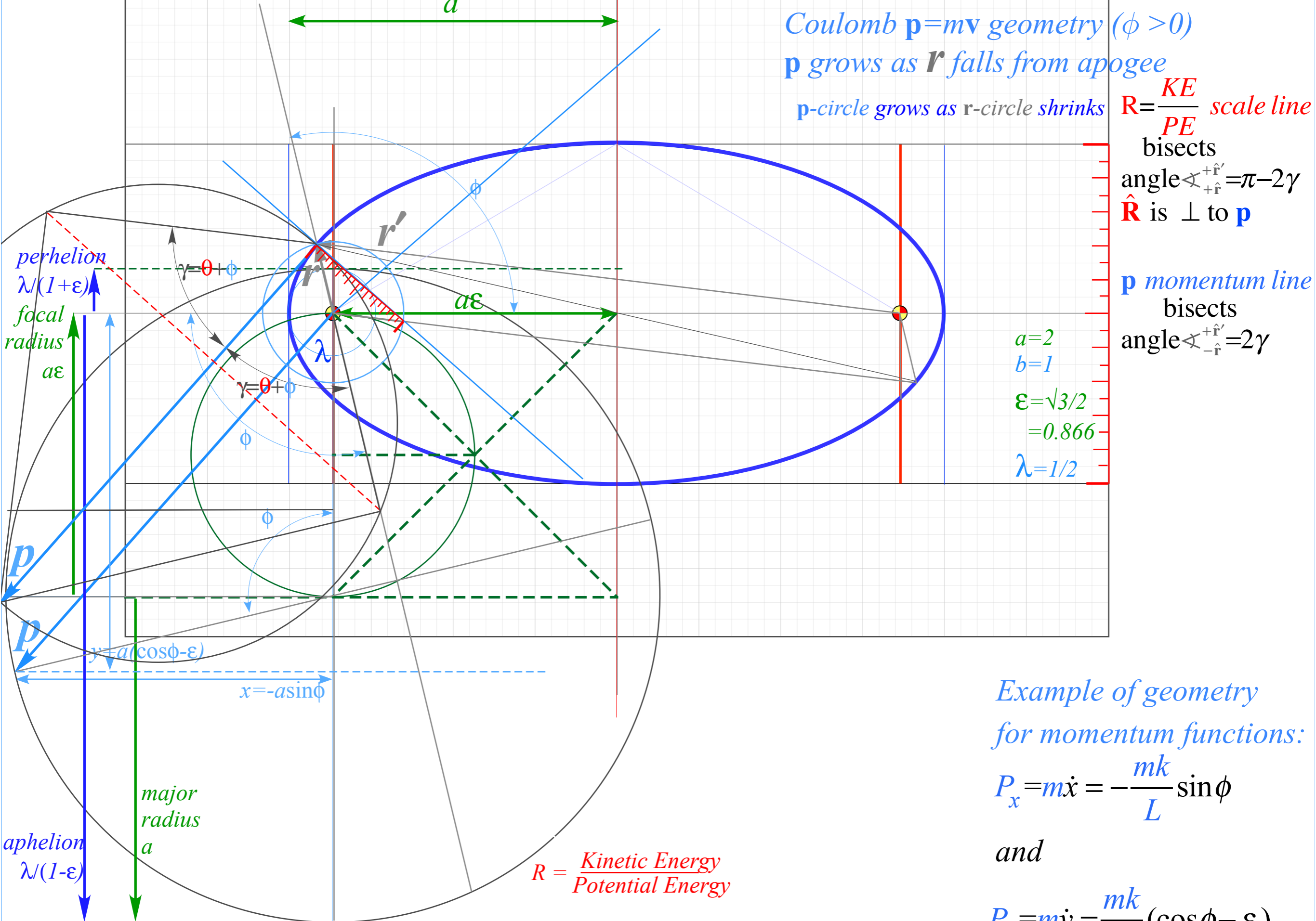
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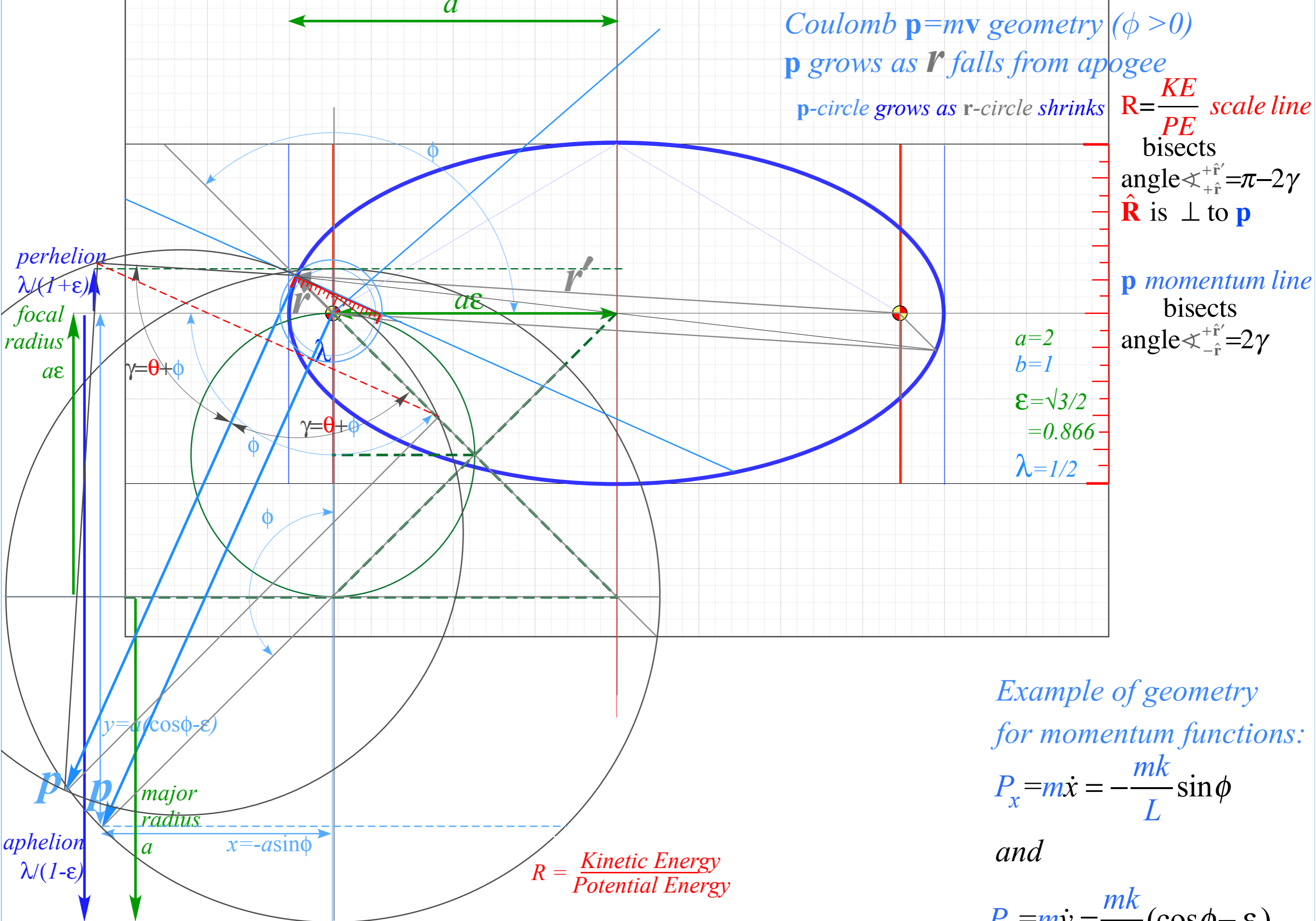
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Coulomb $\mathbf{p}=m\mathbf{v}$ geometry ($\phi > 0$)

\mathbf{p} grows as \mathbf{r} falls from apogee

\mathbf{p} -circle grows as \mathbf{r} -circle shrinks

$R = \frac{KE}{PE}$ scale line

bisects

angle $\angle_{+\hat{r}}^{+\hat{r}'} = \pi - 2\gamma$

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Algebra of ϵ -construction geometry

The *eccentricity* parameter relates ratios $R = \frac{KE}{PE}$ and $\frac{b^2}{a^2}$

$$\epsilon^2 = 1 + 4R(R+1)\sin^2\gamma$$

$$= 1 - \frac{b^2}{a^2} \quad \text{for ellipse} \quad (\epsilon < 1)$$

$$= 1 + \frac{b^2}{a^2} \quad \text{for hyperbola} \quad (\epsilon > 1)$$

Three pairs of parameters for Coulomb orbits:
1. Cartesian (a,b) , 2. Physics (E,L) , 3. Polar (ϵ,λ)

Now we relate a 4th pair: 4. Initial (γ,R)

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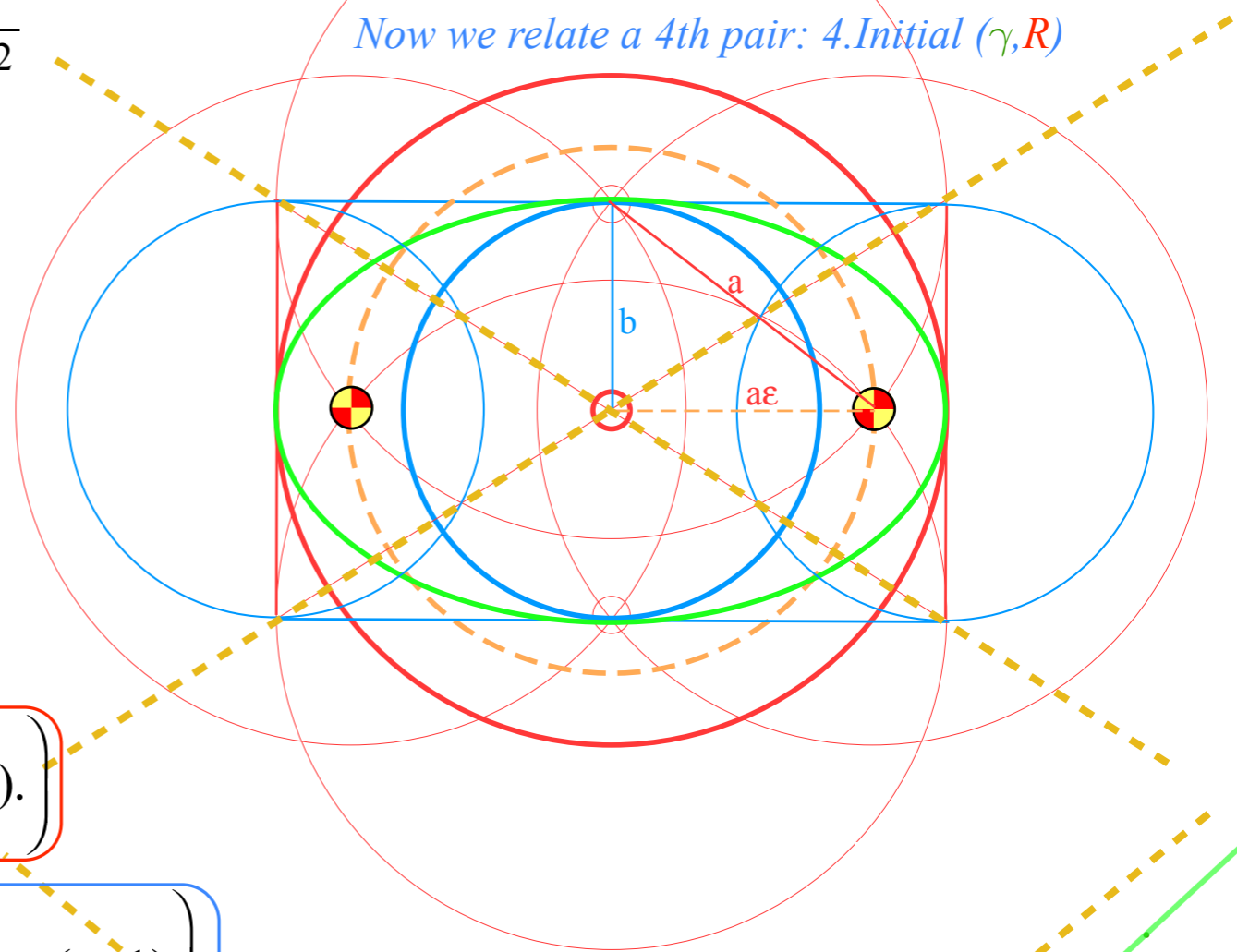
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From ϵ^2 result (at top):

$$\frac{b}{a} = 2\sqrt{\mp R(R+1)} \sin\gamma = \sqrt{\pm(1-\epsilon^2)}$$

